

#3

AIRCRAFT TECHNICAL RECORD

Section 1. AIRFRAME LOG

#3

AIRCRAFT DATA

1. Nationality & Registration N705 R
2. Manufacturer's Designation (Make & Model) Cessna T12182
3. Manufacturer's Serial Number 1218202037
4. Type Approval or Specification Number _____
5. Date of Manufacture _____
6. Aircraft Inspection Cycle _____



Section 1: Airframe Log

DATE / /	AIR TIME				LANDINGS / CYCLES		DETAILS OF TASK (INSPECTION, REPAIRS & REPLACEMENT)
	DAILY TOTAL		TOTAL		DAILY TOTAL	TOTAL	
	HRS	MIN	HOURS	MIN			
							N705R

CAR FOR

IFR CERT

N705R
TACH: 3958.2

MODEL: R182

S/N: R-18202037
W.O.# AVS-18-0038

LEADING EDGE
JET CENTER

DATE: 5/17/2018

PERFORMED A MINOR ALTERATION BY INSTALLING A COBHAM ANTENNA, CI-177-1, TSO: C37B,C38C, S/N 571662, ON TOP OF FUSELAGE AT STATION 78, INSTALLED CONNECTOR, P/N MS3112E12-3S B 31-220NRF X BNC CONNECTOR ON O32" 2024-T3 2" BY 3" MOUNTED TO REAR SEAT TUBE BY 2 CUSHION CLAMPS MS21919-14, PLACED IN-LINE BUS FUSE WITH 3 AMP FUSE ON MAIN BUSS BARE AND ROUTED POWER THROUGH 18 GAGE WIRE M22779/16 TO CONNECTOR. ROUTED ANTENNA COAX, RG-142, FROM ANTENNA TO BNC CONNECTOR WORK ACCOMPLISHED IN ACCORDANCE WITH AC43-13-2B DATED 3/3/2008, CHAPTER 3, AC43-13-1B, CHG 1, 9/8/98 CHAPTER 11, SECTION 11-76. THIS WILL BE USED WITH PORTABLE EQUIPMENT (TDFM TRANSCIEVER) NOT INSTALLED AT THIS TIME.
REVISED WEIGHT AND BALANCE THIS DATE TO REFLECT THIS ALTERATION.
TESTED ANTENNA & COAX IN ACCORDANCE WITH AEROSONIC IFR-4000 OPERATING INSTRUCTIONS AND DETERMINED SWRV VALUE TO BE 185 DB AND WITHIN ACCEPTABLE LIMITS
COMPLIED WITH REQUIREMENTS OF § 91.411 AND § 91.413 - ALTIMETER, STATIC SYSTEM AND TRANSPONDER CHECKS.
ALTIMETER, STATIC SYSTEM AND AUTOMATIC PRESSURE ALTITUDE REPORTING EQUIPMENT HAVE BEEN TESTED TO 20,000 FT. IN ACCORDANCE WITH REQUIREMENTS OF TITLE 14 CFR § 91.411 AND HAVE BEEN FOUND TO COMPLY WITH TITLE 14 CFR PART 43, APPENDIX E B F.
TRANSPONDER, P/N O66-1062-00, S/N 79635, TESTED IN ACCORDANCE WITH REQUIREMENTS OF TITLE 14 CFR § 91.413 AND HAVE BEEN FOUND TO COMPLY WITH TITLE 14 CFR PART 43, APPENDIX F. WITH REFERENCE TO TECHNISONIC INDUSTRIES INC. DRAWING NO.994074 REV. F DATED 09/16/99

SIGNED:

LEADING EDGE JET CENTER, LLC.

FAA CERTIFIED REPAIR STATION #H6MR663L
63132 POWELL BUTTE HWY BEND, OR 97701 541-388-0019

FOR:

AIRFRAME

N705R

MAKE/MODEL: Cessna TR182
SERIAL NUMBER: R18202037
TACH/HOBBS: 0.0
AFTT: 3,959.6
ENGINE MAKE/MODEL: Lycoming O-540-L3C5D
ENGINE SERIAL NUMBER: L-27116-40E
ENGINE TTSN: 956.4
ENGINE SMOH:
PROP MAKE/MODEL: McCauley B3D32C407-B
PROP SERIAL NUMBER: 850395
PROP SMOH: 626.4 TT: 3,959.6

LAWRENCE AIR SERVICE



90295 Air Cargo Way Ste. 4
Eugene, OR 97402
541-689-3331

6-26-18

DESCRIPTION OF WORK:

Performed an annual inspection IAW FAR 43 Appx. D and AMM inspection checklist. Replaced central vacuum filter with new PN: D9-18-1 and vacuum relief filter with new PN: B3-5-1. Re-installed O2 tank after hydrostatic recert and regulator overhaul by C&L Aero. Next Hydrostatic is due 6/2023 serviced with 1600psi ABO. Replace adel clamp on RH landing light wire under alternator, installed standoffs and spiral wrap on horizontal wire bundle on firewall to prevent chaffing, removed zip ties and installed adel clamps on starter power cables, installed smaller adel clamp to provide rigid support for alternator power cables, repositioned carb heat cable to provide more direct routing, replaced safety wire on 2nd inboard and outboard LH wing fuel sumps and outboard RH wing fuel sump, removed safety chain on RH aux fuel tank cap as there are no provisions to secure the chain, replaced hour recording engine tachometer with new Mitchell PN: D1-112-5023 SN: 20253. (Tach time is 0.0 and AFTT is 3959.6) Replaced landing gear hoses with new PN's: AE3663231E0102 & AE3663231E0096. Performed Hydraulic fluid contamination check, inspected cowl flap hinge pins and bushings for wear, performed O2 system functional check, replaced nose landing gear down lock indicator switch with new PN: P6-340004 and left main landing gear down lock indicator switch with new PN: S2088-3 performed a minimum of 5 error free gear cycles including 2 error free manual extentions. Serviced Nose strut with nitrogen IAW AMM, removed LH floor mat heel plate, replaced air filter with new Donaldson PN: P13-1367. Complied with the following ICA's BAS inertia reels IAW ICA 1502 RevC, Garmin 430 IAW ICA 190-00356-65 RevB, Robertson STOL IAW SMM10-1 Section 1, Flint Aero Extended Fuel Tanks from ICA obtained on the Flint Aero website, Concorde Battery IAW CMM 5-0171 RevP, Telemetry antenna mount installation IAW 337 Dated 2/13/18 Pg3-4, STEC electric elevator trim IAW Bulletin 355 Section IV. Inspected and tested ELT IAW Artex ICA 570-1600 RevF Chp4 and FAR 91.207 (d) ELT battery due 5/2021.

NAME/
SIGNATURELICENCE:
AME or
AMO/ACA

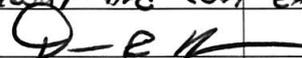
The following AD's were complied with: AD2011-10-09 Effective Date 6/17/11 Seat Tracks IAW para (g) by visual and dimensional inspection, no defects noted, next due 6/30/19 or 4059.6 AFTT whichever occurs first. AD2008-26-10 C effective date 1/5/2009 alternate static air port, inspected IAW para (e) (2) (i) placard not covering port no further action required. AD2008-10-02 effective date 5/12/2008, alternate static air port, inspected IAW para (e) (1) placard not covering port, no further action required. Hand fire extinguisher inspected IAW NFPA 10.2002 Chp. 6 no defects noted. I certify that this AIRCRAFT has been inspected IAW an ANNUAL inspection and was determined to be in an AIRWORTHY condition.

