US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

OMB No. 2120-0020
Exp: 5/31/2018

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a))

1. Aircraft	Nationality and Registration Mark: N4260S	Serial No.: E-959	
	Make: Beechcraft	Model: Bonanza	Series: A-36
2. Owner	Name *(As shown on registration certificate)*: Luqa Aviation LLC	Address *(As shown on registration certificate)*: Address 1517 Perimeter Rd Ste 505; City West Palm Beach; State FL; Zip 33406-1445; Country USA	

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
☐	☒	AIRFRAME	Beechcraft	*(As described in Item 1 above)*	E-959
☐	☐	POWERPLANT			
☐	☐	PROPELLER			
☐	☐	APPLIANCE	Type / Manufacturer		

6. Conformity Statement

A. Agency's Name and Address
Name: Frank Cecilia
Address: 13393 151 Lane N
City: Jupiter State: FL
Zip: 33478 Country: USA

B. Kind of Agency
- [X] U. S. Certificated Mechanic
- [ ] Foreign Certificated Mechanic
- [ ] Certificated Repair Station
- [ ] Certificated Maintenance Organization
- [ ] Manufacturer

C. Certificate No. 3439119

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B ☐

Signature/Date of Authorized Individual: 25 OCTOBER 2021

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is [X] Approved [ ] Rejected

BY								
		FAA Flt. Standards Inspector		Manufacturer		Maintenance Organization		Persons Approved by Canadian Department of Transport
		FAA Designee		Repair Station	X	Inspection Authorization		Other *(Specify)*

Certificate or Designation No. 3439119

Signature/Date of Authorized Individual: 25 OCTOBER 2021

FAA Form 337 (10/06)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

USA N4260S — 25 OCT 2021
Nationality and Registration Mark — Date

Installed ThorPlas self lubricating bushings in the inboard and outboard main landing gear doors IAW instructions provided in STC SA03343NY.

Installed AeroTough GF self lubricating bushings in the nose gear doors IAW instructions provided in STC SA03852NY.

No change to aircraft weight and balance.

----------END----------

☐ Additional Sheets Are Attached

United States of America
Department of Transportation
Federal Aviation Administration

Supplemental Type Certificate

IMPORT

Number: SA03852NY

This certificate issued to: Marsh Brothers Aviation Inc
3225 Mainway
Burlington, Ontario, Canada
L7M 1A6

certifies that the change in the type design for the following product with the limitations and conditions therefore as specified hereon meets the airworthiness requirements of Part 23 *of the* Federal Aviation Regulations.

Original Product – Type Certificate Number:
3A15

Make: Textron Aviation Inc.
Model: 35-33, 35-A33, 35-B33, 35-C33, 35-C33A, E33, E33A, E33C, F33, F33A, F33C, G33, H35, J35, K35, M35, N35, P35, S35, V35, V35A, V35B, 36, A36, A36TC, B36TC

Description of Type Design Change:

1. Installation of Nose Landing Gear Door Hinge Bushings in accordance with Transport Canada Civil Aviation (TCCA) approved Marsh Brothers Aviation Inc Installations Instruction I10005, dated August 9, 2016 or later TCCA approved revision.
2. Maintenance in accordance with TCCA accepted Marsh Brothers Aviation Inc. Maintenance Manual Supplement MMS-0005 dated August 9, 2016 or later Transport Canada accepted revision .

Limitations and Conditions:

1. The installer must determine whether this design change is compatible with previously approved modifications.
2. If the holder agrees to permit another person to use this certificate to alter a product, the holder must give the other person written evidence of that permission.

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, and revoked or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of Application: September 12, 2016

Date Reissued:

Date of Issuance: January 6, 2017

Date Amended:

By Direction of the Administrator

for *Signature* _____

Title Gaetano Sciortino
Manager
New York Aircraft Certification Office

United States of America

Department of Transportation -- Federal Aviation Administration

Supplemental Type Certificate

IMPORT

Number SA03343NY

This certificate issued to Marsh Brothers Aviation Inc.
3225 Mainway
Burlington, Ontario
Canada L7M 1A6

certifies that the change in the type design for the following product with the limitations and conditions therefore as specified hereon meets the airworthiness requirements of Part 3 *of the* Civil Aviation *Regulations.*

Original Product – Type Certificate Number: 3A15

Make: Beechcraft Corporation

Model: 35-33, 35-A33, 35-B33, 35-C33, 35-C33A, 36, A36, A36TC, B36TC, E33, E33A, E33C, F33, F33A, F33C, G33, G36, H35, J35, K35, M35, N35, P35, S35, V35, V35A, V35B

Description of Type Design Change:
Installation of the replacement to the OEM landing gear door hinge bushings in accordance with Marsh Brothers Aviation Inc. Installation Instructions II0001, Revision Original, dated November 1, 2012 or later Transport Canada approved revision.
For Instructions for Continued Airworthiness refer to Marsh Brothers Aviation Inc. Maintenance Manual Supplement, MMS0001, Revision Original, dated November 1, 2012, or later Transport Canada accepted revision.

Limitations and Conditions:

1. Mandatory replacement time and/or inspection intervals, defined in the "Airworthiness Limitation" section of Marsh Brothers Aviation Inc. Maintenance Manual Supplement, MMS0001, Revision Original, dated November 1, 2012, or later Transport Canada accepted revision, are to be complied with.

(See Continuation Sheet 3 of 3)

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application: July 9, 2013

Date of issuance: November 14, 2013

Date reissued:

Date amended:

By direction of the Administrator

(Signature)

For Gaetano Sciortino
Manager
New York Aircraft Certification Office

(Title)

Any alteration of this certificate is punishable by a fine of not exceeding $1,000, or imprisonment not exceeding 3 years, or both.

 This certificate may be transferred in accordance with FAR 21.47.

US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

OMB No. 2120-0020
Exp: 5/31/2018

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a))

1. Aircraft	Nationality and Registration Mark: N4260S	Serial No.: E-959	
	Make: Beechcraft	Model: Bonanza	Series: A36
2. Owner	Name *(As shown on registration certificate)*: Luqa Aviation, LLC	Address *(As shown on registration certificate)*: Address: 1517 Perimeter Road Suite 505; City: West Palm Beach; State: Florida; Zip: 33406-1445; Country: USA	

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
☐	☒	AIRFRAME	Beechcraft	*(As described in Item 1 above)*	E-959
☐	☐	POWERPLANT			
☐	☐	PROPELLER			
☐	☐	APPLIANCE	Type / Manufacturer		

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	
Name: Frank Cecilia	x U. S. Certificated Mechanic	☐ Manufacturer
Address: 13393 151 Lane N	☐ Foreign Certificated Mechanic	C. Certificate No.
City: Jupiter State: Florida	☐ Certificated Repair Station	3439119
Zip: 33478 Country: USA	☐ Certificated Maintenance Organization	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B ☐	Signature/Date of Authorized Individual: 15 OCT 2020

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ Approved ☐ Rejected

BY							
	☐	FAA Flt. Standards Inspector	☐	Manufacturer	☐	Maintenance Organization	☐ Persons Approved by Canadian Department of Transport
	☐	FAA Designee	☐	Repair Station	X	Inspection Authorization	Other *(Specify)*

Certificate or Designation No. 3439119	Signature/Date of Authorized Individual: 15 OCT 2020

FAA Form 337 (10/06)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

USA N4260S	15 October, 2020
Nationality and Registration Mark	Date

Installed rear baggage compartment in accordance with STC SA003004BO (STC Form 8110-2 attached). Aircraft weight and balance revised with changes from the installation.

----------END----------

☐ Additional Sheets Are Attached

United States of America
Department of Transportation
Federal Aviation Administration

Supplemental Type Certificate

Number SA00304BO

This certificate issued to:

Approach Aviation
30 Sheffield Terrace
Marlborough, MA 01752

certifies that the change in the type design for the following product with the limitations and conditions therefore as specified hereon meets the airworthiness requirements of Part 3 of the Civil Air Regulations.

Original Product - Type Certificate Number: 3A15
Make: Beechcraft Corporation
Model: 36, A36

Description of Type Design Change:
Installation of an extended baggage compartment in accordance with Approach Aviation Master Document List (MDL), Document BE36-MASTERDWGLIST, Revision B, dated May 20, 2016, or later FAA approved revision.

Limitations and Conditions:

1. Operation must be in accordance with Approach Aviation Aircraft Flight Manual Supplement (AFMS), Document BE36-AFMS-001, Revision A, dated October 06, 2010, FAA approved on December 17, 2010, or later FAA approved revision.
   The AFMS must be carried in the aircraft during all flights.

(continued on Page 3 of 3)

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, and revoked or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of Application: September 02, 2010
Date of Issuance: December 23, 2010
Date reissued: May 23, 2016
Date amended:

By direction of the Administrator

Signature Robert G. Mann

Title Manager, Boston Aircraft Certification Office

United States of America
Department of Transportation
Federal Aviation Administration

Supplemental Type Certificate

Number SA00304BO
May 23, 2016

Limitations and Conditions: (continued from Page 1 of 3)

2. Maintenance must be in accordance with Approach Aviation Instructions for Continued Airworthiness (ICA), Document BE36-ICA-001, Revision A, dated October 06, 2010, or later FAA accepted revision.

   The ICA must be made available to the operator at the time of installation.
3. Eligible for installation only on Beechcraft Corporation model 36 and model A36, serial numbers E-1 through E-1370.
4. Compatibility of this design with previously approved modifications must be determined by the installer.

Certification Basis:
Based on 14 CFR § 21.115 and § 21.101, and the FAA policy for design changes that are identified as not significant in FAA Order 8110.48, the certification basis for the Beechcraft Corporation model 36 and model A36 aircraft is as follows:

a. The type certification basis for parts **not changed or not affected** by the change is shown on TCDS 3A15.

b. The modification certification basis for parts **changed or affected** by the change since the reference application date, September 02, 2010, is based upon CAR 3 as amended on May 15. 1956, as follows:

   Regulations at the latest amendment May 15, 1956, through 3-8
   3.171. 3.172, 3.173, 3.174. 3.291, 3.292, 3.293, 3.294, 3.295, 3.296, 3.301, 3.302, 3.307, 3.386. 3.392

   NOTE: Findings of compliance were also made to 14 CFR as follows:
   part 23 through amendment 23-59: 23.1529, 23.1583, 23.1585, 23.1587. 23.1589
   part 25 through amendment 25-129: 25.853(a), F25-1

   Regulations at an intermediate amendment
   None

   Regulations at the amendment level in TCDS 3A15
   None

----------END----------

Pursuant to Title 49 United States Code § 44704 (b) (3) (effective October 19, 1996) the signature and seal below constitute the agreement and permission of SalesApproach, Inc. DBA Approach Aviation, allowing the registered owner of Registration #: N4260S Beechcraft 36/A36 S/N E-959, an ONLY that S/N aircraft, by application of STC No. SA00304BO, for the purpose of installing the Extended Baggage Compartment that is the subject of the STC.

SalesApproach, Inc. DBA Approach Aviation

(SEAL)

By Jeffrey Simon, President

9/10/20
Date

NOTE:

IT IS THE RESPONSIBILITY OF THE INSTALLER TO ENSURE THAT THE INTERRELATIONSHIP BETWEEN THIS CHANGE AND ANY PREVIOUSLY APPROVED/INSTALLED MODIFICATIONS WILL INTRODUCE NO ADVERSE EFFECT UPON THE AIRWORTHINESS OF THE AIRPLANE.

US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved OMB No. 2120-0020 2/28/2011

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for ea such violation. (49 U.S.C. §43601(a))

1. Aircraft	Nationality and Registration Mark **N4260S**	Serial No. **E-959**	
	Make **BEECH**	Model **A36**	Series
2. Owner	Name *(As shown on registration certificate)* LUQA AVIATION LLC	Address *(As shown on registration certificate)* Address 1517 Perimeter Rd STE 505 Ctiy West Palm Beach State FL Zip 33406 Country USA	

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
☐	☐	AIRFRAME	—	*(As described in Item 1 above)*	—
☐	☒	POWERPLANT	**Continental**	**IO-520-BA**	**249720-R**
☐	☐	PROPELLER			
☐	☐	APPLIANCE	Type		
			Manufacturer		

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency		
Name Frank CECILIA	☒ U.S. Certified Mechanic	☐ Manufacturer	
Address 13393 151 Lane N	☐ Foreign Certified Mechanic	C. Certificate No.	
City Jupiter State FL	☐ Certified Repair Station	3439119	
Zip 33478 Country USA	☐ Certificated Maintenance Organization		

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulation and the that information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B ☐

Signature/Date of Authorized Individual: Frank C 28 May 2019

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ Approve ☐ Rejected

BY								
		FAA Flt. Standards Inspector		Manufacturer		Maintenance Organization		Persons Approved by Canadian Department of Transport
		FAA Designee		Repair Station	X	Inspection Authorization		Other *(Specify)*

Certificate or Designation No. 3439119

Signature/Date of Authorized Individual: Frank C 28 May 2019

FAA Form 337 (10-06)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N4260S	28 May 2019
Nationality and Registration Mark	Date

Removed six (6) Continental fuel nozzles P/N 627335D13A and installed General Aviation Modifications, Inc. GAMIjectors® kit No. GAD13A S/N 28203 STC No. SE09217SC PMA No. PQ821SW per GAMIjectors® Installation Procedure No. IP-96-001 (Rev 005) dated 11 Oct 2002. No change in weight or balance.