Mike Bonertz AP3170252IA



Date: 6/28/2018

7-1-2018 The TAB-5 mount and RA-7A ANTENNAS AND ASSOCIATED COAX
TACK 0.9 Run From Antenna TO BPC Bulkhead AT TOP OF THE WING STRUT
TAE 3960.5 have been installed. TWO RESTRICTED PLACARDS HAVE BEEN
INSTALLED. THE RESTRICTED CATEGORY AIRMAN'S CERTIFICATE
IS DISPLAYED AND THE WEIGHT & BALANCE ADJUSTED TO REFLECT
THE INSTALLATION OF ANTENNA MOUNT AND COAX EXTENSION

 ATP 5346181

7-6-2018 Garmin 430 Updated
Fire Ext Visual CK OK
Altimeter CK \pm 75' OK
Dual VOR's X-CK South
OF 775 on 112.9 OK
LJ Church 1452928 ATP

7-7-2018 ALL interior & exterior lights
CK OK. Pitot HT CK OK.
LJ Church 1452928 ATP

Section 1: Airframe Log

DATE	AIR TIME				LANDINGS / CYCLES		DETAILS OF TASK (INSPECTION, REPAIRS & REPLACEMENT)
	DAILY TOTAL		TOTAL		DAILY TOTAL	TOTAL	
	HRS	MIN	HOURS	MIN			
1 1			3960.5				
7-29-18			0000.7				Garmin 430W Updated. Five EXT Visual CK. Altimeter $\pm 75'$ CK. VOR's X-CK within ± 4 deg 10E TTS. W Church 1452928 ATP
			TTAF 3960.8				
7-30-2018			TACH 7.5				The TAB-5 mount And RA-2A Antennas And Associated COAX Run From Antenna To BNC Bulkhead AT THE TOP of the wing STRIP HAVE BEEN REMOVED. TWO RESTRICTED PLACARDS HAVE BEEN REMOVED. THE STANDARD CATEGORY AIRWORTHINESS CERTIFICATION IS DISPLAYED AND WEIGHT + BALANCE HAS BEEN SO RETURNED THE REMOVAL OF THE ANTENNA, ANTENNA MOUNT AND COAX EXTENSION J-R ATP 3346181
			TTAF 3968.0				
8-1-2018			TACH 9.4				REMOVED DUKES ELECTRIC FUEL PUMP 1515-00-1RX S/N 0415 INSTALLED O/H DUKES FUEL PUMP P/N 1515-00-1 S/N 78460 O/H BY FUEL SUPPLY SOURCE INC FS 28-013 DATE 7-10-2018 OHS + LORRAN CHECK GOOD J-R ATP 3346181
			TTAF 3969.9				
9-15-2018			TACH 9.4				GARMIN 430W Updated. Five EXT visual CK OK. ALTIMETER WITHIN $\pm 75'$ CK OK VOR's X-CK within ± 4 Deg @ 112.9 Approx 20 mi SW EUG W Church ATP 1452928
			TTAF 3969.9				
4-26-19			TACH 34.7				Garmin 430W Updated. Five EXT Visual CK. VOR Compared within 30. ALT $\pm 75'$ CK Pitot HT + EXT lights CK. W Church
			TT 3994.3				
							ATP 1452928

	NAME/ SIGNATURE	LICENCE: AME or AMO/ACA
<p>9-23-18 Para 9.4 The TAB-5 mount + RA-2A Antennas and Associated COAX Run From ANTENNA TO BNC Bulkhead at top of the wing strut have been installed. Two Restricted placards have been installed. The RESTRICTED CATEGORY AIRWORTHINESS CERTIFICATE is DISPLAYED AND THE WEIGHT + BALANCE Amended TO REFLECT THE INSTALLATION OF ANTENNA MOUNT AND COAX EXTENSION</p> <p style="text-align: right;">D-R N AFA 3346181</p>		
<p>11-13-18 Table 20.9 The TAB-5 mount and RA-2A Antennas and Associated COAX Run From ANTENNA TO BNC Bulkhead at the top of the wing strut have been removed. Two Restricted placards have been removed. The STANDARD CATEGORY AIRWORTHINESS CERTIFICATE is displayed. The weight + BALANCE Amended TO REFLECT Removal of the Antennas, Antenna mount and COAX extension</p> <p style="text-align: right;">D-R N AFA 3346181</p>		

Date: 04/13/2019 Hours: 22.4 Tail: N705R

Make: Cessna Model: TR182 S/N: R18202037

Name: Thomas Andrews Cert: 36451381A

Co: Lake County Aviation Signed: 

CW AD2011-10-09 by visual and measurement inspection and found no discrepancies. Next due at annual or tach 122.4 CW FAR 91.207d ELT test and inspection, all ok. No corrosion found. Functional check good. Batteries next due replacement 05/2021. No new Ads at this time. Installed Overhauled propeller SN 850395 model B3D32C407-B. Installed new Garmin GTX 345 in place of existing transponder to CW ADS-B requirements. I find this installation to be a minor alteration due to replacing same or better than existing transponder. Installed new AMS 44 audio panel. The following equipment was removed DME, ADF, ADF HEAD, ADF ANTENNA BENDIX AUDIO PANEL AND KT 76 TRANSPONDER. Weight and balance was revised at this time. FAR 91.411/413 was performed by JLC Avionics 4/17/2019 and log entry was placed in Avionics log book. Performed an Annual inspection IAW FAR 43 Appendix D and found this Aircraft to be Airworthy.

Section 1: Airframe Log

DATE / /	AIR TIME				LANDINGS / CYCLES		DETAILS OF TASK (INSPECTION, REPAIRS & REPLACEMENT)
	DAILY TOTAL		TOTAL		DAILY TOTAL	TOTAL	
	HRS	MIN	HOURS	MIN			
5-1-2019			34	7			ACFT Fire extinguisher re certified + W/Charch 1452928 ATP installed.

Date: 05/07/2019 **Hours:** Tach 32.4 **Tail:** N705R
Make: Cessna **Model:** TR182 **S/N:** R-18202037
Name: Thomas Andrews **Cert:** 3645138IA
Co: Lake County Aviation **Signed:** *[Signature]*

Installed Technisonic TDFM-136b FM transceiver PN 091258-1-10 SN FDA1353IAW with STC number SA00812NY, installation manual TiL Document No. 08RE398 Revision B Issue 2 JANUARY 2013 and ICAs manual TIL document No. 06RE379 Rev N/C dated July 2006. TR 182 s on the AML dated December 13, 2010. See form 337 dated this.

Date: 05/09/2019 **Hours:** Tach 34.2 **Tail:** N705R
Make: Cessna **Model:** TR182 **S/N:** R-18202037
Name: Thomas Andrews **Cert:** 3645138IA
Co: Lake County Aviation **Signed:** *[Signature]*

Replaced 720 channel KX155 TSO radio with serviceable 760 channel KX155 radio SN118363. Tested both the com and nav side of the system. The VOR Needle was within two degrees. Installed new cotter pin and checked throttle rigging. Shimmed electric trim pulley referencing S-tech drawings. Tested trim system and found to be working properly. Performed dynamic propeller balance using a micro vibe balancer. Removed spinner installed weights on spinner back plate and installed spinner on aircraft. Starting IPS were .51 IPS and final balance IPS reading was .08. Found grounded PTT switch on instrument panel and repaired. Installed klix on 3AMP circuit breaker for TDFM 136 FM radio.

5-28-19			37	5			Garmin 430W updated. Fire ext Visual CK OK. ALT within +/- 75' of Field Elev. VOR' Compared #2 plus 3 deg error - within limits. Pilot ht CK. All int + ext lights CK. W/Charch ATP 1452928
PAGE TOTALS							CARRY FORWARD to Top of Next Page

	NAME/ SIGNATURE	LICENCE: AME or AMO/ACA
<p>6-11-2019 TACH 43.6 The TAB-5 multi RA-2A ANTENNAS AND ASSOCIATED COAX RUN FROM ANTENNA TO BNC Bulkhead AT TOP OF THE wing strut HAVE BEEN installed. TWO RESTRICTED PLACARDS HAVE BEEN installed. THE RESTRICTED CATEGORY AIRWORTHINESS CERTIFICATE IS DISPLAYED AND THE WEIGHT + BALANCE AMENDED TO REFERENCE THE INSTALLATION OF THE ANTENNA MOUNT.</p>	<p><i>[Signature]</i> ATP 3346181</p>	
<p>6-12-2019 TACH 48.7 The TAB-5 multi RA-2A ANTENNA AND ASSOCIATED COAX RUN FROM ANTENNA TO BNC Bulkhead AT TOP OF THE wing strut HAVE BEEN REMOVED. TWO RESTRICTED CATEGORY PLACARDS HAVE BEEN REMOVED. THE STANDARD CATEGORY AIRWORTHINESS CERTIFICATE IS DISPLAYED, THE WEIGHT + BALANCE IS AMENDED TO REFERENCE THE REMOVAL OF THE ANTENNAS, ANTENNA MOUNTS, + COAX EXTENSION.</p>	<p><i>[Signature]</i> ATP 3346181</p>	
<p>6-20-19 TACH 50.7 Garmin 430W updated. Altimeter CK $\pm 75'$. VOR's Compared #2 +2 deg error within limits. All interior/exterior lights checked. Pitot heat CK OK. Fire extinguisher Visual Insp CK OK. VJ Church ATP/CFI 1452928</p>		
<p>7-29-19 TACH 58.2 Garmin 430W updated. Altimeter CK $\pm 75'$ OK. VOR's Compared not bearing found within ± 4 deg. All interior + exterior lights CK OK. Pitot heat CK OK. Fire ext visual + pass CK OK. VJ Church 1452928 ATP</p>		
<p>7-30-19 TACH: 58.2 Installed propeller S/N: 850395 after bench test @ American Propeller Service. ground run, leak check OK. <i>[Signature]</i> ATP 3890271</p>	<p><i>[Signature]</i> ATP 3890271</p>	

Section 1: Airframe Log

DATE / /	AIR TIME				LANDINGS / CYCLES		DETAILS OF TASK (INSPECTION, REPAIRS & REPLACEMENT)
	DAILY TOTAL		TOTAL		DAILY TOTAL	TOTAL	
	HRS	MIN	HOURS	MIN			
							<p>7-30-19 TACH: 58.2 Installed Door STOWARDS per manufacture's installation manual & STC SA01120SE PIN: MVA-200C10MBO. J. Church ABP 3890271</p>
							<p>7-30-19 TACH: 58.9 Installed TDFM-136 A loaner S/N: FDA1353. J. Church ABP 3890271</p>
8-17-19			59.4				<p>Garmin 430 updated. Altimeter OK I 75' VOR₂ CK TO checker - 3 errors on EUG. All ext + int lights checked. Pitot HT CK OK, stall warning CK OK. Fire ext Visual Insp CK J. Church ATP1452928</p>
10-10-2019			121.7				<p>Garmin 430W updated to 11-7-19. Altimeter CK I 75'. Dual VOR₂ X-check TO within -2° error using EUG VOR. Ext + int lights CK. Pitot HT CK OK, stall warning system CK OK. Fire ext visual CK + Sign OFF. J. Church ATP1452928</p>

Date: 12/04/2019 **Hours:** 122.5 **Tail:** N705R
Make: Cessna **Model:** TR182 **S/N:** R18202037
Name: Thomas Andrews **Cert:** 36451381A
Co: Lake County Aviation **Signed:** 

CW FAR 91.207d ELT test and inspection, all ok. No corrosion found. Functional check good. Batteries next due replacement 10/2020. CW AD2011-10-09 par. G section 1 through 8 buy visual and measurement inspection and found to be within limits. Next inspection due at tach 222.5 or next Annual/100 hour insp. CW recurring AD's, No new Ads at this time see AD compliance report dated this date for details. I certify that this Aircraft has been inspected in accordance with an annual inspection using criteria found in FAR part 43 appendix D and AMM check list and has been found to be airworthy.

		NAME/ SIGNATURE	LICENCE: AME or AMO/ACA
12-21-2019 TACH 127.0	Garmin 430 Updaters Checked Baro ALT CK $\pm 75'$ e TTS Dual VOR's X-checked AT TTS Within 2° deg of bearing w/EF EUB VOR. Int + Ext lights Checked Pitot HT CK, Stall Warn Sys CK, Fire tag Visual CK + tagged. W/Church 1452928 ATP		
1-27-20 TACH 127.0	Garmin 430W updated Fire ext Visual CK + tag updated ALT meter within limit ($\pm 75'$) e TTS VOR's X-Check REUB VOR 2 deg. All ext + INT lights check Pitot heat CK Standby Vac + Elec fuel pump CK Stall Warn Sys Check W/Church 1452928 ATP		
3-5-20 TACH TACH 127.0	Garmin 430W updated Fire ext Visual CK OK. All ext + INT lights CK OK Pitot heat CK OK. Stall Warning CK OK Elec Vac pump + standby static CK OK W/Church 1452928 5/20 ATP		
4-5-20 TACH 127.0	Garmin 430W updated. Fire Ext CK. All ext + INT lights CK. Stall Warn CK. Pitot HT CK VOR's CK. Aux fuel pump CK. Standby Vac pump CK ALT static CK ALT CK $\pm 75'$ W/Church 1452928 ATP		

Section 1: Airframe Log

DATE	AIR TIME				LANDINGS / CYCLES		DETAILS OF TASK (INSPECTION, REPAIRS & REPLACEMENT)
	DAILY TOTAL		TOTAL		DAILY TOTAL	TOTAL	
	HRS	MIN	HOURS	MIN			
1 / 1							
5-17-20			128	8			Today Cks Completed. Five ext inspected, UOR OK EUB within 4", All interior & exterior lights CK, Pitot Heat CK OK, Stdby VAC pump CK, Elec Aux fuel pump CK, Garmin 430W updated. Stall Warning CK. Altimeter CK within 75' stdby static CK ATP Church ATP 1452928
6-25-20			128	8			Garmin 430W updated. Pitot Heat CK OK. All ext + int lights CK OK. Stby VAC pump CK OK. ALT static CK OK. Elec fuel pump CK OK. D/timeter CK within ±75' limits. UOR bearings CK OK on EUB 112.9 within ±3 deg limits Stall Warning Sys CK OK. Five Ext visual Insp CK OK + Signed off by Church ATP ATP 1452928
7-19-20			128	8			Garmin 430W updated. Pitot Heat Stall Warning Sys CK. ALT static CK Stby VAC pump CK. Aux fuel pump CK. All ext + int lights CK. Altimeter CK within 75' ±. Five ext CK + updated by Church ATP 1452928

Date: 06/25/2020

Hours: 122.5

Tail: N705R

Make: Cessna

Model: TR182

S/N: R18202037

Name: Thomas Andrews

Cert: 36451381A

Co: Lake County Aviation

Signed: 

(This entry supersedes prior entry dated 12/04/2019 in engine repair logbook and is amended to clarify part numbers used in repair of #2 cylinder during the annual inspection on 12/4/2019.) Removed #2 cylinder due to low compression and found burnt exhaust valve. Replaced exhaust valve PN. LW-19001 with new valve, piston ring set PN 05K22035 gasket 75906. Gasket 71973 and exhaust valve keepers PN 17C21191. All work was performed referencing Lycoming OH manual 60294-7 Rev 7-10D 2011 and Lycoming parts manual PN PC-415-1.