//

//END///

☐ Additional Sheets Are Attached

United States Of America

Department of Transportation – Federal Aviation Administration

Supplemental Type Certificate

Number SE09217SC

This Certificate issued to General Aviation Modifications, Inc.
2800 Airport Road
Hangar A
Ada, OK 74820

*certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part * of the * Regulations.*

Original Product Type Certificate Number: *[see attached FAA Approved Model List (AML)
Make: for list of Approved models and applicable airworthiness
Model: regulations]

Description of Type Design Change: Installation of modified fuel injector nozzles in accordance with General Aviation Modifications, Inc. "Data List", Revision 003, dated September 24, 1996, and Installation Procedure No. IP-96-001, "GAMIJECTOR Installation Procedure for Teledyne Continental Motors (TCM) IO-550/520/470 Engin Revision 004, dated September 25, 1996, or later FAA approved revisions.

Limitations and Conditions: Compatibility of this design change with previously approved modifications must be determined by the installer. This STC does not allow any adjustment to the fuel injection control unit. Compliance with the following Teledyne Continental Service Bulletins is required: M78-11 Fuel Injection System Inspection; M85-19 Fuel Injection System Application Guide; M87-12 Rev. 1 Recommended Fuel and Oil Grades; M89-10 Fuel System Adjustment; and M89-18 EGT recommendations.

This certificate and the supporting data which is the basis for the approval shall remain in effect until surrendered, suspended, revoked or a termination date is otherwise established by the Adminstrator of the Federal Aviation Administration.

Date of application: December 22, 1995

Date reissued:

Date of issuance: April 30, 1996

Date ammended: 9/27/1996; January 08, 1997, Rev.

By direction of the Administrator

(Signature)

A. J. Merrill
Manager, Special Certification Office
Southwest Region

(Title)

Any alteration of this certificate is punishable by a fine of not exceeding $1,000, or imprisonment not exceeding 3 years, or both.

FAA APPROVED MODEL LIST (AML)

STC No. SE09217SC

General Aviation Modifications, Inc.
2800 Airport Road, Hangar A
Ada, OK 74820

Date of Issuance: April 30, 1996
Date Amended: January 08, 1997, Rev. 2

Item	Aircraft Make	Aircraft Model	Original Type Certificate Number	Regulation / Part
1	Teledyne Continental Motors	IO-470-A & C	E-273	CAR 13
2	Teledyne Continental Motors	IO-470-D, E, F, G, H, J, K, L, LO, M, N, P, R, S, T, U, V, VO, & L/IO-470-A	3E1	CAR 13
3	Teledyne Continental Motors	IO-520-A, B, BA, BB, C, CB, D, E, F, J, K, L, M, & MB	E5CE	CAR 13
4	Teledyne Continental Motors	IO-550-A, B, C, D, E, F, & L	E3SO	FAR 33

FAA Approved:

Date: January 08, 1997

Persuant to Title 49 United States Code § 44704 (B) (3) (effective October 19, 1996) the signature below constitutes the agreement and permission of General Aviation Modifications, Inc., allowing the registered owner of **N4260S**, to alter that certain Continental Motors engine:

Model **IO-520-BA**, SN **249720-R**, and ONLY that serial number, by application of STC No. SE9217SC, to that specific engine, for the purpose of installing the fuel injector nozzles which are the subject of that STC.

General Aviation Modification, Inc.

By Tim Boehl, President
Or George W. Braly, CEO

[SEAL]

7-May-2019
Date

U.S Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020
11/30/2007

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed $1,000 for each such violation (Section 901 Federal Aviation Act 1958)

1. Aircraft	Nationality and Registration Mark USA – N4260S	Serial No. E-959	
	Make BEECH	Model A36	Series
2. Owner	Name *(As shown on registration certificate)* LUQA AVIATION LLC	Address *(As shown on registration certificate)* Address: 1517 PERIMETER RD STE 505 City: WEST PALM BEACH State: FLORIDA Zip: 33406-1445 Country: UNITED STATES	

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial Number
☐	☒	AIRFRAME	—	*(As described in Item 1 above)*	—
☐	☐	POWERPLANT			
☐	☐	PROPELLER			
☐	☐	APPLIANCE	Type Manufacturer		

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency		
Name: Banyan Air Service Inc.	☐ U.S. Certificated Mechanic	☐ Manufacturer	
Address: 5360 NW 20th Terrace	☐ Foreign Certificated Mechanic	C. Certificate No.	
City: Fort Lauderdale State: FL	☒ Certificated Repair Station	WD4R618M	
Zip: 33309 Country: U.S.A.	☐ Certificated Maintenance Organization		

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B ☐	Signature/Date of Authorized Individual Date: January 28, 2019 Robert Tuttle, Chief Inspector

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ APPROVED ☐ REJECTED

BY								
		FAA Flt Standards Inspector		Manufacturer		Maintenance Organization		Person Approved by Canadian Department of Transport
		FAA Designee	X	Repair Station		Inspection Authorization		Other *(Specify)*

Certificate or Designation No. WD4R618M Page 1 of 3	Signature/Date of Authorized Individual Date: January 28, 2019 Robert Tuttle, Chief Inspector

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

USA – N4260S	E-959	January 28, 2019
Nationality and Registration Mark	Serial Number	Date

The following work was performed under Banyan Air Services work order number 42542:

1. Removed existing Autopilot System and associated hardware and wiring to perform installation of Autopilot System upgrade.
2. Removed existing Flight Instruments and Indicators to perform installation of Integrated Avionics System upgrade.
3. Removed existing KT-76A Transponder System and associated hardware to perform installation of Transponder System with ADSB-Out functionality upgrade.
4. Performed installation of Garmin International, Inc. STC No. SA01844WI entitled "Installation of Garmin GFC 600 Automatic Flight Control System" in accordance with Garmin Master Drawing List (MDL) Document No. 005-01009-01 Revision 7, dated October 12, 2018, Garmin GFC 600 Automatic Flight Control System Part 23 AML STC Installation Manual Document No. 190-01937-00 Revision 10, dated September 13, 2018 and Garmin GFC 600 Automatic Flight Control System Part 23 AML STC Maintenance Manual Document No.190-01938-00, Revision 5, dated October 12, 2018.
5. Performed installation of Garmin International, Inc. STC No. SA02571SE entitled "Installation of Garmin G500/600 TXi Integrated Avionics System (Primary Flight Display, Multi-Function Display, and Engine Indicating System Display)" in accordance with Garmin Master Drawing List (MDL) Document No. 005-00795-C2 Revision 5, dated December 05, 2018, Garmin G500/G600 TXi Part 23 AML STC Installation Manual Document No. 190-01717-B3 Revision 3, dated November 13, 2018 and Garmin G500/G600 TXi Part 23 AML STC Maintenance Manual Document No. 190-01717-B1 Revision 2, dated August 15, 2018.
6. Performed installation of Garmin International, Inc. STC No. SA01714WI entitled "Installation of Garmin transponders: (a) GTX 330/330D/33/33D with ADS-B Out functionality; (b) GTX 335/335R/345/345R with ADS-B Out and In functionality; or (c) GTX 335R/345R with ADS-B Out functionality in select airplanes installed with G950/G1000 systems" in accordance with Garmin Drawing List (MDL) Document No. 005-00734-04 Revision 11, dated March 19, 2018, Garmin GTX 3XX Part 23 AML STC Installation Manual Document No. 190-00734-10 Revision 10, dated March 14, 2018 and Garmin GTX 33X and GTX 3X5 ADS-B Maintenance Manual Document No. 190-00734-11 Revision 6, dated December 7, 2017.
7. Functional test of the above equipment has been performed with reference to 14 CFR Part §23.2505 for proper operation and does not adversely affect any other aircraft systems nor do any existing systems adversely affect the installed systems. This installation meets the requirements of 14 CFR Part §23.2500. An electrical load review was conducted and it was confirmed that this installation does not exceed 80% of the generators output.
8. The installed ADS-B OUT System was shown to meet the equipment requirements of 14 CFR Part 91.227.
9. The required FAA Approved Flight Manual Supplement (AFMS) for the installation of GFC 600 Autopilot Installed in Beechcraft 36 / 36A Document No. 190-02011-00 Revision 3, dated August 16, 2018 has been inserted into the Aircraft Flight Manual.
10. The required FAA Approved Flight Manual Supplement (AFMS) for the installation of the Garmin G500/G600 TXi Display System Document No. 190-01717-B2 Revision 2, dated August 16, 2018 has been inserted into the Aircraft Flight Manual.
11. The required FAA Approved Flight Manual Supplement (AFMS) for the installation of the Garmin GTX 33X and GTX 3X5 Transponders with ADS-B Document 190-00734-15 Revision 3, dated December 21, 2017 has been inserted into the Aircraft Flight Manual.
12. Equipment List and Weight & Balance have been revised to reflect these installations.
13. Log entry generated to reflect these installations.

INSTRUCTIONS FOR CONTINUED AIRWORTHINESS

1. The Instructions for Continued Airworthiness (ICA) for the Installation of the Garmin GFC 600 Automatic Flight Control System can be found within the Garmin GFC600 Automatic Flight Control System Part 23 AML STC Maintenance Manual document 190-01938-00, Revision 5, dated October 12, 2018 (or later approved revision). Operator should as well inspect system wiring, mounting and security during associated aircraft area inspection.
2. The Instructions for Continued Airworthiness (ICA) for the Installation of the Garmin G500/G600 TXi Integrated Avionics System (Primary Flight Display, Multi-Function Display, and Engine Indicating System Display) can be found within the Garmin G500/G600 TXi Part 23 AML STC Maintenance Manual Document No. 190-01717-B1 Revision 2, dated August 15, 2018 (or later approved revision). Operator should as well inspect system wiring, mounting and security during associated aircraft area inspection.
3. The Instructions for Continued Airworthiness (ICA) for the installation of the Garmin GTX 33X and GTX 3X5 with ADS-B Out System can be found within the Garmin GTX 33X and GTX 3X5 ADS-B Maintenance Manual Document No. 190-00734-11 Revision 6, dated December 7, 2017 (or later approved revision). Operator should as well inspect system wiring, mounting and security during associated aircraft area inspection.

----------CONTINUED----------

☒ Additional Sheets Are Attached

8. Description of Work Accomplished

USA – N4260S	E-959	January 28, 2019
Nationality and Registration Mark	Serial Number	Date

Removed Following Equipment	Weight LBS	Arm In.
KT-76A ATC P/N 066-1062-00 S/N 11423	1.70	57.05
VSI P/N 7040-B41 S/N C5880	0.70	57.05
Turn & Slip Ind. P/N 52D75-4M S/N 1966E	2.70	57.05
DG P/N 1U262-002-8 S/N T40650F	2.50	57.05
HSI P/N 4000172-8503 S/N J268	3.70	57.05
Trim SVO P/N 0105-T1 S/N 5125A	2.90	220.55
Pitch SVO P/N 0107-P4 S/N 0626-17083	2.90	211.55
KI-214 Course Ind. P/N 066-3025-01 S/N 12779	2.90	57.05
WX-950 Stormscope P/N 805-10950-001 S/N JCP07801700	2.90	57.05
Antenna P/N 805-10930-001 S/N JCA07802071	0.84	230.05
Gyro P/N 4000869-3201 S/N 4467	4.30	204.05
Autopilot Comp. P/N 01261-8-14 S/N 0626-5446A	1.10	177.25
Flux Corrector P/N 4000202-8501 S/N 2582	0.20	179.75
Flux Valve P/N 4000191-8502 S/N 4315	0.3	99.75
Fast Gyro Slaving Meter P/N 4000192-8501 S/N SS	0.51	57.05
Static Inverter P/N SPC-5(D) S/N 272	3.00	187.25
DC Converter P/N 12-12-6I S/N 0295	3.39	183.25
Pressure Transducer P/N 0111 S/N 0622-36543AA	0.23	39.00
Altitude Encoder P/N SSD-120N S/N 34713	0.50	39.00

Installed Following Equipment	Weight LBS	Arm In.
GDU 1060 P/N 011-03308-30 S/N 3M5001292	6.49	57.05
GMN 01218 P/N 011-03457-00 S/N 3M0001038	0.55	55.05
GTX 345 P/N 011-03302-00 S/N 3E6024107	2.3	57.05
GMC 605 P/N 011-02967-02 S/N 384200514	2.19	53.70
GSA 87 P/N 011-03780-00 S/N 4JM000957 (Roll)	2.03	113.50
GSA 87 P/N 011-03780-00 S/N 4JM000916 (Pitch)	2.03	211.80
GSA 87 P/N 011-03780-00 S/N 4JM000971 (Trim)	2.03	210.10
GSA 87 P/N 011-03780-00 S/N 4JM000969 (Yaw)	2.03	212.40
GMU 44 P/N 011-00870-10 S/N 1CM026416	0.35	193.75
GTP 59 AOA Probe P/N 011-00978-00 S/N 47942346	0.3	91.50

----------END----------

☐ Additional Sheets Are Attached

US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

OMB No. 2120-0020
Exp: 5/31/2018

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a))

1. Aircraft	Nationality and Registration Mark: N4260S	Serial No.: E-959	
	Make: BEECHCRAFT	Model: A36	Series:
2. Owner	Name *(As shown on registration certificate)*: Executive Airshare LLC	Address *(As shown on registration certificate)*: Address 8345 Lenexa Dr STE 120; City Lenexa; State KS; Zip 66214-1654; Country United States	

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
☐	☒	AIRFRAME	BEECHCRAFT	*(As described in Item 1 above)*	E-959
☐	☐	POWERPLANT			
☐	☐	PROPELLER			
☐	☐	APPLIANCE	Type / Manufacturer		

6. Conformity Statement

A. Agency's Name and Address
Name: Executive Airshare
Address: 3406 N Jabara Rd
City: Wichita State: KS
Zip: 67226 Country: United States

B. Kind of Agency
- ☐ U. S. Certificated Mechanic
- ☐ Foreign Certificated Mechanic
- ☒ Certificated Repair Station
- ☐ Certificated Maintenance Organization
- ☐ Manufacturer

C. Certificate No.: XMXR954Y

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B ☐

Signature/Date of Authorized Individual: 6/22/2017

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ Approved ☐ Rejected

BY								
	☐	FAA Flt. Standards Inspector	☐	Manufacturer	☐	Maintenance Organization	☐	Persons Approved by Canadian Department of Transport
	☐	FAA Designee	X	Repair Station	☐	Inspection Authorization		Other *(Specify)*

Certificate or Designation No. XMXR954Y

Signature/Date of Authorized Individual: 6/22/2017

FAA Form 337 (10/06)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N4260S	6/22/2017
Nationality and Registration Mark	Date

Removed N.L.G. shimmy dampener assembly P/N 35-825145 S/N N/A at fuselage station 8.25. Installed New Lord N.L.G. shimmy dampener assembly P/N SE1076-1 S/N LK01555 at fuselage station 8.25 I.A.W. Lord Corporation Supplemental Type Certificate number SA00609DE.