9-16-20

128.9

Garmin Updated - Five Ext CK - Pitot Heat CK -
Int + Ext lights CK - Aux VAC pump CK
ALT static CK - Turn Coord CK - Garmin
lights CK - ALT CK 75' - UOR 112.9
I 30 CK by Church ATP 1452928

PAGE TOTALS

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10-6-20 TACH 128.9

Garmin 430W updated + all 30 day Checks Completed.

W Church 1452928 ATP

NAME/
SIGNATURE

LICENCE:
AME or
AMO/ACA

11-8-20 TACH 128.9

Garmin 430W updated + all 30 day Checks Completed.

W Church 1452928 ATP

12-1-20 TACH 128.9

Garmin 430W updated + all 30 day Checks Completed.

W Church 1452928 ATP

12-30-20 TACH 128.9

Garmin 430W updated + all 30 day CKs Completed.

W Church 1452928 ATP

1-27-21 TACH 128.9

Garmin 430W updated + all 30 day CKs Completed

W Church ATP 1452928

3-24-21 TACH 128.9

Garmin 430W updated. Pitot heat CK OK. All external +
internal lights CK OK. Stby Vac pump CK OK. Alternate
Static System CK OK. Elec/Aux fuel pump CK OK.

Altimeter CK $\pm 75'$ within limits. Stall Warning System CK OK.

Fire Ext Visual CK OK + Signed OFF. VGRS NOT checked this time

W Church ATP 1452928

Inspect
REMOVE
VALVES
POST
inspect

Section 1: Airframe Log

6a

DATE / /	AIR TIME				LANDINGS / CYCLES		DETAILS OF TASK (INSPECTION, REPAIRS & REPLACEMENT)
	DAILY TOTAL		TOTAL		DAILY TOTAL	TOTAL	
	HRS	MIN	HOURS	MIN			
CARRIED FORWARD							



Lake County Aviation
17819 Airport Road
Lakeview Oregon 97630
541-951-8883

Make: Cessna
Model: TR182
Tail: N705R
S/N: R18202037

Hours: AC: 129.0 TT: 4083.3 EngTT: 1097.9 PropTT: 4083.3

CW FAR 91.207d ELT test and inspection, all ok. No corrosion found. Functional check good. Batteries next due replacement 05/2021. CW AD2011-10-09 par. G section 1 through 8 buy visual and measurement inspection and found to be within limits. Next inspection due at tach 229.0 or next Annual/100 hour insp. CW recurring AD's, one new inspection, CW 24 month fuel line, 5 year fuel drain inspection and 3 years 1000 hour fuel gauge inspection. I certify that this Aircraft has been inspected in accordance with an annual inspection using criteria found in FAR part 43 appendix D and AMM check list and has been found to be airworthy.

Signed: *[Signature]*
Name: Thomas Andrews

Cert: 36451381A
Date: 04/06/2021

Date: 04/06/2021 Make Model: Cessna R182 Tail #: N705R Serial #: R182-02037 Airworthiness Directive Compliance Report

Airframe Hours: Tach 129.0

Misc:
Engine: Lycoming O-540-L3C5D S/N: L-27116-40E Hours: 101.3 TSOH
Propeller: McCauley B3D32C407 S/N: B3D32C407-B Hours: 101.3 TSO

AD#	Category	Subject	Amend #	Eff Date	Rec	Remarks
Date/Hours Compliance	Next Due	Name	Signature	Cert#		
Airframe: Cessna R182						
83-13-01	Airframe	Improper Fuel Cap Sealing Placard	39-4672	08/01/83	Yes	NA
4/13/2019		NA	Thomas Andrews	36451381A		
84-10-01 R1	Airframe	Quick Drains For Water In Fuel	39-5940	07/05/88	Yes	NA BY SN
4/13/2019			Thomas Andrews	36451381A		
11-10-09	Airframe	Seat Rail/Track Assembly	39-16690	06/17/11	Yes	CW par. G section 1 through 8 buy visual and measurement inspection and found to be within limits.
04/06/2021 129.0		Tach 229.0 or annual insp.	Thomas Andrews	36451381A		
20-18-01	Airframe	Forward Cabin Doorpost Bulkhead Cracks	39-21222	11/12/20	Yes	Complied with a visual inspection of the lower forward doorpost at the strut attach fitting for cracks in accordance with steps 1.A., 1.B. and 1.C of the Accomplishment Instructions in SEB95-19 and found no visual cracks or damage.
4/06/2021 Tach Time 129.0		329.0	Thomas Andrews	36451381A		
Engine: Lycoming O-540-L3C5D						
63-14-03	Engine	Oil Pump Drive Shaft	583	07/06/63	Yes*	NA BY SN
4/13/2019			Thomas Andrews	36451381A		
02-12-07	Engine	Oil Filter Converter Plate Gasket or Plate Kit	39-12779	07/03/02	Yes*	NA BY ENGINE MODEL
4/13/2019			Thomas Andrews	36451381A		
09-26-12	Engine	ECI Cracked Cylinder Assemblies	39-16151	02/04/10	Yes	NA BY CYLINDERS INSTALLED.
4/13/2019			Thomas Andrews	36451381A		
Appliance: Power Plant - Magnetos						
05-12-06	Appliance	S-20, S-1200, D-2000, D-3000 Series - Impulse Coupling Assembly	39-14122	07/19/05	Yes*	CW impulse coupling repetitive inspection (table 2)
4/13/2019		1479 ENT.	Thomas Andrews	36451381A		

[Handwritten signatures and initials]
POT HTC
BCK
hann

4-12-2021
 Tach 129.1
 Removed & Replaced EGT BATTEN I/A/W
 ARTEX 406 EGT 570-1600 Rev AE
 CW EGT MAINT & FUEL CHECK 1/4/W FOR 91.207 DITLY
 ALL OK AT THIS TIME NEXT BAT DUE MAY 2028 NEXT
 INSPECTION DUE 4-2022
 D-R 72 - A2P3346181

ACR ELECTRONICS, Inc.
 ARTEX PRODUCTS - CAGE Code: 18560
 Fort Lauderdale, FL 33312
 ME406 LITHIUM BATTERY PACK MnO2
 PART NO: 452-6499 REV: F
 TSO C126 S.N.:390611-012
 EXPIRATION DATE: May /2028 DO-227
 ENV CAT D1XBC[204][204]XRXXXXZAZZ[204]H[XXXX]XX

Pacific Coast Avionics		CRS OPXR455L	
Registration: N705R	Make: Cessna	Model: R182	S/N: R-1820237
Aircraft Time: Hobbs: N/A		Tach: 130.1	

Troubleshoot #1 VHF-AM- Opens squelch on any frequency when transmitting on COM 2. Could not duplicate on ground, the initial momentary squelch break is expected and falls under normal operation of radios. Performed 14 CFR 91.411 certification of the altimetry system. The altimeter, static system, and altitude reporting system have been tested and found to meet the requirements of 14 CFR Part 43 Appendix E to a pressure altitude of 20,000 feet. Performed 14 CFR 91.413 certification of the Garmin GTX 345 Transponder s/n 3L0024326 and found the transponder to meet the requirements of 14 CFR Part 43 Appendix F.

This aircraft has been inspected in accordance with current federal aviation regulations and found to be airworthy with respect to the work performed. Details will be on file at Pacific Coast Avionics under State Wide Contract Services LLC, work order 21-5352.

Signature: Chad Robertson Date: 4/16/2021

Page 1 of 1

4-26-2021 TACH 131.0
 Updated Garmin 430W. VOR's X Check ~~BE~~ 112.9 less than
 4 deg error. All INT & EXT lights CK. Aux Fuel pump &
 Visc pump CK. Coast Warning Sys Check. ALT IZS Check.
 Pitot Heat Check & Stall Warning CK. FIVE EXT CK & Sign OFF.
 J Church ATP CFII 1452928

Date: 07/27/2021	Hours: AC: 4084.4	Tach: 135.5	Hobbs: 3041.5	Tall: 705R
Make: CESSNA	Model: TR182RG	S/N: R18202037		
Name: Thomas Andrews	Cert: 36451381A			
Co: Lake County Aviation	Signed: <u>[Signature]</u>			

VISUAL INSPECTION REQUIRED BY SEB02-2 performed visual inspection of oxygen lines and found no defects at this time. Inspected Ground straps IAW SEB06-5 and found closed end straps installed. No further action is required. INSPECTION OF GEAR PIVOTA AND BUSHING REPLACEMENT IAW SEB90-1R. Removed floor panels and started inspection. Customer requested the aircraft back, installed all floor panes and SEB90-1R was not accomplished. REMOVE AND REPLACE TURBO CHECK VALVES EVERY 1000HR TIS PER SEB92-7-R01. CURRENT CHECK VALVES HAVE 100.6HRS REMAINING BEFORE COMPLIANCE IS DUE. OPEN FLOORS AND INSPECT DOOR POST/WING STRUT FITTINGS FOR CRACKS. No cracks were found and next due at annual inspection. SEB95-4 inspection CW by 12 month inspection next due at 100hour/Annual inspection. INSPECTION OF FUEL STRAINER INTERNALS AS REQUIRED BY SEB97-9. Inspected fuel strainer internals IAW SEB97-9 and found no defective parts. CW SEB 99-18R1 by draining fuel tanks and verified empty and full marks indicated properly. Removed propeller governor for calendar life OH. Propeller governor was installed on engine and not overhauled. This governor is only airworthy for Part 91 flight. No part 135 operations can be performed.

Section 1: Airframe Log

Tach Time 138.1 hrs.
TTAF: 4097.7 hrs.
Engine TT: 1094.5 hrs.
Eng. SMOH: 115.7 hrs.

N705R
Cessna TR182
Lycoming O-540_L3C5D

September 10, 2021
S/N R18202037
S/N L-27116-40E

1. Cessna SEB02-3. C/W SEB02-3, dated 2-18-2002, I/A/W Para 1) Replaced both engine driven and electrically driven standby Vacuum Pumps with new Rapco units, P/N RAP215CC, S/N A75601 and P/N RAP216CW, S/N B42264. Next replacement due: Engine Driven at Tach 638.1 or 9-2027, Electrically driven at 500 hrs. of operation or 9-2031.
2. Cessna SEB09-6-R1. C/W SEB09-6-R1, dated 11-23-2009. Replaced Landing Light, Taxi Light, and Beacon Light Switches due to time in service. All replaced with new, P/N TA201-TW-B, switches. Next switch replacement due 9-2025.
3. Cessna SEB86-05R1. C/W SEB86-5R1, dated 8-28-1987, "Seat Locked" and "Fuel Contamination Warning" placards. Seat Locked Placard, P/N 0705097-1 and Fuel Contamination Warning placard, P/N 070598-1. Installed in full view of pilot.
4. Cessna SEB01-2R2. C/W SEB01-2R2, dated 6-4-2007. Performed Main Landing Gear Actuator Inspection I/A/W Para 1.) through 8.) – all okay at this time, no cracks noted. Next Actuator inspection due at Tach 638.1 hrs.
5. Cessna SEB90-1R3. C/W Cessna SEB90-1R3, dated 3-15-1999 I/A/W Para A, performed Fluorescent Penetrant Inspection – all okay at this time. Bushing replaced on previous inspection, SEB90-1-R2. Performed 11-8-1990, Tach Time 601.5 hrs. per Para D.2.B. Does not need replaced.
6. Cessna SEB95-19, dated 12-29-1995 and A.D. 2020-18-01, dated 11-12-2020, Forward Cabin Door Post Bulkhead Inspection, previously C/W at Tach Time 129.0 hrs., TTAF 4083.3 hrs., 4-6-2021. Next inspection due: Service Bulletin requirement, 1000 hrs. at Tach 1129.0 hrs; A.D. requirement, 1000 hrs. or 36 months at Tach 1129.0 hrs. or 4-2024.
7. SEB95-3. C/W SEB95-3, dated 3-10-95, Flap Support and Roller Inspection. Performed inspection by removing flaps and inspected flap support arms I/A/W Para 1.) through 9.) – no cracks or wear noted. Reinstalled flaps with S1450-3510-032 washers on forward rollers as per Para 6.) A.).
8. SEB95-4. C/W SEB95-4, dated 3-24-95 and Lycoming MSB518A, dated 2-3-95, Thermostatic Bypass Valve. Inspected I/A/W Lycoming MSB51A, step 1 and step 2 – all okay at this time. Next inspection due 9-2022.
9. SEB96-10. C/W SEB96-10, dated 6-7-96, Vacuum Hose Replacement. Replaced all aircraft instrument vacuum hoses I/A/W Para 1.) through 3.) with Cessna P/N R221248 Hose and R221314 Hose.
10. Prop Governor O/H by Wings West Governor Exchange and O/H under W/O WW25709 I/A/W McCauley Service Manual 780401. C/W A..2020-19-06, Dated 11-3-2020, Idler Gear Bearing. Governor installed with new gasket. Run up, ops, leak and performance check good. *Next O/H Due Tach 1438.1 TTAF 5897.7 or 9-2026*

Maintenance performed by Pat Nixon, A&P and Evan Bishop, Assistant.


Darrin R. Humble A&P 3346181

9-18-21 TACH 140.3
 Garmin 430W Updated NOW DUE 10-7-21
 PITOT HEAT CK
 ALL EXT LIGHTS CK
 ALL INT LIGHTS CK
 FIRE EXT CK + SIGN OFF
 VOR'S CK BY COMPARISON LESS THAN 2° DIFF
 FUG VOR 112.9 110G
 STALL WARNING CK
 GEAR HORN + LIGHTS CK
 AUX FUEL PUMP CK
 STBY VAC PUMP CK
 ALT CK I 75 FT
 L A Church ATP 1452928

NAME/
SIGNATURE

LICENCE:
AME or
AMO/ACA

AIRFRAME LOG

LEADING EDGE
JET CENTER

N705R TACH: 139.2 TT: 4098.8 Model: R182 Hobbs: 3046.1 S/N: R-18202037 W.O.# AVS-21-0747

Date: 9/17/2021

Troubleshoot. Received previous troubleshooting that was accomplished, verified discrepancy. Accessed unit connectors to verify cable type. Removed tray for KX155 and relocated away from GNS430, system still broke squelch with no change from original findings. Positioned KX155 and tray back to the original location. Reassembled Accessed area. Contacted customer with findings and reconnected cable replacement to the KX155 using RG400 coaxial cable. System still does not meet required outlined for USFS and is functional, system operational checks good in accordance with Bendix King KX155/165 Installation Manual 006-00179-0006. Rev. 6 dated 10/6/1994. Pertinent details of work performed on file at this location.