Instructions for Continued Airworthiness (ICA) to be accomplished I.A.W. Lord Bulletin number SM-6527 Rev. 1.

Weight and Balance No Change.**********END**********

☐ Additional Sheets Are Attached

US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

OMB No. 2120-0020
Exp: 5/31/2018

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a))

1. Aircraft	Nationality and Registration Mark: N4260S	Serial No.: E-959	
	Make: Beechcraft	Model: Bonanza	Series: A36
2. Owner	Name *(As shown on registration certificate)*: Executive Airshare Corp	Address *(As shown on registration certificate)*: Address 1600 S Airport Rd; City Wichita; State Kansas; Zip 67209-1945; Country United States of America	

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
☐	☒	AIRFRAME	Beechcraft	*(As described in Item 1 above)*	E-959
☐	☐	POWERPLANT			
☐	☐	PROPELLER			
☐	☐	APPLIANCE	Type / Manufacturer		

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency		
Name: Scott Hobbs	☒ U. S. Certificated Mechanic	☐ Manufacturer	
Address: 1600 S. Airport Rd	☐ Foreign Certificated Mechanic	C. Certificate No.	
City: Wichita; State: Kansas	☐ Certificated Repair Station	2761933	
Zip: 67209-1945; Country: United Staes of America	☐ Certificated Maintenance Organization		

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B ☐

Signature/Date of Authorized Individual: 4-15-2016

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☐ Approved ☐ Rejected

BY								
	☐	FAA Flt. Standards Inspector	☐	Manufacturer	☐	Maintenance Organization	☐	Persons Approved by Canadian Department of Transport
	☐	FAA Designee	☐	Repair Station	X	Inspection Authorization		Other *(Specify)*

Certificate or Designation No. 2761933

Signature/Date of Authorized Individual: 4-15-2016

FAA Form 337 (10/06)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N4260S	4-15-2016
Nationality and Registration Mark	Date

Removed standby generator power system in accordance with the A36 Shop Manual Section 13 and AC 43.13-1B chapters 4,9,11 & 12. Pilot's Operating Handbook supplement for standby generator system has been removed. Operational check of remaining avionics and systems were satisfactory. Aircraft weight and balance has been revised. End----------

☐ Additional Sheets Are Attached

US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

OMB No. 2120-0020
Exp: 5/31/2018

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. §44701). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a))

1. Aircraft	Nationality and Registration Mark: N4260S	Serial No.: E-959	
	Make: Beechcraft	Model: Bonanza	Series: A36
2. Owner	Name *(As shown on registration certificate)*: Executive Airshare Corp	Address *(As shown on registration certificate)*: Address: 1600 S Airport Rd; City: Wichita; State: Kansas; Zip: 67209-1945; Country: United States of America	

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial No.
☐	☒	AIRFRAME	Beechcraft	*(As described in Item 1 above)*	E-959
☐	☐	POWERPLANT			
☐	☐	PROPELLER			
☐	☐	APPLIANCE	Type / Manufacturer		

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	
Name: Scott Hobbs	X U. S. Certificated Mechanic	☐ Manufacturer
Address: 1600 S. Airport Rd	☐ Foreign Certificated Mechanic	C. Certificate No.
City: Wichita State: Kansas	☐ Certificated Repair Station	2761933
Zip: 67209-1945 Country: United Staes of America	☐ Certificated Maintenance Organization	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B ☐

Signature/Date of Authorized Individual: 4-15-2016

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☐ Approved ☐ Rejected

BY							
	☐	FAA Flt. Standards Inspector	☐	Manufacturer	☐	Maintenance Organization	☐ Persons Approved by Canadian Department of Transport
	☐	FAA Designee	☐	Repair Station	X	Inspection Authorization	Other *(Specify)*

Certificate or Designation No. 2761933

Signature/Date of Authorized Individual: 4-15-2016

FAA Form 337 (10/06)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N4260S	4-15-2016
Nationality and Registration Mark	Date

Removed Northstar M1 Loran C system from aircraft in accordance with AC 43.13-1B chapters 4,9,11 & 12. Operational check of remaining avionics and systems were satisfactory. Aircraft weight and balance has been revised. End----------

☐ Additional Sheets Are Attached

FAA Form 337 (10/06)

U.S Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only

Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed $1,000 for each such violation (Section 901 Federal Aviation Act 1958)

1. Aircraft	Make Beechcraft	Model A36
	Serial No. E-959	Nationality and Registration Mark N4260S
2. Owner	Name *(As shown on registration certificate)* Executive Airshare Corp.	Address *(As shown on registration certificate)* 1600 S. Airport Rd. Wichita, KS 67209-1945

3. For FAA Use Only

4. Unit Identification / **5. Type**

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~~~~~~~~~~~~~~~~*(As described in item 1 above)*~~~~~~~~~~~~~~~~~~~~				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
Executive Airshare 1600 S. Airport Rd. Wichita, KS 67209-1945	☐ U.S. Certified Mechanic ☐ Foreign Certified Mechanic ☒ Certified Repair Station ☐ Manufacturer	XMXR954Y

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date	Signature of Authorized Individual
10-29-08	Ross O Stark

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ APPROVED ☐ REJECTED

BY							
		FAA Flt Standards Inspector		Manufacturer		Inspection Authorization	Other *(Specify)*
		FAA Designee	X	Repair Station		Person Approved by Transport Canada Airworthiness Group	

Date of Approval or Rejection	Certificate or Designation No.	Signature of Authorized Individual
10-29-08	XMXR954Y	Ross O Stark

FAA Form 337 (12-88)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Installed Concorde sealed lead-acid battery P/N RG-35AXC S/N 40259202 I.A.W. Concorde battery STC SA00638WI.

No changes to weight and balance.

**********END**********

☐ Additional Sheets Are Attached

U.S.GPO:1989-0-663-171

U.S Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only

Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed $1,000 for each such violation (Section 901 Federal Aviation Act 1958)

1. Aircraft	Make Beechcraft	Model A36
	Serial No. E-959	Nationality and Registration Mark N4260S
2. Owner	Name *(As shown on registration certificate)* Executive Airshare Corp.	Address *(As shown on registration certificate)* 1600 S. Airport Rd Wichita, KS 67209-1945

3. For FAA Use Only

4. Unit Identification				**5. Type**	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~~~~~~~~~~~~~~~~~~*(As described in item 1 above)*~~~~~~~~~~~~~~~~~~~~~~				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
Executive Airshare 1600 S. Airport Rd Wichita, KS 67209-1945	☐ U.S. Certified Mechanic ☐ Foreign Certified Mechanic ☒ Certified Repair Station ☐ Manufacturer	XMXR954Y

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date	Signature of Authorized Individual
6-2-08	Ross O Stark

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ APPROVED ☐ REJECTED

BY							
		FAA Flt Standards Inspector		Manufacturer		Inspection Authorization	Other *(Specify)*
		FAA Designee	X	Repair Station		Person Approved by Transport Canada Airworthiness Group	

Date of Approval or Rejection	Certificate or Designation No.	Signature of Authorized Individual
6-2-08	XMXR954Y	Ross O Stark

FAA Form 337 (12-88)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Installed Brackett Aero Filter Assembly P/N BA-7210 I.A.W. Brackett Aero Filters STC SA71GL.

Installed D'Shannon Engine Baffle Kit IO-550/520 Canted P/N 2441052 S/N BAF040808-001 I.A.W. D'Shannon Products STC SA368CH.

**********END**********

☐ Additional Sheets Are Attached

U.S.GPO:1989-0-663-171

U.S. Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only

Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed $ 1000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make BEECH	Model A36
	Serial No. E-959	Nationality and Registration Mark N4260S
2. Owner	Name (As shown on registration certificate) AJ Acid	Address (As shown on registration certificate) P.O. Box 758 225 Franklin Ave. Blackwell, OK 74631 USA

3. For FAA Use Only

4. Unit Identification / **5. Type**

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	*(As described in Item 1 above)*				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency		C. Certificate No.
BEVAN-RABELL, INC. 1880 AIRPORT ROAD WICHITA, KS 67209 KG2R951K		U. S. Certified Mechanic	KG2R951K
		Foreign Certified Mechanic	
	X	Certified Repair Station	
		Manufacturer	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U. S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date	Signature of Authorized Individual
21-July-2006	Kent McIntyre, Insp.

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ APPROVED ☐ REJECTED

BY							Other (Specify)
		FAA Flt. Standards Inspector		Manufacturer		Inspection Authorization	
		FAA Designee	X	Repair Station		Person Approved by Transport Canada Airworthiness Group	

Date of Approval or Rejection	Certificate or Designation No.	Signature of Authorized Individual
21-July-2006	CRS KG2R951K	Kent McIntyre, Insp.

FAA Form 337 (12-88)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record.
An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Installed Stec System 30 ALT Altitude Hold System Model ST-834, in accordance with STC SA09376AC-D, Bulletin No. 934, MDL No 921106. All work done using AC43.13-1B, Chapters 11 & 12 and AC43.13-2A, Sections 2 & 3 as a reference. Aircraft Weight and Balance and Equipment list amended as required. An FAA Approved Flight Manual Supplement has been included in the aircraft records.

Instructions for Continued Airworthiness: The System ALT30 should be inspected and maintained in accordance with Stec Instructions Form 86240, Part No. 81356, REv. 1, Page 1-3.

THE END.

☐ ADDITIONAL SHEETS ARE ATTACHED

* U.S.GPO:1991-0-568-012/40035

FAA

US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only
Office Identification CE-05 ST

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed $1,000 for each violation (Section 901 of Federal Aviation Act of 1958).

1. Aircraft	Make: BEECH	Model: A36
	Serial No.: E-959	Nationality and Registration Mark: N4260S
2. Owner	Name *(As shown on registration certificate)*: DE VINCENTIS THOMAS M DE VINCENTIS ROSALINDA T	Address *(As shown on registration certificate)*: 406 N BROOKLINE DR COLUMBIA MO 65203-5606

3. For FAA Use Only

4. Unit Identification				**5. Type**	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~~~~~~ *(As described in Item 1 above)* ~~~~~~~~~~				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency		C. Certificate No.
Central Missouri Aviation Inc. 11200 S. Airport Rd Columbia Mo 65201		U.S. Certificated Mechanic	AKVR148C
		Foreign Certificated Mechanic	
	X	Certified Repair Station	
		Manufacturer	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date	Signature of Authorized Individual
JUNE 11, 2003	James W. Bohn

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ APPROVED ☐ REJECTED

BY							
		FAA Flt. Standards Inspector		Manufacturer		Inspection Authorization	Other *(Specify)*
		FAA Designee	X	Repair Station		Person Approved by Transport Canada Airworthiness Group	

Date of Approval or Rejection	Certificate or Designation No.	Signature of Authorized Individual
JUNE 11, 2003	AKVR148C	James W. Bohn

FAA Form 337 (12-88)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Removed original landing light assembly from the engine nose cowling and installed LoPresti Speed Merchants boom beam bulb, starter, and ballast in accordance with LoPresti STC SA02279AT Master Drawing List, Report Number 43 Revision D dated 08/25/00. Revised equipment list/weight and balance.

CONTINUING AIRWORTHINESS:

1. Check for security of all fasteners and cable ties securing light hardware. Loose or broken fasteners must be replaced or tightened.
2. The bulb/starter, ballast and reflector will be inspected every 100 hours or during an annual inspection.
3. Cracked or broken bulbs, ballast or reflectors must be replaced.
4. Inspect for frayed insulation on wires. Replace if damaged.