Signed:

For:

LEADING EDGE JET CENTER, LLC.

FAA certified Repair Station #1648R063L
63138 Powell Butte Hwy Bend, OR 97701 541-388-6019



LOG BOOK ENTRY

PAGE NO: 1

LOG ENTRY TYPE:	Airframe Log Entry	Airframe Tach: 141 20 hours
AIRCRAFT		Engine 1:
AIRCRAFT TYPE:	Cessna R182	Engine 2:
JOB NO:	10356	Propeller 1:
REGISTRATION:	N705R	Propeller 2:
SERIAL NO:	R-18202037	

Jobs Title: #1 VHF Comm Breaks Squelch when #2 VHF Comm Transmi

- Completed a preliminary inspection. Defects noted, if any, will be listed as new items on this work order.
- #1 VHF-AM Opens squelch on any frequency when transmitting on COMM 2. Removed both #1 and #2 VHF Comm antennas. Removed paint from underneath both VHF comm antennas footprint. Cleaned and Alodined surface. Cleaned both comm antennas base and mounting holes. Replaced mounting hardware on both antennas. Replaced both VHF comm coax BNCs connector at the comm antennas. Re-terminated Pilots, Co-Pilots, and rear pax mic and phone jacks inputs at the AMS-44 audio panel. Rewired mic, phone, and nav audio from the GNC 430W and KX 155 to the AMS 44 audio panel. Replaced and reposition #2 comm antenna to lower right fuselage belly. All work was performed IAW Garmin 400W Series installation manual p/n 190-00356-02 rev M, Bendix/King KX 155 installation manual 006-00179-0006 rev 6, NAT AMS 44 (SM19) installation manual rev 4, and Comant Industries installation instructions II A12206 rev A. Ops check good. Comant CI 122 VHF Comm antenna, S/N 610189
- Performed final inspection, no defects noted.

I certify this aircraft, article, or component was inspected and/or repaired in accordance with current Federal Aviation Regulations and is approved for return to service for the work described.

NAME: Mike Dixon

SIGNATURE:

DATE: 27 Sep 2021

12-8-21 TACH 142.0 TTAF 4101.6

Garmin 430W updated. All EXT + INT LIGHTS CK.

Pitot HT CK. Stall Warn CK. Green Warn CK.

Stby Vbc CK. Aux. elec fuel pump CK. FIRE EXT CK

Act Vbc CK. Altimeter CK ✓ Church ATP 1452928

3-3-2022 TACH 144.5 TTAF 4104.1

The TAB-5 MOVIE & RA-2A ANTENNA AND ASSOCIATED COAX RUN FROM ANTENNA TO BNC BULKHEAD AT TOP OF THE WING STRUT HAVE BEEN INSTALLED TWO RESTRICTED PLACARDS HAVE BEEN INSTALLED, THE RESTRICTED CATEGORY AIRWORTHINESS CERTIFICATE DISPLAYED AND THE WEIGHT & BALANCE AMENDED TO REFLECT THE INSTALLATION OF THE ANTENNA MOVIE

Q-R 2 ATP 3346181

4-26-22 TACH 152.1 TTAF 4111.2

The TAB-5 MOVIE & RA-2A ANTENNA AND ASSOCIATED COAX RUN FROM ANTENNA TO BNC BULKHEAD AT TOP OF THE WING STRUT HAVE BEEN REMOVED TWO RESTRICTED PLACARDS HAVE BEEN REMOVED, THE STANDARD AIRWORTHINESS CERTIFICATE IS DISPLAYED, THE WEIGHT & BALANCE AMENDED TO REFLECT REMOVAL OF THE ANTENNA, ANTENNA MOVIE & COAX CABLES

Q-R 2 ATP 3346181

5-4-22 TACH 152.6 TTAF 4112.2

REMOVED RH HEAT EXCHANGE FWD BRAD CLAMP &

INSTALLED NEW CLAMP ASSY P/N 24540-300

S/N/BAROL NUMBER 8009749338, TORQUED TO 40 IN LBS

SATRED WITH .032 SAFETY WIRE

Q-R 2 ATP 3346181

Section 1: Airframe Log

Tach Time 152.2 hrs.
TTAF: 4111.7 hrs.
Engine TT: 1108.5 hrs.
Eng. SMOH: 129.7 hrs.
Prop TT 4111.7 hrs.
Prop SMOH 129.7 hrs.

N705R
Cessna TR182
Lycoming O-540_L3C5D

April 26, 2022
S/N R18202037
S/N L-27116-40E

Performed Annual Inspection this date I/A/W FAR 43 appendix "D". Performed service items – inspected and cleaned fuel strainer, greased wheel bearings, inspected and serviced brakes. Replaced nose tire and tube, P/N 500-5 6 ply Goodyear Flight Special II and balanced. Replaced R/H Main Tire, P/N 15x600-6 ply Goodyear Flight Custom III. Rotated L/H Main Tire. Performed gear swing and inspection I/A/W 33 item gear inspection checklist. Inspected and cleaned power pack filter, checked system pressures and pressure switch settings using Pressure Gauge, adjusted L/H Main Gear Down Lock clearances. Repositioned Pilot and Co-pilot seats articulating mechanism roll pins, re-centered as necessary, Co-pilot seat back stop clevis pins replaced cotter pins. Replaced Seat Locked Placard with new, P/N 0705071-5, replaced Central Gyro Filter, P/N RA-D9-18-1, replaced Fuel Cap Placards, P/N 0705071- 5 (2 each), C/W ELT Inspection and Function check I/A/W Artex ME406 Doc. Install and Maintenance Manual 570-1600 Rev. V, dated 3-19-15, items A through R with reference to FAR 91.207 D1 through 4. All OK at this time, next inspection due 5-2028.

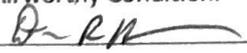
C/W A.D. 2011-10-09, dated 6-17-2011, Seat Rail and Roller Housing Inspection I/A/W Para "G" – all OK at this time, next due Tach 252.2 or 4-2023. A.D. 2022-03-15, dated 3-21-22, Garmin G3X/GI275 Fuel Indicating System – N/A System not installed.

The Following instructions for continued airworthiness have been performed:

- Garmin GNS 430 WAAS ICA I/A/W Garmin ICA. DOC 190-00356-65 Rev. D
- Garmin GTX 345 Transponder ICA. I/A/W Garmin Doc 190-00734-11 Rev 8 Section 4
- FM Transceiver Tecinonics TDF 136B ICA. I/A/W ICA Til Dec. 06Re379 Rev N/C
- Audio Panel NAT ASM 44 on condition only, periodic maintenance not required
- Engine Monitor J.P. Instrumars EGT Scanner #102. On condition only, periodic maintenance not required
- Air Filter Donaldson P13-1367 ICA I/A/W DOC P469075 Rev 8 Repaired at 0.0 Tach. Next replacement due per A.D. 86-24-02 EA 500 hours
- McCauley Propeller B3D32C407-C I/A/W McCauley Propeller Systems Manual, P/N MPC 26-05 Rev 5, dated 8-1-2020 Intro Section 4
- S-TEC ST-255 Electric Trim System ICA S-TEC Doc. 81364
- BAS Inc. Inertia Reel Shoulder Harness System ICA I/A/W DOC 1 Items 1. Through 9.
- Telonics Inc. Tracking Antennas ICA See 337 form page 4 of 4 para 1
- Zeftronics Model No. R25400 ICA See DOC/Drawing ZOO1D5 Rev A, dated 5-17-99
- M-20 AIR/OIL Separator No ICA Listed
- SK210-174B Pilot Secondary Seat Stop ICA I/A/W SEB07-5 Rev 4 ICA Rev A page 14-7
- R/STOL System ICA. DOC Robertson Aircraft Corp. Service and Maintenance Manual 10, Series Rev. dated 4-1-74, sections 1 through 7
- Flint Aero Inc. 20 each 12 gallon Aux Fuel Tanks No ICA. Listed. Check Fittings for leaks, fuel cap and seal for condition inspect and clean fuel pump filter and main fuel strainer
- Shadin Fuel Flow Indicator System Mini-flow 912041T-D on Condition Only. Periodic Maintenance not Required

A.D.'s current to 2022-08. Maintenance Performed by Evan Bishop, A&P and Darrin Humble A&P3346181.