**********END**********

☐ Additional Sheets are Attached

United States of America

Department of Transportation -- Federal Aviation Administration

Supplemental Type Certificate

Number SA02279AT

This certificate issued to

LoPresti Speed Merchants
2620 Airport North Drive
Vero Beach, FL. 32960

certifies that the change in the type design for the following product with the limitations and conditions thereto as specified hereon meets the airworthiness requirements of Part 3 *of the* Civil Air *Regulations.*

Original Product Type Certificate Number: 3A15

Make: Beechcraft

Model: 33, 35, 36

Description of Type Design Change:

Installation of a Boom Beam Bulb, Starter and Ballast in accordance with LoPresti Master Drawing List Report Number 43 Revision D dated 08/25/00 or later FAA Approved Revision.

Limitations and Conditions:

This approval shall not be extended to other aircraft of this model on which other previously approved modifications are incorporated unless it is determined by the installer that the interrelationship between this change and any of those other previously approved modifications will produce no adverse affect upon the airworthiness of that airplane. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission.

Authorization for the application of this modification is given only for use on Airplane # 42605

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application: August 15, 2000

Date reissued: June 13, 2001

Date of issuance: February 14, 2001

Date amended: April 5, 2001

By direction of the Administrator

(Signature)

Melvin D. Taylor,
Manager, Atlanta Aircraft
Certification Office

(Title)

Any alteration of this certificate is punishable by a fine of not exceeding $1,000, or imprisonment not exceeding 3 years, or both.

This certificate may be transferred in accordance with FAR 21.47.

U.S Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only

Office Identification
ACE-FSPO-05

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed $1,000 for each such violation (Section 901 Federal Aviation Act 1958)

1. Aircraft	Make: Beech	Model: A36
	Serial No.: E-959	Nationality and Registration Mark: N4260S
2. Owner	Name *(As shown on registration certificate)*: Devincintis, Thomas M Devincintis, Rosalinda T	Address *(As shown on registration certificate)*: 406 N. Brookline Dr. Columbia, MO 65203-5606

3. For FAA Use Only

4. Unit Identification / **5. Type**

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~~~~~~~~~~~~~~~~~~~~~~(As described in item 1 above)~~~~~~~~~~~~~~~~~~~~~~~~~~			☐	☒
POWERPLANT				☐	☐
PROPELLER				☐	☐
APPLIANCE	Type			☐	☐
	Manufacturer			☐	☐

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
Columbia Avionics, Inc. 11200 Airport Road Columbia, MO 65201	☐ U.S. Certified Mechanic ☐ Foreign Certified Mechanic ☒ Certified Repair Station ☐ Manufacturer	C42R104N

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date	Signature of Authorized Individual
7/11/02	Eric Shimp *Eric Shimp*

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ APPROVED ☐ REJECTED

BY						
	☐	FAA Flt Standards Inspector	☐	Manufacturer	☐	Inspection Authorization
	☐	FAA Designee	☒	Repair Station	☐	Person Approved by Transport Canada Airworthiness Group

Other *(Specify)*

Date of Approval or Rejection	Certificate or Designation No.	Signature of Authorized Individual
7/11/02	C42R104N	Eric Shimp *Eric Shimp*

FAA Form 337 (12-88)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

A) Removed KMA-20 Audio Panel and Flightcom 403 Intercom system.

B) Installed Garmin GMA-340 Audio Panel / Intercom system in accordance with Garmin installation manual P/N 190-00149-01, Rev. J, dated 12/01. System installed using manufacturer supplied installation kit and AN hardware.

C) All work performed in accordance with AC43.13-2A Chapter 2.

D) The manufacturer of the installed item does not have any requirements for continued airworthiness, therefore items 1-16 of the ICA checklist do not apply.

E) Ground tests show no interference with installed system or any existing systems.

F) Revised aircraft Weight and Balance and Equipment List to reflect the above revisions.

----------END----------

☐ Additional Sheets Are Attached

FAA

US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only
Office Identification CE05 EJ

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed $1,000 for each violation (Section 901 of Federal Aviation Act of 1958).

1. Aircraft	Make: BEECH	Model: A36
	Serial No.: E-959	Nationality and Registration Mark: N4260S
2. Owner	Name *(As shown on registration certificate)*: DE VINCENTIS THOMAS M DE VINCENTIS ROSALINDA T	Address *(As shown on registration certificate)*: 406 N BROOKLINE DR COLUMBIA MO 65203-5606

3. For FAA Use Only

4. Unit Identification / **5. Type**

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~~~~~~ *(As described in Item 1 above)* ~~~~~~~~~~				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency		C. Certificate No.
Central Misouri Aviation Inc. 11200 S. Airport Rd Columbia Mo 65201		U.S. Certificated Mechanic	AKVR148C
		Foreign Certificated Mechanic	
	X	Certified Repair Station	
		Manufacturer	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date	Signature of Authorized Individual
August 6, 2002	James W. Bahm

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ APPROVED ☐ REJECTED

BY								
		FAA Flt. Standards Inspector		Manufacturer		Inspection Authorization	Other *(Specify)*	
		FAA Designee	X	Repair Station		Person Approved by Transport Canada Airworthiness Group		

Date of Approval or Rejection	Certificate or Designation No.	Signature of Authorized Individual
August 6, 2002	AKVR148C	James W. Bahm

FAA Form 337 (12-88)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Removed McCauley propeller, Model 3A32C76-SALMR and installed Hartzell propeller Model PHC-C3YF-1RF/F8068 inaccordance with D'Shannon Products STC SA01548CH . **********END**********

☐ Additional Sheets are Attached

United States of America

Department of Transportation -- Federal Aviation Administration

Supplemental Type Certificate

Number SA01548CH

This certificate issued to

D'Shannon Products
13610 Glader Lane
Lindstrom, MN 55045

"PROPRIETARY"
Information contained herein is considered Trade Secret. Copies of this document may not be given to any person unless written permission is obtained from D'Shannon Products.

certifies that the change in the type design, for the following product with the limitations and conditions thereto as specified herein meets the airworthiness requirements of Part 3 of the Civil Air Regulations.
See Aircraft Specification No. 3A15 for complete certification basis.

Original Product Type Certificate Number: 3A15

Make: Raytheon Aircraft Company (Beech)

Model: 35-33, 35-A33, 35-B33, 35-C33, 35-C33A, E33, E33A, E33C, F33, F33A, F33C, G33, J35, K35, M35, N35, P35, S35, V35, V35A, V35B, 36, A36

Description of Type Design Change:
Installation of Hartzell Model PHC-(C,L)3YF-1RF/F8068(-2) three bladed propeller on Raytheon Aircraft Company (Beech) aircraft equipped with Continental six-cylinder engines in accordance with D'Shannon Products Installation Instructions DSP-043001A, Initial Release, dated April 30, 2001, or later FAA approved revision.

Limitations and Conditions:

1. Compatibility of this design change with previously approved modifications must be determined by the installer.
2. "-L" hub is eligible on 35 models S/N D-5331 through D-7309 and all 33 models S/N CD-1 through CD-1254 to provide adequate nosebowl/backplate clearance. All other eligible models use "-C" hub.
3. On Models J35, K35, M35, N35 and P35, blades F8068-2 (80" dia.) eligible only.
4. Check aircraft Weight and Balance.
5. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission.

This certificate and the approved data which is the basis for approval shall remain in effect until surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application: April 30, 2001

Date of issuance: October 9, 2001

Date reissued:

Date amended:

By direction of the Administrator

(Signature)

Thaddeus D. Krolicki Jr.
Manager, Propulsion Branch
Chicago Aircraft Certification Office

(Title)

Any alteration of this certificate is punishable by a fine of not exceeding $1,000, or imprisonment not exceeding 3 years, or both.

FAA FORM 8110-2 (10-68) PAGE 1 OF 1 PAGES

This certificate may be transferred in accordance with FAR 21.47.

APR 0 1 2002 REC

US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only

Office Identification
GL03

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This form is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed $1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make: Beechcraft	Model: A36
	Serial No.: E-959	Nationality and Registration Mark: N4260S
2. Owner	Name *(As shown on registration certificate)*: Thomas M. De Vincentis	Address *(As shown on registration certificate)*: 406 N. Brookline Dr. Columbia, MO 65203-5606

3. For FAA Use Only

4. Unit Identification / **5. Type**

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	*(As described in item 1 above)*				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type / Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
J.A. Air Center DuPage Airport West Chicago, IL 60185	U.S. Certificated Mechanic Foreign Certificated Mechanic X Certificated Repair Station Manufacturer	RATING: RADIO CLASS 1, 2 & 3 NF2R029L

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date	Signature of Authorized Individual
March 29, 2002	Bruce P. Rebechini

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is [X] APPROVED [ ] REJECTED

BY							Other (Specify)
		FAA Flt. Standards Inspector		Manufacturer		Inspection Authorization	
		FAA Designee	X	Repair Station		Person Approved by Transport Canadian Airworthiness Group	

Date of Approval or Rejection	Certificate or Designation No.	Signature of Authorized Individual
3/29/02	NF2RO29L	Bruce Rebechini

FAA Form 337 (12-88)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N4260S E-959 A36

**INSTALL UGB-16 ULTIMATE BAR GRAPH ENGINE ANALYZER-
REMOVED THE COMPLETE PREVIOSLY INSTALLED SMART SCANER SYSTEM
P/N SR-8, S/N 6099 INSTALLED 1-26-89 UNDER STC SA4302NM.**

INSTALLED UGB-16 ENGINE ANALYZER S/N 53090 WITH VOLTAGE MODULE P/N FM-VA JO, S/N 053091, AND VACUUM MODULE P/N FM-GYRO, S/N 053092 AS PER STC SA000680SE INSTALLATION OF ELECTRINICS INTERNATIONAL INC. MODEL UGB-16 BAR GRAPH ENGINE ANALYZER IN ACCORDANCE WITH ELECTRONICS INTERNATIONAL INC. MASTER DRAWING LIST MDL 1111982, REVISIONS "NEW", DATED NOVEMBER 11, 1998, AND INSTALLATION INSTRUCTIONS II 1111981, DATED NOVEMBER 11, 1998, OR LATER FAA APPROVED REVISIONS. REVISED THE AIRCRAFT WEIGHT AND BALANCE AND EQUIPMENT LIST.

----------END----------

☐ Additional Sheets Are Attached

U.S. GOVERNMENT PRINTING OFFICE: 1992-769-012/60157

United States of America

Department of Transportation—Federal Aviation Administration

Supplemental Type Certificate

Number **SA00680SE**

This certificate, issued to **Electronics International**
2092 NW Aloclek #523
Hillsboro, OR 97124

*certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part * of the * Regulations.*

Original Product—Type Certificate Number: *See attached FAA Approved Model List (AML) No. SA00680SE for a list of the approved airplane models and applicable airworthiness regulations.

Make:

Model:

Description of the Type Design Change: Installation of Electronics International Inc. Model UBG-16, or UBG-16T bar graph engine analyzer in accordance with Electronics International Inc. Master Drawing List MDL 1111982, Revision "New", dated November 11, 1998, and Installation Instructions II 1111981, dated November 11, 1998, or later FAA approved revisions.

NOTE: The UBG-16 and UBG-16T listed here are designed as advisory engine analyzers, and are not to be used as primary engine instruments.

Limitations and Conditions: Approval of this change in type design applies to the aircraft models listed on the AML only. This approval should not be extended to other aircraft of these models on which other previously approved modifications are incorporated unless it is determined that the relationship between this change and any of those other previously approved modifications, including changes in type design, will introduce no adverse effect upon the airworthiness of that aircraft. A copy of this Certificate and AML No. SA00680SE must be maintained as part of the permanent records for the modified aircraft.

If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission.

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application: November 12, 1998

Date of issuance: March 10, 1999

Date reissued:

Date amended:

By direction of the Administrator

(Signature)

Acting Manager, Seattle
Aircraft Certification Office

(Title)

Any alteration of this certificate is punishable by a fine of not exceeding $1,000, or imprisonment not exceeding 3 years, or both.

This certificate may be transferred in accordance with FAR 21.47.

FAA FORM 8110-2(10-68)

US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only
Office Identification ACE-FSPO-05

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed $1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make: Beech	Model: A36
	Serial No.: E-959	Nationality and Registration Mark: N4260S
2. Owner	Name *(As shown on registration certificate)*: Devincentis, Thomas M Devincentis, Rosalinda T	Address *(As shown on registration certificate)*: 406 N. Brookline Dr. Columbia, MO 65203-5606

The data identified herein complies with the applicable airworthiness requirements and is approved for the above described aircraft, subject to conformity inspection by a person authorized in FAR Part 43, Section 43.7.

3. For FAA Use Only

CE05 05/20/99 [signature]
District Office Date Signature of FAA Inspector

4. Unit Identification / **5. Type**

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	(As described in Item 1 above)				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
Columbia Avionics, Inc. 11200 Airport Road Columbia, MO 65201	[ ] U.S. Certificated Mechanic [ ] Foreign Certificated Mechanic [X] Certificated Repair Station [ ] Manufacturer	C42R104N

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date	Signature of Authorized Individual
6/3/99	Lance Fox [signature]

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is [X] APPROVED [ ] REJECTED

BY							Other *(Specify)*
	[ ]	FAA Flt. Standards Inspector	[ ]	Manufacturer	[ ]	Inspection Authorization	
	[ ]	FAA Designee	[X]	Repair Station	[ ]	Person Approved by Transport Canada Airworthiness Group	

Date of Approval or Rejection	Certificate or Designation No.	Signature of Authorized Individual
6/3/99	C42R104N	Lance Fox [signature]

FAA Form 337 (12-88)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

A) Removed Mac1700 Nav/Comm system, KN-77 VOR/LOC Converter and KN-65 DME System.