I certify that this AIRCRAFT has been inspected I/A/W an Annual Inspection and was determined to be in Airworthy Condition.


Darrin R Humble A&P3346181 I.A.

AIRFRAME

N705R
May 4, 2022

TR182

S/N R18202037
Tach Time 152.6 hrs.

NAME/
SIGNATURE

LICENCE:
AME or
AMO/ACA

Removed R/H heat exchanger forward band clamp and installed new clamp assembly, P/N 24540-300, S/N Batch number 8009749338. Torqued to 40 inches, saftied with .032 safety wire.
Replaced rusty bellcrank stop hardware at turbo waste gate. Replaced waste gate Control Cable and rigged to R182 and TR182 Service Manual, P/N D2069-3-13.
Untwisted seat belts as necessary and re-installed correctly.


Darrin R Humble A&P3346181

N705R
June 30, 2022

TR182
Tach Time 160.2 hrs.

S/N R18202037
TTAF 4119.8 hrs.

Removed Technisonic, P/N TDFM-136B, S/N FDB5457, Transceiver and sent to factory for repair; reinstalled after repair.
See Technisonic W/O #R31516, dated 6/3/2022. Ops check good.


Darrin R Humble A&P3346181

N705R
June 30, 2022

TR182
Tach Time 160.2 hrs.

S/N R18202037
TTAF 4119.8 hrs.

Placed Aircraft on Cessna Maintenance Program. Performed all Recommended and Mandatory Inspections and Service items I/A/W R182 and TR182 Service Manual, P/N D2069-3-13, up to latest Temporary Revision TR9-1. Completed required Operations/Inspections (SID) and (CPCP) as called out in temporary revision 8, dated 12-1-2011. Performed Cessna Corrosion Inspection and Prevention Program (CPCP) I/A/W Cessna R182 and TR182 Service Manual, P/N D2069-3-13 Temporary Revision 8, dated 12-1-011, Section 2A-30-00 with reference to FAA AC43-4B. The Aircraft is very clean, always hangered and has no corrosion visible during all (SID) and (CPCP) Inspections. As a Corrosion prevention, Corrosion X Corrosion treatment was applied to all exposed internal aluminum surfaces. See the following (SID) (CPCP) Compliance and Current Status of Operations/Inspection listing.


Darrin R Humble A&P3346181

Section 1: Airframe Log

Listing of OEM Programs (CAP), (SID), (CPCP)

Compliance and Current Status

N705R

S/N R1820237

6/30/2022

1. Inspection Items

Complied With		TASK Compliance Action	INTERVAL	OPERATION	ZONE	Next Due	
Date	Tach					Date	Tach
6/30/2022	TTAF 4119.8 TACH 160.2	Inspect aircraft records to verify that all applicable Cessna Service Information Letters, Cessna Service Bulletins and Supplier Service Bulletins are complied with. Verified by Inspecting Aircraft Record	Every 100 hours or 12 months, whichever occurs first.	1	-	Jun-23	260.2
6/30/2022	TTAF 4119.8 TACH 160.3	Inspect aircraft records to verify that all applicable Airworthiness Directives and Federal Aviation Regulations are complied with. Verified by Inspecting Aircraft Record	Every 100 hours or 12 months, whichever occurs first.	1	-	Jun-23	260.2
6/30/2022	TTAF 4119.8 TACH 160.4	Inspect aircraft records to verify that all logbook entries required by the Federal Aviation Regulations are complied with. Verified by Inspecting Aircraft Records	Every 100 hours or 12 months, whichever occurs first.	1	-	Jun-23	260.2
6/30/2022	TTAF 4119.8 TACH 160.5	Inspect aircraft records to verify that all SID Inspections have been complied with as scheduled. Verified by Inspecting Aircraft Records	Every 100 hours or 12 months, whichever occurs first.	1	-	Jun-23	260.2
6/30/2022	TTAF 4119.8 TACH 160.6	Inspect rudder pedal torque tube and cable attachment arms. Refer to 2A-14-01, Supplemental Inspection Document 27-20-01, for inspection procedure. Inspected as per Referenced 2A-14-01, Rudder Pedal Torque Tube Cable Attach Arms. All okay at this time.	Initial: 10,000 hours or 20 years, whichever occurs first; Repeat: 3,000 hours or 5 years, whichever occurs first.	7	211	6/1/2027 or 1160.2	
6/30/2022	TTAF 4119.8 TACH 160.7	Elevator trim system. 1. Inspect elevator trim brackets and actuator support brackets. 2. Inspect pulleys, attaching structure and fasteners. Refer to Section 2A-14-02, Supplemental Inspection Document 27-30-01, for inspection procedures. Inspected as per Section 2A-14-02, Supplemental Inspection Document 27-30-01. All okay at this time.	Initial: 1,000 hours; Repeat: 1,000 hours	14	320,330		1160.2
9/10/2021	TTAF 4097.7 TACH 178.1	Inspect main landing gear retraction system. Refer to Section 2A-14-03, Supplemental Inspection Document 32-10-01, for inspection procedure. Previously Complied With by Compliance of SEB 90-1R3	Initial: 3,000 hours or 10 years, whichever occurs first; Repeat: 500 hours or 5 years, whichever occurs first.	15	721, 722	Sep-26	638.1
6/30/2022	TTAF 4119.8 TACH 160.6	This inspection is for mild/moderate corrosion environment. Inspect main landing gear spring for rust or damage to finish. Refer to Section 2A-14-04, Supplemental Inspection Document 32-13-01, for inspection procedure. All okay at this time - no defects noted.	Initial: 20 years; 10 Repeat: 10 years	10	721, 722	Jun-32	
		This interval is for severe corrosion environment. Inspect main landing gear spring for rust or damage to finish. Refer to Section 2A-14-04, Supplemental Inspection Document 32-13-01, for inspection procedure. Not Applicable	Initial: 10 years; Repeat: 5 year	12	721, 722		
6/30/2022	TTAF 4119.8 TACH 160.2	Inspect main landing gear fittings and attachment of the fittings to the bulkheads. Refer to Section 2A-14-05, Supplemental Inspection Document 32-13-02, for inspection procedure. Inspected Main Gear Fittings and Attachment of the Fittings to the Bulkheads as per Section 2A-14-05, Supplemental Inspection Document 32-13-02.	Initial: 3,000 hours or 5 years, whichever occurs first; Repeat: 1,000 hours or 5 years, whichever occurs first.	16	210	6/1/2027 or 1160.2	
6/30/2022	TTAF 4119.8 TACH 160.2	Inspect nose landing gear torque links, bolts, bushings, and nose landing gear fork. Refer to Section 2A-14-06, Supplemental Inspection Document 32-20-01, for inspection procedure. Inspected nose landing gear torque links, bolts, bushings and nose landing gear fork as per Section 2A-14-06, Supplemental Inspection Document 32-20-01. All okay at this time.	Initial: 3,000 hours or 5 years, whichever occurs first; repeat: 3,000 hours or 5 years, whichever occurs first.	8	720	6/1/2027 or 1160.2	