B) Installed Garmin GNS-430 GPS\COMM\NAV system with GA-56 GPS antenna, TSO C129 (A1) C40c, C36e, C34e, C37d and C38d. System installed in accordance withGarmin installation manual P/N 190-00140-02, revision D, dated March 18, 1999 and AC20-138. System installed using manufacturer's supplied installation kits, P/N 011-00351-00 and010-10040-01. GPS system coupled to the pilot's HSI and to the Autopilot via the NAV/GPS selector switch on the face of the GNS-430 unit and internal relays as specified in the GNS-430 installation manual. All required and appropriate annunciators installed.

C) Placard "GPS FOR VFR USE ONLY" placed in clear view of pilot.

D) Installed Bendix/King KN-64 DME System in accordance with Bendix/King installation manual, P/N 066-00144-0006, revision 6, dated April, 1997. System installed using manufacturer's supplied installation kit and existing antenna.

E) Ground and flight tests show no interference with installed system or any existing systems.

E) The Aircraft Weight and Balance and Equipment List has been revised to reflect the above installation.

----------END----------

☐ Additional Sheets Are Attached

US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only
Office Identification
ACE-FSDO-05

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed $1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make: Beech	Model: A36
	Serial No.: E-959	Nationality and Registration Mark: N4260S
2. Owner	Name (*As shown on registration certificate*): Devincentis, Thomas M Devincentis, Rosalinda T	Address (*As shown on registration certificate*): 406 N. Brookline Dr. Columbia, MO 65203-5606

The data identified herein complies with the applicable airworthiness requirements and is approved for the above described aircraft, subject to conformity inspection by a person authorized in FAR Part 43, Section 43.7.

CE05 05/20/99 [signature]
District Office Date Signature of FAA Inspector

3. For FAA Use Only

AIRPLANE FLIGHT MANUAL SUPPLEMENT IS COMPATIBLE WITH THIS AIRCRAFT

4. Unit Identification / **5. Type**

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	(*As described in Item 1 above*)				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
Columbia Avionics, Inc. 11200 Airport Road Columbia, MO 65201	[ ] U.S. Certificated Mechanic [ ] Foreign Certificated Mechanic [X] Certificated Repair Station [ ] Manufacturer	C42R104N

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date	Signature of Authorized Individual
MAY 20 1999	Lance Fox [signature]

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ APPROVED ☐ REJECTED

BY							Other (*Specify*)
		FAA Flt. Standards Inspector		Manufacturer		Inspection Authorization	
		FAA Designee	X	Repair Station		Person Approved by Transport Canada Airworthiness Group	

Date of Approval or Rejection	Certificate or Designation No.	Signature of Authorized Individual
6/5/99	C42R104N	Lance Fox [signature]

FAA Form 337 (12 88)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

The Garmin GNS-430 Communication/Navigation System with GA-56 GPS Antenna previously installed on FAA Form 337 dated 6/3/99 for VFR use only has been re-evaluated.

A) 1. The system meets the requirements of TSO-C129 A(1). A ground and flight test has been performed.

2. The system was found to meet the accuracy requirements of AC20-138.

3. The GNS-430 GPS receiver is mounted in the instrument panel in clear view of the pilot. The GA-56 GPS antenna is mounted on top of the fuselage.

4. The system is selectable to the pilot's HSI and to the autopilot via the NAV/GPS button on the face of the GNS-430 unit..

5. All appropriate and required annunciators are installed.

B) The Aircraft Weight and Balance and Equipment List has been updated and Aircraft Flight Manual Supplement completed.

C) System meets the accuracy requirements of AC91-49 and AC120-33.

D) System is approved for VFR/IFR enroute, terminal and approach navigation (GPS, LORAN-C, VOR, VOR-DME, TACAN, NDB, NDB-DME, RNAV) operation within the US National Airspace System.

E) The Flight Manual Supplement dated MAY 20 1999 must be available and used when Flight Navigation is predicated on the use of the Garmin GNS-430 GPS System.

----------END----------

☐ Additional Sheets Are Attached

US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only
Office Identification
ACE-FSDO-05

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed $1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make: BEECH	Model: A36
	Serial No.: E-959	Nationality and Registration Mark: N4260S
2. Owner	Name *(As shown on registration certificate)*: DE VINCENTIS THOMAS M DE VINCENTIS ROSALINDA T	Address *(As shown on registration certificate)*: 406 N BROOKLINE DR COLUMBIA MO 65203-5606

3. For FAA Use Only

4. Unit Identification / **5. Type**

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	*(As described in Item 1 above)*				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
COLUMBIA AVIONICS 11200 AIRPORT ROAD COLUMBIA MO 65201	U.S. Certificated Mechanic Foreign Certificated Mechanic X Certificated Repair Station Manufacturer	C42R104N

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date	Signature of Authorized Individual
06 03 99	ERIC SHIMP Eric Shimp

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☐ APPROVED X ☐ REJECTED

BY							Other *(Specify)*
		FAA Flt. Standards Inspector		Manufacturer		Inspection Authorization	
		FAA Designee	X	Repair Station		Person Approved by Transport Canada Airworthiness Group	

Date of Approval or Rejection	Certificate or Designation No.	Signature of Authorized Individual
06 03 99	C42R104N	Eric Shimp

FAA Form 337 (12-88)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

A) Removed King KX-170B/MAC 1700 Nav/Comm system, King KN-77 VOR/LOC Converter, King KN-65A DME system with KI-266 DME Indicator and DME antenna, RSU-21 switching relay.

B) Installed Garmin GNS-430 GPS\COMM\NAV system with GA-56 GPS antenna, TSO C129 (A1) C40c, C36e, C34e, C37d and C38d. System installed in accordance with Garmin installation manual P/N 190-00140-02, rev. D, dated March 1999 and AC20-138. System installed using manufacturer's supplied installation kits, P/N 011-00351-00 and 010-10040-01. GPS system coupled to the pilot's HSI and to the Autopilot via the NAV/GPS selector switch on the face of the GNS-430 unit and internal relays as specified in the GNS-430 installation manual. All required and appropriate annunciators installed.

C) Installed Ameri-King AK-550-6 power converter, TSO C71, in accordance with Ameri-King installation manual P/N IM-550. System installed using manufacturer supplied installation kit P/N 55001.

D) Installed Bendix/King KN-64 DME system with KA-60 antenna, TSO C66a and C74c, in accordance with Bendix/King installation manual P/N 006-05144-006, rev. 6, dated June 1997. System installed using manufacturer supplied installation kit P/N 050-01611-0000.

E) All systems installed with reference to AC 43.13-2A, Chapters 1,2, and 3.

F) Placard "GPS FOR VFR USE ONLY" placed in clear view of pilot.

G) Instructions for Continued Airworthiness:
   1) There are no life limited parts or Continued Airworthiness Requirements for the components installed above.

H) Ground and flight tests show no interference with installed system or any existing systems.

I) The Aircraft Weight and Balance and Equipment List has been revised to reflect the above installation.

----------END----------

☐ Additional Sheets Are Attached

US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only
Office Identification
ACE-FSDO-05

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed $1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make: BEECHCRAFT	Model: A36
	Serial No.: E959	Nationality and Registration Mark: N4260S
2. Owner	Name *(As shown on registration certificate)*: DEVINCENTIS, THOMAS M. DEVINCENTIS, ROSALINDA T.	Address *(As shown on registration certificate)*: 406 N BROOKLINE DR. COLUMBIA, MO 65203

3. For FAA Use Only

4. Unit Identification / **5. Type**

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	*(As described in Item 1 above)*				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
COLUMBIA AVIONICS INC 11200 AIRPORT ROAD COLUMBIA, MO 65201	U.S. Certificated Mechanic Foreign Certificated Mechanic X Certificated Repair Station Manufacturer	C42R104N

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date	Signature of Authorized Individual
9/21/98	ERIC SHIMP Eric Shimp

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is X APPROVED ☐ REJECTED.

BY							
		FAA Flt. Standards Inspector		Manufacturer		Inspection Authorization	Other *(Specify)*
		FAA Designee	X	Repair Station		Person Approved by Transport Canada Airworthiness Group	

Date of Approval or Rejection	Certificate or Designation No.	Signature of Authorized Individual
9/21/98	C42R104N	ERIC SHIMP Eric Shimp

FAA Form 337 (12-88)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

A) Installed BF Goodrich WX-950 Stormscope system in accordance with manufacturer's installation manual P/N 009-10950-001, Revision 00, dated Feb 9 1996. System installed using manufacturer supplied installation kit P/N 817-10950-001.

B) System installed in accordance with AC43-13-2A, Chapter 2 and Chapter 3.

C) Ground and flight tests show no interference with installed system or any existing systems.

D) The Aircraft Weight and Balance and Equipment List has been revised to reflect the above installation.

----------END----------

☐ Additional Sheets Are Attached

*U.S.GPO:1994-568-012/00019

RWA

U.S. DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
Budget Bureau No. 04-R060.1

FOR FAA USE ONLY

OFFICE IDENTIFICATION: SEA-PSO

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.

1. AIRCRAFT	MAKE: Beechcraft	MODEL: A36
	SERIAL NO.: E959	NATIONALITY AND REGISTRATION MARK: N4260S
2. OWNER	NAME (As shown on registration certificate): TST Systems	ADDRESS (As shown on registration certificate): P.O. Box 2208, Seattle, Wa 98111

3. FOR FAA USE ONLY

4. UNIT IDENTIFICATION				5. TYPE	
UNIT	MAKE	MODEL	SERIAL NO.	REPAIR	ALTERATION
AIRFRAME	(As described in item 1 above)				X
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE / MANUFACTURER				

6. CONFORMITY STATEMENT

A. AGENCY'S NAME AND ADDRESS	B. KIND OF AGENCY	C. CERTIFICATE NO.
Arthur P. Loring, Jr. 13221 51st Pl W Edmonds, WA 98026	X U.S. CERTIFICATED MECHANIC FOREIGN CERTIFICATED MECHANIC CERTIFICATED REPAIR STATION MANUFACTURER	A&P 554 38 7332

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

DATE	SIGNATURE OF AUTHORIZED INDIVIDUAL
June 14, 1997	Arthur P. Loring Jr

7. APPROVAL FOR RETURN TO SERVICE

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ APPROVED ☐ REJECTED

	FAA FLT. STANDARDS INSPECTOR		MANUFACTURER	X	INSPECTION AUTHORIZATION	OTHER (Specify)
	FAA DESIGNEE		REPAIR STATION		CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	

DATE OF APPROVAL OR REJECTION	CERTIFICATE OR DESIGNATION NO.	SIGNATURE OF AUTHORIZED INDIVIDUAL
6/17/97	537309707	[illegible]

FAA Form 337 (7-67) (8320)

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED *(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)*

Installed Aerosafe Guardian I standby pneumatic pressure system in accordance with STC SA4835 SW, Aerosafe drawing 820601 and AC43-13 Chapter 5 paragraphs 227B, 227E, 230A(1), 230A(2) and 231A as well as Chapter 11, paragraphs 445 thru 452 of section 2, paragraphs 482 & 483 of section 5 and paragraphs 514 thru 519 of section 7. Weight & balance recalculated and electrical load recalculated in accordance with AC43-13 Chapter 11, section 2 paragraphs 424, 425 & 426.

— end —

☐ ADDITIONAL SHEETS ARE ATTACHED

☆U.S. GOVERNMENT PRINTING OFFICE: 1986-659-121/40260

FAA AIRCRAFT REGISTRY
CAMERA NO. 2-N DATE: 12/15/90

0 0 0 0 0 0 0 0 4 7 7

13-1

DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION

FAA APPROVAL NOT REQUIRED

THIS FORM SERVES TWO PURPOSES
PART I acknowledges the recording of a security conveyance covering the aircraft shown below.
PART II is a suggested form of release which may be used to release the collateral from the terms of this conveyance.

U50121

CONVEYANCE RECORDED
Dec 3 12 07 PM '90
FEDERAL AVIATION ADMINISTRATION

PART I – CONVEYANCE RECORDATION NOTICE

NAME (list name first) OF DEBTOR
T. ROBERT MCCULLY, JAMES K. HOWERTON
GARY C. YANT

NAME and ADDRESS OF SECURED PARTY/ASSIGNEE
CROCKER NATIONAL BANK
1001 STATE STREET
SANTA BARBARA, CA 93102

NAME OF SECURED PARTY'S ASSIGNOR (if assigned)

Do Not Write In This Block FOR FAA USE ONLY

FAA REGISTRATION NUMBER	AIRCRAFT SERIAL NUMBER	AIRCRAFT MFR., BUILDER, and MODEL
N4260S	E-959	1977 BEECH BONANZA A36

ENGINE MFR. and MODEL	ENGINE SERIAL NUMBER(S)
BEECH BONANZA	UNKNOWN

PROPELLER MFR. and MODEL	PROPELLER SERIAL NUMBER(S)

SEE RECORDED CONVEYANCE NUMBER N29953 FICHE 1 PAGE 10-1

THE SECURITY CONVEYANCE DATED 12-1-78 COVERING THE ABOVE COLLATERAL WAS RECORDED BY THE FAA AIRCRAFT REGISTRY ON 2-2-79 AS CONVEYANCE NUMBER N29953

FAA CONVEYANCE EXAMINER

PART II – RELEASE – (This suggested release form may be executed by the secured party and returned to the FAA Aircraft Registry when terms of the conveyance have been satisfied. See below for additional information.)

THE UNDERSIGNED HEREBY CERTIFIES AND ACKNOWLEDGES THAT HE IS THE TRUE AND LAWFUL HOLDER OF THE NOTE OR OTHER EVIDENCE OF INDEBTEDNESS SECURED BY THE CONVEYANCE REFERRED TO HEREIN ON THE ABOVE-DESCRIBED COLLATERAL AND THAT THE SAME COLLATERAL IS HEREBY RELEASED FROM THE TERMS OF THE CONVEYANCE. ANY TITLE RETAINED IN THE COLLATERAL BY THE CONVEYANCE IS HEREBY SOLD, GRANTED, TRANSFERRED, AND ASSIGNED TO THE PARTY WHO EXECUTED THE CONVEYANCE, OR TO THE ASSIGNEE OF SUCH PARTY IF THE CONVEYANCE SHALL HAVE BEEN ASSIGNED: PROVIDED, THAT NO EXPRESS WARRANTY IS GIVEN OR IMPLIED BY REASON OF EXECUTION OR DELIVERY OF THIS RELEASE. WELLS FARGO BANK AS SUCCESSOR IN INTEREST OF CROCKER NATIONAL BANK

DATE OF RELEASE: OCTOBER 31, 1990

(Name of security holder)

SIGNATURE (in ink)

TITLE M. A. RUIZ ASSISTANT VICE PRESIDENT OFFICER

(A person signing for a corporation must be a corporate officer or hold a managerial position and must show his title. A person signing for another should see Parts 47 and 49 of the Federal Aviation Regulations (14 CFR).

This form is only intended to be a suggested form of release, which meets the recording requirements of the Federal Aviation Act of 1958, and the regulations issued thereunder. In addition to these requirements, the form used by the security holder should be drafted in accordance with the pertinent provisions of local statutes and other applicable federal statutes. This form may be reproduced. There is no fee for recording a release. Send to FAA Aircraft Registry, P. O. Box 25504, Oklahoma City, Oklahoma 73125.

ACKNOWLEDGEMENT (If Required By Applicable Local Law):

AC Form 8050-41 (8-77) (0052-00-543-9001)

U.S. GOVERNMENT PRINTING OFFICE 1977-721 089 845

FAA AIRCRAFT REGISTRY
CAMERA NO. 24 DATE 12-12-90
CAMERA NO. 24 DATE

13

VOID

OKLAHOMA CITY
OCT 9 11 33 AM '90
CONVEYANCE
FILED WITH FAA
AIRCRAFT REGISTRY

FAA AIRCRAFT REGISTRY
CAMERA NO. 2/7 DATE STAMP 12 - 12 - 90
CAMERA NO. 2/7 DATE 12-12-90

12-1

0 0 0 0 0 0 0 3 4 1 0

DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION

FOR FAA USE ONLY
DO NOT WRITE

050120

CONVEYANCE RECORDED
Dec 3 12 07 PM '90
FEDERAL AVIATION ADMINISTRATION

THIS FORM SERVES TWO PURPOSES
PART I: [illegible]
PART II is a suggested form of release which may be used to release the collateral from the lien of this conveyance.

PART I — CONVEYANCE RECORDATION NOTICE

NAME (first name first) OF DEBTOR
T. Robert McCully
Gary C Yant James K Homerton

NAME and ADDRESS OF SECURED PARTY/ASSIGNEE
Crocker National Bank
1001 State Street
Santa Barbara, Ca. 93102

NAME OF SECURED PARTY'S ASSIGNEE (if suggested)

Do Not Write In This Block
FOR FAA USE ONLY

FAA REGISTRATION NUMBER	AIRCRAFT SERIAL NUMBER	AIRCRAFT MFR. (SUBLINE) and MODEL
N4260S	E-9[illegible]	1977 Beech Bonanza A 36

ENGINE MFR. and MODEL	ENGINE SERIAL NUMBER(S)
Beech Bonanza	Unknown

PROPELLER MFR. and MODEL	PROPELLER SERIAL NUMBER(S)

THE SECURITY CONVEYANCE DATED 12-1-78 COVERING THE ABOVE COLLATERAL WAS RECORDED BY THE FAA AIRCRAFT REGISTRY ON 2-21-79 AS CONVEYANCE NUMBER N29953

FAA CONVEYANCE EXAMINER

PART II — RELEASE — (This suggested release form may be executed by the secured party and returned to the FAA Aircraft Registry when terms of the conveyance have been satisfied. See below for additional information.)

THE UNDERSIGNED HEREBY CERTIFIES AND ACKNOWLEDGES THAT HE IS THE TRUE AND LAWFUL HOLDER OF THE NOTE OR OTHER EVIDENCE OF INDEBTEDNESS SECURED BY THE CONVEYANCE REFERRED TO HEREIN ON THE ABOVE-DESCRIBED COLLATERAL AND THAT, THE SAME COLLATERAL IS HEREBY RELEASED FROM [illegible] TERMS OF THE CONVEYANCE. ANY TITLE RETAINED IN THE COLLATERAL BY THE CONVEYANCE IS HEREBY S[illegible]ED, TRANSFERRED, AND ASSIGNED TO THE PARTY WHO EXECUTED THE CONVEYANCE, OR TO THE ASS[illegible] OF SUCH PARTY IF THE CONVEYANCE SHALL HAVE BEEN ASSIGNED: PROVIDED, THAT NO EXPRESS WARRANTY [illegible] GIVEN OR IMPLIED BY REASON OF EXECUTION OR DELIVERY OF THIS RELEASE.

This form is only intended to be a suggested form of release, which meets the recording requirements of the Federal Aviation Act of 1958, and the regulations issued thereunder. In addition to these requirements, the form used by the security holder should be drafted in accordance with the pertinent provisions of local statutes and other applicable federal statutes. This form may be reproduced. There is no fee for recording a release. Send to FAA Aircraft Registry, P. O. Box 25504, Oklahoma City, Oklahoma 73125.

ACKNOWLEDGEMENT (If Required By Applicable Local Law)

DATE OF RELEASE: October 27, 1987
AS SUCCESSOR IN INTEREST: Well[illegible] Fargo Bank, NA
(Name of secured party)

SIGNATURE (in ink) [illegible]
TITLE R. L. Sharp Asst. Vice P[illegible]

(A person signing for a corporation must be a corporate officer or hold a managerial position and must show his title. A person signing for another should see Parts 47 and 49 of the Federal Aviation Regulations (14 CFR).

AC Form 8050-41 (8-77) (0052-00-542-0001)

U.S. GOVERNMENT PRINTING OFFICE 1977—771-000/845

FAA AIRCRAFT REGISTRY
CAMERA NO. 2N DATE: 12 - 12 - 90

V O I D

FAA AIRCRAFT REGISTRY
CAMERA NO. DATE: 6 - 5 - 90

US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

RECEIVED FAA DEC 18 1989 VNY FSDO

Form Approved
OMB No. 2120-0020

For FAA Use Only
Office Identification
VNY FSDO

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed $1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make: BEECH	Model: A36
	Serial No.: E-959	Nationality and Registration Mark: N4260S
2. Owner	Name (As shown on registration certificate): BEECHCRAFT N4260S	Address (As shown on registration certificate): 51 N. SAN MARCOS RD B, SANTA BARBARA, CA. 93111

3. For FAA Use Only

4. Unit Identification				5. Type	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	(As described in Item 1 above)				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
GEORGE T KIRSTNER 350 A EAST SANTA MARIA ST. SANTA PAULA, CA. 93060	X U.S. Certificated Mechanic Foreign Certificated Mechanic Certificated Repair Station Manufacturer	AHP 336548575

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date	Signature of Authorized Individual
12-14-89	[signature]

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ APPROVED ☐ REJECTED

BY							
		FAA Flt. Standards Inspector		Manufacturer	X	Inspection Authorization	Other (Specify)
		FAA Designee		Repair Station		Person Approved by Transport Canada Airworthiness Group	

Date of Approval or Rejection	Certificate or Designation No.	Signature of Authorized Individual
12-14-89	336548575	[signature]

FAA Form 337 (12-88)

FAA AIRCRAFT REGISTRY
CAMERA NO. DATE: 6-5-90

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

INSTALLED WALKER ENGINEERING COMPANY AIR/OIL SEPERATOR KIT RW-3000-1 STC# SA-3548-WE AS PER WALKER ENGINEERING COMPANY INSTALLATION INSTRUCTIONS/DRAWINGS RW 3000-1 REV. E DATED 5-20-79

END

☐ Additional Sheets Are Attached

FAA AIRCRAFT REGISTRY
CAMERA NO. 3N DATE: 6-55 990

FAA

RECEIVED FAA
ASW-FSDO-OKC
FEB 0 9 1990

DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
Budget Bureau No. 04-R060.1
FOR FAA USE ONLY
OFFICE IDENTIFICATION
ASW 7500 15

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.

1. AIRCRAFT	MAKE: Fairchild	MODEL: SA227-AC
	SERIAL NO.: AC-640	NATIONALITY AND REGISTRATION MARK: N425MA
2. OWNER	NAME (As shown on registration certificate): Sta[illegible] St[illegible]t Bank & Trust Co.	ADDRESS (As shown on registration certificate): 225 Franklin Street, Boston, MA. 02101

3. FOR FAA USE ONLY The alteration/data identified herein complies with the applicable airworthiness requirements and is approved only for the above described aircraft subject to conformity inspection by a person authorized in FAR 43, section 43.7.

DATE 2/6/90 FAA INSPECTOR-SW-FSDO-67

4. UNIT IDENTIFICATION

UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE: REPAIR	ALTERATION
AIRFRAME	(As described in item 1 above)			X	
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE / MANUFACTURER				

6. CONFORMITY STATEMENT

A. AGENCY'S NAME AND ADDRESS	B. KIND OF AGENCY	C. CERTIFICATE NO.
AAR OKLAHOMA INC. 6611 S. Meridian Okla. City, OK 73159	U.S. CERTIFICATED MECHANIC FOREIGN CERTIFICATED MECHANIC X CERTIFICATED REPAIR STATION MANUFACTURER	JR[illegible]36K

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

DATE	SIGNATURE OF AUTHORIZED INDIVIDUAL
January 19, 1990	Jackie L. Smith 10125

7. APPROVAL FOR RETURN TO SERVICE

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is [X] APPROVED [ ] REJECTED

BY							
	FAA FLT. STANDARDS INSPECTOR		MANUFACTURER		INSPECTION AUTHORIZATION	OTHER (Specify)	
	FAA DESIGNEE	X	REPAIR STATION		CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT		

DATE OF APPROVAL OR REJECTION	CERTIFICATE OR DESIGNATION NO.	SIGNATURE OF AUTHORIZED INDIVIDUAL
Feb. 6, 1990	JR2R936K	[illegible]

FAA Form 337 (7-67) (8320)

VOID

FAA AIRCRAFT REGISTRY
CAMERA NO. 31 DATE: 12-5-90

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED *(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)* N425MA W/O 71991 1-19-90

1. Removed [illegible]in[illegible]or Step Covering and installed Aero-Mat PT# W-071980-83000018 Vinyl a[illegible]" [illegible]olite Foam. Material meet Airworthiness requirements of FAR part 25. [illegible]. Ref; DER 8110-3 dated 1-30-90 for the Vinyl and FAA CRS 202-91 f[illegible] Ensolite.
Weight a[illegible]ance change negligible.

VOID

☐ ADDITIONAL SHEETS ARE ATTACHED

FAA AIRCRAFT REGISTRY
CAMERA NO. 31 DATE: 6-5-98

FAA

US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only
Office Identification
ASW 750 15

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed $1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make: Fairchild	Model: SA227-AC
	Serial No.: AC-	Nationality and Registration Mark: N425MA
2. Owner	Name (As shown on registration certificate): State St[illegible]nk & Trust Co.	Address (As shown on registration certificate): 225 Franklin Street, Boston, MA 02101

3. For FAA Use Only

Unit	4. Identification: Make	Model	Serial No.	5. Type: Repair	Alteration
AIRFRAME	(As described in item 1 above)				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

RECEIVED FAA
JAN 23 1990

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. [illegible] No.
AAR OKLAHOMA INC. 6611 S. Meridian Okla. City, OK 73169	U.S. Certificated Mechanic Foreign Certificated Mechanic X Certificated Repair Station Manufacturer	2R9[illegible]

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the re[illegible]e or [illegible]hments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations [illegible] that t[illegible]formation furnished herein is true and correct to the best of my knowledge.

Date	Signature of Authorized Individual
January 19, 1990	Thomas J. Howard

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ APPROVED ☐ REJECTED

BY							Other (Specify)
		FAA Flt. Standards Inspector		Manufacturer		Inspection Authorization	
		FAA Designee	X	Repair Station		Person Approved by Transport Canada Airworthiness Group	

Date of Approval or Rejection	Certificate or Designation No.	Signature of Authorized Individual
Jan. 19, 1990	JR2R936K	[illegible]

FAA Form 337 (12-88)

VOID

FAA AIRCRAFT REGISTRY
CAMERA NO. 31 DATE: 6-5-90

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished 1-19-90 N425MA W/O 71991

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

1. Removed #2 Comm antenna from lower fuselage Sta. 177.0, installed flush doubler repair, in accordance with SA226/227 Series Structural Repair Manual, 53-90-20, page 67, fig.39.

2. Installed same antenna at upper fuselage Sta. 146.0, weight .5#. Installed in accordance with AAR Oklahoma, Inc. Sketch 06-900118-1 dated 1-18-90 and FAA form 81[illegible]3, [illegible] structural approval dated 1-19-90.

Weight and b[illegible] change negligible.

END

VOID

☐ Additional Sheets Are Attached.

FAA AIRCRAFT REGISTRY
CAMERA NO. 3N DATE: 6-5-90

FAA

U.S. DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
Budget Bureau No. 04-R060.1

FOR FAA USE ONLY

OFFICE IDENTIFICATION: MSP FSDO

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.

1. AIRCRAFT	MAKE: Fairchild	MODEL: SA227-AC
	SERIAL NO.: AC-640	NATIONALITY AND REGISTRATION MARK: N425MA
2. OWNER	NAME (As shown on registration certificate): State Street Bank & Trust Co.	ADDRESS (As shown on registration certificate): 225 Franklin Street Boston, MA. 02101

3. FOR FAA USE ONLY

VOID

4. IDENTIFICATION

UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE: REPAIR	ALTERATION
AIRFRAME	(As shown in item 1 above)				
POWERPLANT		M C	RECEIVED FAA ASW-FSDO-OKC		
PROPELLER		P S	JAN 24 1990		
APPLIANCE	TYPE: Force Flap Assy. / MANUFACTURER: Fairchild	P/N: 27-3600 06	BETHANY, OK / None Listed	X	

6. CONFORMITY STATEMENT

A. AGENCY'S NAME AND ADDRESS	B. KIND OF AGENCY	C. CERTIFICATE NO.
Mesaba Airlines 7501 26th Ave. S Minneapolis, MN 55450	U.S. CERTIFICATED MECHANIC FOREIGN CERTIFICATED MECHANIC CERTIFICATED REPAIR STATION X Air Carrier	14GL7

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

DATE	SIGNATURE OF AUTHORIZED INDIVIDUAL
3-30-89	

7. APPROVAL FOR RETURN TO SERVICE

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ APPROVED ☐ REJECTED

BY	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	INSPECTION AUTHORIZATION	OTHER (Specify)
	FAA DESIGNEE	REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	Air Carrier

DATE OF APPROVAL OR REJECTION	CERTIFICATE OR DESIGNATION NO.	SIGNATURE OF AUTHORIZED INDIVIDUAL
3-30-89	14GL7	

FAA Form 337 (7-67) (8320)

FAA AIRCRAFT REGISTRY
CAMERA NO. 31 DATE: 6-5-90

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED *(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)*

Repaired RH fore flap assy. P/N 27-36000-006 by installing new skin. Used same rivet spacing as original. Assembled IAW structural repair manual [illegible] practices, Chapter 51-30-04 and Chapter 5[illegible]-0[illegible]

[illegible]

VOID

☐ ADDITIONAL SHEETS ARE ATTACHED

☆U.S. GOVERNMENT PRINTING OFFICE: 1986-659-121/40260

FAA AIRCRAFT REGISTRY
CAMERA NO. 3N DATE: 6-5-90

UNITED STATES OF AMERICA
DEPARTMENT OF TRANSPORTATION—FEDERAL AVIATION ADMINISTRATION

STANDARD AIRWORTHINESS CERTIFICATE

1. NATIONALITY AND REGISTRATION MARKS	2. MANUFACTURER AND MODEL	3. AIRCRAFT SERIAL NUMBER	4. CATEGORY
N42[illegible]	Fairchild Aircraft SA227-AC	AC-640	Normal

5. AUTHORITY AND BASIS FOR ISSUANCE
This airworthiness certificate is issued pursuant to the Federal Aviation Act of 1958 and certifies that, as of the date of issuance, the aircraft to which issued has been inspected and found to conform to the type certificate therefor, to be in condition for safe operation, and has been shown to meet the requirements of the applicable comprehensive and detailed airworthiness code as provided by Annex 8 to the Convention on International Civil Aviation except as noted herein.

★ NONE ★

6. TERMS AND CONDITIONS
Unless sooner surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator, this airworthiness certificate is effective as long as the maintenance, preventative maintenance, and alterations are performed in accordance with Parts 43 and 91 of the Federal Aviation Regulations, as appropriate, and the aircraft is registered in the United States.

DATE OF ISSUANCE	FAA REPRESENTATIVE	DESIGNATION NUMBER
REPLACEMENT 3/13/86	[illegible] S. Smith	ACL-01

Any alteration, reproduction, or misuse of this certificate may be punishable by a fine not exceeding $1,000, or imprisonment not exceeding 3 years, or both. THIS CERTIFICATE MUST BE DISPLAYED IN THE AIRCRAFT IN ACCORDANCE WITH APPLICABLE FEDERAL AVIATION REGULATIONS.

FAA Form 8100-2 (7-67) FORMERLY FAA Form 1362 GPO : 1967—O-270-931

VOID

FAA AIRCRAFT REGISTRY
CAMERA NO. 31 DATE: 6-5-99

VOID

CAMERA NO. 3N DATE: 6-5-92

UNITED STATES OF AMERICA
DEPARTMENT OF TRANSPORTATION—FEDERAL AVIATION ADMINISTRATION

STANDARD AIRWORTHINESS CERTIFICATE

1. NATIONALITY AND REGISTRATION MARKS	2. MANUFACTURER AND MODEL DESIGNATION	3. AIRCRAFT SERIAL NUMBER	4. CATEGORY
N4252L	CESSNA 172G	17254321	NORMAL

5. AUTHORITY AND BASIS FOR ISSUANCE: This airworthiness certificate is issued pursuant to the Federal Aviation Act of 1958 and certifies that, as of the date of issuance, the aircraft to which issued has been found to conform to the type certificate therefor, to be in condition for safe operation, and to meet the requirements of the applicable comprehensive and detailed airworthiness code established by Annex 8 to the Convention on International Civil Aviation, except as noted herein.

(Exceptions)

NONE

6. TERMS AND CONDITIONS: Unless sooner surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator, this airworthiness certificate is valid for as long as the maintenance, preventative maintenance, and alterations are performed in accordance with Parts 21, 43, and 91 of the Federal Aviation Regulations, as appropriate, and the aircraft is registered in the United States.

DATE OF ISSUANCE	FAA REPRESENTATIVE	DESIGNATION NUMBER
REPLACEMENT 03-18-66	WILLIAM H. JOHNSON	MSP-FSDO

Any alteration, reproduction, or misuse of this certificate may be punishable by a fine not exceeding $1,000, or imprisonment not exceeding [illegible] or both. THIS CERTIFICATE MUST BE DISPLAYED IN THE AIRCRAFT IN ACCORDANCE WITH APPLICABLE [illegible] AVIATION REGULATIONS.

FAA Form 8100-2 [illegible] FORMERLY FAA FORM 1362

VOID

FAA AIRCRAFT REGISTRY
CAMERA NO. 3 DATE: 6-29-89

DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION

RECEIVED FAA
MAR 13 1989
VNY FSDO

Form Approved
Budget Bureau No. 04-R000.1

FOR FAA USE ONLY
OFFICE IDENTIFICATION

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.

1. AIRCRAFT	MAKE: BEECH	MODEL: A36
	SERIAL NO.: E-959	NATIONALITY AND REGISTRATION MARK: USA N42605
2. OWNER	NAME (As shown on registration certificate): BEECH CRAFT N42605	ADDRESS (As shown on registration certificate): 51 N. SAN MARCOS RD. B SANTA BARBARA, CA 93111

3. FOR FAA USE ONLY

4. UNIT IDENTIFICATION / 5. TYPE

UNIT	MAKE	MODEL	SERIAL NO.	REPAIR	ALTERATION
AIRFRAME	(As described in item 1 above)				X
POWERPLANT	CONTINENTAL	10-520-BA	562564		X
PROPELLER					
APPLIANCE	TYPE / MANUFACTURER		AERR		

6. CONFORMITY STATEMENT

A. AGENCY'S NAME AND ADDRESS	B. KIND OF AGENCY	C. CERTIFICATE NO.
MERCURY AIR CENTER 404 MOFFETT PLACE GOLETA, CA 93117	U.S. CERTIFICATED MECHANIC FOREIGN CERTIFICATED MECHANIC X CERTIFICATED REPAIR STATION MANUFACTURER	401-171

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

DATE	SIGNATURE OF AUTHORIZED INDIVIDUAL
1-26-89	

7. APPROVAL FOR RETURN TO SERVICE

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ APPROVED ☐ REJECTED

BY					
	FAA FLT. STANDARDS INSPECTOR	MANUFACTURER	INSPECTION AUTHORIZATION	OTHER (Specify)	
	FAA DESIGNEE	X REPAIR STATION	CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT		

DATE OF APPROVAL OR REJECTION	CERTIFICATE OR DESIGNATION NO.	SIGNATURE OF AUTHORIZED INDIVIDUAL
1-26-89	401-171	

FAA Form 337 (7-67) ☆ U.S. Government Printing Office 1977—772-845/141 (8320)

FAA AIRCRAFT REGISTRY
CAMERA NO. 3 DATE 6-29-89

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED: (If more space is required attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

INSTALLED ELECTRONICS INTERNATIONAL MODEL SR-8 DIGITAL ~~ENGINE~~ AUTOMATIC ENGINE ANALYZER (THE SMART SCANNER) IN ACCORDANCE WITH INSTALLATION INSTRUCTIONS No. II 012881 DATED 1/28/88 AND STC # SA4302NM. 6 EGT PROBES, 2 CHT PROBES

ELECTRICAL LOAD DOES NOT EXCEED 80% OF CHARGING SYSTEM

EMI CHECK MADE AND NO INTERFERENCE NOTED

WEIGHT AND BALANCE REVISED

NO EXHAUST LEAKS DETECTED IN ENGINE RUN UP

LAST ITEM

☐ ADDITIONAL SHEETS ARE ATTACHED

FAA AIRCRAFT REGISTRY
CAMERA NO. 3N DATE: 4 - 24 - 89

U.S. DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION

RECEIVED
MAR 29 1988
IGR FSDO 5

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
Budget Bureau No. 04-R060.1

FOR FAA USE ONLY
OFFICE IDENTIFICATION
~~WP-FSDO-5~~

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.

1. AIRCRAFT	MAKE: Beech	MODEL: A36
	SERIAL NO.: E-959	NATIONALITY AND REGISTRATION MARK: USA N4260S
2. OWNER	NAME (As shown on registration certificate): Beechcraft N4260S	ADDRESS (As shown on registration certificate): 51 N San Marcos Rd B Santa Barbara, Ca. 93111

3. FOR FAA USE ONLY

4. UNIT IDENTIFICATION

UNIT	MAKE	MODEL	SERIAL NO.	5. TYPE: REPAIR	5. TYPE: ALTERATION
AIRFRAME	(As described in item 1 above)				X
POWERPLANT					
PROPELLER					
APPLIANCE	TYPE / MANUFACTURER				

6. CONFORMITY STATEMENT

A. AGENCY'S NAME AND ADDRESS	B. KIND OF AGENCY	C. CERTIFICATE NO.
Flite-Comm Electronics, Inc. 3271 Airflite Way Long Beach, Ca. 90807	U.S. CERTIFICATED MECHANIC FOREIGN CERTIFICATED MECHANIC X CERTIFICATED REPAIR STATION MANUFACTURER	Radio Class I, II, III(465-20)Limited Instrument, Limited Airframe, Specialized Services

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

DATE	SIGNATURE OF AUTHORIZED INDIVIDUAL
24 March 1988	Todd H. Foster [signature]

7. APPROVAL FOR RETURN TO SERVICE

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ APPROVED ☐ REJECTED

BY							
		FAA FLT. STANDARDS INSPECTOR		MANUFACTURER		INSPECTION AUTHORIZATION	OTHER (Specify)
		FAA DESIGNEE	X	REPAIR STATION		CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT	

DATE OF APPROVAL OR REJECTION	CERTIFICATE OR DESIGNATION NO.	SIGNATURE OF AUTHORIZED INDIVIDUAL
24 March 1988	465-20	Todd H. Foster [signature]

FAA Form 337 (7-67)

(8320)

FAA AIRCRAFT REGISTRY
CAMERA NO. 3N DATE: 4 - 24 - 89

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED *(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)*

Installed King KA 33 cooling kit in accordance with King Installation Manual.

Electrical load does not exceed 80% of charging system.

EMI check made and no interference noted.

Weight and balance revised.

LAST ITEM

☐ ADDITIONAL SHEETS ARE ATTACHED

☆ U.S. GOVERNMENT PRINTING OFFICE: 1986 [illegible]-121/40146

FAA AIRCRAFT REGISTRY
CAMERA NO. 3 N DATE: 4 - 24 - 89

DEG/JD

US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only
Office Identification
WP FSDO-5

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed $1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make: Beech	Model: A36
	Serial No.: E-059	Nationality and Registration Mark: USA N4260S
2. Owner	Name (As shown on registration certificate): Beechcraft N4260S	Address (As shown on registration certificate): 51 N San Marcos Rd B, Santa Barbara, Ca. 93111

3. For FAA Use Only

4. Unit Identification				5. Type	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	(As described in Item 1 above)				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

RECEIVED
SEP 28 1987
LGB FSDO 5

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
Flite-Comm Electronics, Inc. 3501 Lakewood Blvd. Long Beach, Ca. 90808	U.S. Certificated Mechanic Foreign Certificated Mechanic X Certificated Repair Station Manufacturer	Radio Class I, II, III (465-20) Limited Instrument, Limited Airframe, Specialized Services

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date	Signature of Authorized Individual
22 September 1987	Todd H. Foster

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is ☒ APPROVED ☐ REJECTED

BY							
		FAA Flt. Standards Inspector		Manufacturer		Inspection Authorization	Other (Specify)
		FAA Designee	X	Repair Station		Person Approved by Transport Canada Airworthiness Group	

Date of Approval or Rejection	Certificate or Designation No.	Signature of Authorized Individual
22 Sept 1987	465-20	Todd H. Foster

FAA Form 337 (4-87)

FAA AIRCRAFT REGISTRY
CAMERA NO. 3N DATE: 4 - 24 - 89

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Removed DG, KR 85 ADF System and KI 201C. Installed Northstar M1 Loran C, Bendix HSD-830 HSI, KN 77 and S-TEC ST-256 Manual Trim System in accordance with manufacturers' instructions.

S-TEC ST-256 Manual Trim System installed in accordance with STC SA52728SW-D dated 1-31-84 and manufacturer's instruction manual.

Airplane Flight Manual Supplement for S-TEC ST-256 added.

Aircraft flight tested for proper operation per STC instructions.

Equipment is mounted on existing radio racks or instrument hole.

Northstar M1 Loran C installed in accordance with AC 43.13-1A, Chapters 11 and 15, AC 43-13-2A, Chapters 2 and 3, AC 20-121, Paragraph 3, Appendix I and manufacturer's instructions.

Loran left/right and nav flag information are switched to the HSI which is shared with the #1 nav receiver. Also installed ILS energize interlock. VOR and Loran are switched with a combination switch annunciator.

Instrument panel placarded as follows: "Loran C System Approved for VFR Operation Only."

Ground and flight check made and Loran C exhibits no interference to other aircraft communications or navigations systems.

All antenna installations made in accordance with Advisory Circular 43.13-2A, Paragraph 38, 39 and 44.

Electrical load does not exceed 80% of charging system.

Weight and balance revised.

LAST ITEM

☐ Additional Sheets Are Attached

FAA AIRCRAFT REGISTRY
CAMERA NO. 2N DATE: 10-18-84

Form Approved
Budget Bureau No. 04-R0058

DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION

APPLICATION FOR AIRWORTHINESS CERTIFICATE

INSTRUCTIONS—Print or type. Do not write in shaded areas; these are for FAA use only. Submit original only to an authorized FAA Representative. If additional space is required, use an attachment. For special flight permits complete Sections II and VI or VII as applicable.

I. AIRCRAFT DESCRIPTION

1. REGISTRATION MARK: N4260S	2. AIRCRAFT BUILDER'S NAME (make): BEECH AIRCRAFT CORP.	3. AIRCRAFT MODEL DESIGNATION: A36	4. YR. MFG.: 1976
5. AIRCRAFT SERIAL NO.: E-959	6. ENGINE BUILDER'S NAME (make): Continental	7. ENGINE MODEL DESIGNATION: IO-520-BA	FAA CODING
8. NUMBER OF ENGINES: 1	9. PROPELLER BUILDER'S NAME (make): McCauley	10. PROPELLER MODEL DESIGNATION: 3A32C76	11. AIRCRAFT IS: NEW / USED / IMPORT

II. CERTIFICATION REQUESTED

APPLICATION IS HEREBY MADE FOR: (Check applicable items)

A [1] [X] STANDARD AIRWORTHINESS CERT. (Indicate category) — NORMAL [X] UTILITY [ ] ACROBATIC [ ] TRANSPORT [ ] GLIDER [ ] BALLOON

B [ ] SPECIAL AIRWORTHINESS CERTIFICATE (Check appropriate items)

- [2] LIMITED
- [5] PROVISIONAL (Indicate class): [1] CLASS I; [2] CLASS II
- [3] RESTRICTED (Indicate operation(s) to be conducted): [1] AGRICULTURE & PEST CONTROL; [2] AERIAL SURVEYING; [3] AERIAL ADVERTISING; [4] FOREST (Wild life conservation); [5] PATROLLING; [6] WEATHER CONTROL; [0] OTHER (Specify)
- [4] EXPERIMENTAL (Indicate operation(s) to be conducted): [1] RESEARCH AND DEVELOPMENT; [2] AMATEUR BUILT; [3] EXHIBITION; [4] RACING; [5] CREW TRAINING; [6] MKT. SURVEY; [0] TO SHOW COMPLIANCE WITH FAR
- [8] SPECIAL FLIGHT PERMIT (Indicate operation to be conducted then complete Section VI or VII as applicable on reverse side): [1] FERRY FLIGHT FOR REPAIRS, ALTERATIONS, MAINTENANCE OR STORAGE; [2] EVACUATE FROM AREA OF IMPENDING DANGER; [3] OPERATION IN EXCESS OF MAX. CERTIFICATED TAKE-OFF WEIGHT; [4] DELIVERING OR EXPORT; [5] PRODUCTION FLIGHT TESTING

C [6] MULTIPLE AIRWORTHINESS CERTIFICATE (Check appropriate Restricted Operation and Standard or Limited as applicable above)

III. OWNER'S CERTIFICATION

A. REGISTERED OWNER (As shown on Certificate of Aircraft Registration) — IF DEALER, CHECK HERE → X

NAME: BEECH AIRCRAFT CORP.
ADDRESS: 9709 E. Central, Wichita, Kansas 67201

B. AIRCRAFT CERTIFICATION BASIS (Check applicable blocks and complete items as indicated)

- X AIRCRAFT SPECIFICATION OR TYPE CERTIFICATION DATA SHEET (Give No. and Revision No.): 3A15, Rev. 57
- X AIRWORTHINESS DIRECTIVES (Check if all applicable AD's complied with and give latest AD No.): Issue 76-23
- AIRCRAFT LISTING (Give page No(s).): N/A
- SUPPLEMENTAL TYPE CERTIFICATE (List number of each STC incorporated): N/A

C. AIRCRAFT OPERATION AND MAINTENANCE RECORDS

- X CHECK IF RECORDS IN COMPLIANCE WITH FAR 91.173
- TOTAL AIRFRAME HOURS ~~Enter for used aircraft only~~: 2:15
- [3] EXPERIMENTAL ONLY—Enter hours flown since last certificate issued or renewed: N/A

D. **CERTIFICATION**—I hereby certify that I am the owner (or his agent) of the aircraft described above; that the aircraft is registered with the Federal Aviation Administration in accordance with Section 501 of the Federal Aviation Act of 1958, and applicable Federal Aviation Regulations; and that the aircraft has been inspected and is airworthy and eligible for the airworthiness certificate requested.

DATE OF APPLICATION	NAME AND TITLE (Print or type)	SIGNATURE
11-29-76	C. F. Grace - Foreman Flight Inspection	C F Grace

IV. INSPECTION AGENCY VERIFICATION

A. THE AIRCRAFT DESCRIBED ABOVE HAS BEEN INSPECTED AND FOUND AIRWORTHY BY: (Complete this section only if FAR 21.183 (d) applies.)

- [2] FAR PART 121 OR 127 CERTIFICATE HOLDER (Give Certificate No.)
- [3] CERTIFICATED MECHANIC (Give Certificate No.)
- [4] CERTIFICATED REPAIR STATION (Give Certificate No.)
- [5] AIRCRAFT MANUFACTURER (Give Name of Firm)

DATE	TITLE	SIGNATURE

V. FAA REPRESENTATIVE CERTIFICATION

(Check ALL applicable blocks) I find that the aircraft described in Section I or VII meets the requirements for: [X] The certification requested, or [1] Amendment or modification of its current airworthiness certificate. Inspection for a special flight permit under Section VII was conducted by: [ ] FAA Inspector; certificate holder under [ ] FAR 65, [ ] FAR 121 or 127, or [ ] FAR 145.

DATE	DISTRICT OFFICE	DESIGNEE'S SIGNATURE AND NO.	FAA INSPECTOR'S SIGNATURE
11-29-76	ACE EMDO 3-0-43	D M Porteous D. M. Porteous, DOA, PC#8	

FAA Form 8130-6 (7-70)

FAA AIRCRAFT REGISTRY
CAMERA NO. 2N DATE: 10-18-84

VI. PRODUCTION FLIGHT TESTING

A. MANUFACTURER

NAME | ADDRESS

B. PRODUCTION BASIS *(Check applicable item)*

- [ ] PRODUCTION CERTIFICATE *(Give production certificate number)*
- [ ] TYPE CERTIFICATE ONLY
- [ ] APPROVED PRODUCTION INSPECTION SYSTEM

C. GIVE QUANTITY OF CERTIFICATES REQUIRED FOR OPERATING NEEDS:

DATE OF APPLICATION | NAME AND TITLE *(Print or type)* | SIGNATURE

VII. SPECIAL FLIGHT PERMIT PURPOSES OTHER THAN PRODUCTION FLIGHT TEST

A. DESCRIPTION OF AIRCRAFT

REGISTERED OWNER | ADDRESS

BUILDER *(Make)* | MODEL

SERIAL NUMBER | REGISTRATION MARK

B. DESCRIPTION OF FLIGHT

FROM | TO

VIA | DEPARTURE DATE | DURATION

C. CREW REQUIRED TO OPERATE THE AIRCRAFT AND ITS EQUIPMENT

- [ ] PILOT
- [ ] CO-PILOT
- [ ] NAVIGATOR
- [ ] OTHER *(Specify)*

D. THE AIRCRAFT DOES NOT MEET THE APPLICABLE AIRWORTHINESS REQUIREMENTS AS FOLLOWS:

E. THE FOLLOWING RESTRICTIONS ARE CONSIDERED NECESSARY FOR SAFE OPERATION *(Use attachment if necessary)*

F. **CERTIFICATION**—I hereby certify that I am the registered owner (or his agent) of the aircraft described above; that the aircraft is registered with the Federal Aviation Administration in accordance with Section 501 of the Federal Aviation Act of 1958, and applicable Federal Aviation Regulations; and that the aircraft has been inspected and is airworthy for the flight described.

DATE | NAME AND TITLE *(Print or type)* | SIGNATURE

VIII. AIRWORTHINESS DOCUMENTATION (FAA use only)

X	A. Operating Limitations and Markings in Compliance with FAR 91.31 as Applicable		G. Statement of Conformity, FAA Form 317 *(Attach when required)*
	B. Current Operating Limitations Attached		H. Foreign Airworthiness Certification for Import Aircraft *(Attach when required)*
	C. Data, Drawings, Photographs, etc. *(Attach when required)*		I. Previous Airworthiness Certificate Issued in Accordance with FAR ____ CAR ____ *(Original attached)*
X	D. Current Weight and Balance Information Available in Aircraft		
	E. Major Repair and Alteration, FAA 337 *(Attach when required)*		J. Current Airworthiness Certificate Issued in Accordance with FAR 21.183(a) per 21.273 *(Copy attached)*
X	F. This Inspection Recorded in Aircraft Records	X	

C FAA AC 78-9655

FAA AIRCRAFT REGISTRY
CAMERA NO. 2N DATE: 10-18-84

UNITED STATES OF AMERICA
DEPARTMENT OF TRANSPORTATION—FEDERAL AVIATION ADMINISTRATION

STANDARD AIRWORTHINESS CERTIFICATE

1. NATIONALITY AND REGISTRATION MARKS	2. MANUFACTURER AND MODEL	3. AIRCRAFT SERIAL NUMBER	4. CATEGORY
N2605	BEECH AIRCRAFT CORP. A36	E-959	UTILITY

5. AUTHORITY AND BASIS FOR ISSUANCE
This airworthiness certificate is issued pursuant to the Federal Aviation Act of 1958 and certifies that, as of the date of issuance, the aircraft to which issued has been inspected and found to conform to the type certificate therefor, to be in condition for safe operation, and has been shown to meet the requirements of the applicable comprehensive and detailed airworthiness code as provided by Annex 8 to the Convention on International Civil Aviation, except as noted herein.
Exceptions:

NONE

6. TERMS AND CONDITIONS
Unless sooner surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator, this airworthiness certificate is effective as long as the maintenance, preventative maintenance, and alterations are performed in accordance with Parts 21, 43, and 91 of the Federal Aviation Regulations, as appropriate, and the aircraft is registered in the United States.

DATE OF ISSUANCE		DESIGNATION NUMBER
Nov. 29, 1976	D. M. Fortescue	DOA, PC#8

Any alteration, reproduction, or misuse of this certificate may be punishable by a fine not exceeding $1,000, or imprisonment not exceeding 3 years, or both. THIS CERTIFICATE MUST BE DISPLAYED IN THE AIRCRAFT IN ACCORDANCE WITH APPLICABLE FEDERAL AVIATION REGULATIONS.

FAA Form 8100-2 (7 67) FORMERLY FAA FORM 1362 GPO 1967 O-270 931

FAA AIRCRAFT REGISTRY
CAMERA NO. 2N DATE: 10-18-84