Darrin R Humble, A&P 3446181

**Listing of OEM Programs (CAP), (SID), (CPCP)
Compliance and Current Status**

N705R

S/N R18202037

6/30/2022

Complied With		TASK Compliance Action	INTERVAL	OPERATION	ZONE	Next Due	
Date	Tach					Date	Tach
6/30/2022	TTAF 4119.8 TACH 160.2	This interval is for mild/moderate corrosion environment. Inspect the carry-thru spar area, door post bulkhead attach fittings and spar channel. Refer to Section 2A-14-07, Supplemental Inspection Document 53-11-01, for inspection procedure. Inspected the carry-thru spar area, door post bulkhead attach fittings and spar channel as per Section 2A-14-07, Supplemental Inspection Document 53-11-01. All okay at this time.	Initial: 20 years; Repeat 10 years	10	210	Jun-32	
		This interval is for severe corrosion environment. Inspect the cabin interior skin panels, frames and stringers. Refer to Section 2A-14-09, Supplemental Inspection Document 53-30-01, for inspection procedure. Not Applicable	Initial: 10 years; Repeat 5 years.	12	210		
4/6/2021	TTAF 4083.5 TACH 129.0	Inspect forward doorpost and surrounding structure. Refer to Section 2A-14-08, Supplemental Inspection Document 53-12-01, for inspection procedure. Previously Complied With per A.D. 2020-18-01 and Cessna SEB95-19.	Initial: 4,000 hours or 10 years, whichever occurs first; Repeat: 1,000 hours or 3 years, whichever occurs first.	22	210	Apr-24	1127.0
6/30/2022	TTAF 4119.8 TACH 160.2	This interval is for mild/moderate corrosion environment. Inspect the cabin interior skin panels, frames and stringers. Refer to Section 2A-14-09, Supplemental Inspection Document 53-30-01, for inspection procedure. Inspected cabin interior skin panels, frames, and stringers as per Section 2A-14-09, Supplemental Inspection Document 53-30-01. All okay at this time.	Initial: 20 years; Repeat 10 years	10	210	Jun-32	
		This interval is for severe corrosion environment. Inspect the cabin interior skin panels, frames and stringers. Refer to Section 2A-14-09, Supplemental Inspection Document 53-30-01, for inspection procedure. Not Applicable	Initial: 20 years; Repeat 10 years	10	210	Jun-32	
6/30/2022	TTAF 4119.8 TACH 160.2	This interval is for typical usage environment. Inspect wing strut fitting and adjacent bulkhead. Refer to Section 2A-14-10, Supplemental Inspection Document 53-30-02, for inspection procedure. Inspected wing strut fitting and adjacent bulkhead as per Section 2A-14-10, Supplemental Inspection Document 53-30-02. All okay at this time.	Initial: 12,000 hours or 20 years, whichever occurs first; Repeat: 2,000 hours or 10 years, whichever occurs first.	19	210	6/1/2032 or 2160.2	
		This interval is for severe usage environment. Inspect wing strut fitting and adjacent bulkhead. Refer to Section 2A-14-10, Supplemental Inspection Document 53-30-02, for inspection procedure. Not Applicable	Initial: 6,000 hours or 10 years, whichever occurs first; Repeat: 1,000 hours or 5 years, whichever occurs first.	20	210		
6/30/2022	TTAF 4119.8 TACH 160.2	This interval is for mild/moderate corrosion environment. Inspect seat rails for corrosion. Refer to Section 2A-14-11, Supplemental Inspection Document 53-47-01, for inspection procedure. Inspected seat rails for corrosion as per Section 2A-14-11, Supplemental Inspection Document 53-47-01 - none found.	Initial: 10 years; Repeat 10 years.	17	211	Jun-32	
		This interval is for severe corrosion environment. Inspect seat rails for corrosion. Refer to Section 2A-14-11, Supplemental Inspection Document 53-47-01 for inspection procedure. Not Applicable	Initial: 5 years; Repeat: 5 years	18	211		
6/30/2022	TTAF 4119.8 TACH 160.2	Inspect horizontal stabilizer and elevator, including spars, ribs, hinge bolts, hinge bearings, attach fittings and torque tube. Refer to Section 2A-14-12, Supplemental Inspection Document 55-10-01, for inspection procedures. Inspected horizontal stabilizer and elevator, including spars, ribs, hinge bolts, hinge bearings, attach fittings and torque tube as per Section 2A-14-12, Supplemental Inspection Document 55-10-01. All okay at this time.	Initial: 10,000 hours or 20 years, whichever occurs first; Repeat: 3,000 hours or 5 years, whichever occurs first.	7	320, 330	Jun-27	3160.2

D-R-H
Darrin R Humble, A&P 3446181

**Listing of OEM Programs (CAP), (SID), (CPCP)
Compliance and Current Status**

N705R

S/N R18202037

6/30/2022

Complied With		TASK Compliance Action	INTERVAL	OPERATION	ZONE	Next Due	
Date	Tach					Date	Tach
6/30/2022	TTAF 4119.8 TACH 160.2	Inspect vertical stabilizer and rudder, including spars, ribs, hinge bolts, hinge bearings and attach fittings. Refer to Section 2A-14-13, Supplemental Inspection Document 55-30-01, for inspection procedure. Inspected vertical stabilizer and rudder, including spars, ribs, hinge bolts, hinge bearings and attach fittings as per Section 2A-14-13, Supplemental Inspection Document 55-30-01. No damage found - All okay at this time.	Initial: 10,000 hours or 20 years, whichever occurs first; Repeat: 3,000 hours or 5 years, whichever occurs first.	7	310, 340	Jun-27	3160.2
6/30/2022	TTAF 4119.8 TACH 160.2	The interval is for typical usage environment. 1. Inspect inboard wing structure for damage and working rivets. 2. Inspect flap actuator support structure. Refer to Section 2A-14-14, Supplemental Inspection Document 57-11-01, for inspection procedure. Inspected inboard wing structure for damage and working rivets as per Section 2A-14-14, Supplemental Inspection Document 57-11-01. No damage found - All okay at this time.	Initial: 12,000 hours or 20 years, whichever occurs first; Repeat: 2,000 hours or 10 years, whichever occurs first.	19	510, 610	Jun-27	3160.2
		This interval is for severe usage environment. 1. Inspect inboard wing structure for damage and working rivets. 2. Inspect flap actuator support structure. Refer to Section 2A-14-14, Supplemental Inspection Document 57-11-01, for inspection procedure. Not Applicable	Initial: 6,000 hours or 10 years, whichever occurs first; Repeat: 1,000 hours or 5 years, whichever occurs first.	20	510, 610		
6/30/2022	TTAF 4119.8 TACH 160.2	This interval is for mild/moderate corrosion environment. Inspect wing for corrosion and missing or loose fasteners. Refer to Section 2A-14-15, Supplemental Inspection Document 57-11-02, for inspection procedure. Inspected wings for corrosion as per Section 2A-14-15, Supplemental Inspection Document 57-11-02. None found. Very clean, Corrosion prevention applied Corrosion X Treatment.	Initial: 20 years; Repeat: 10 years	10	510, 520, 610, 620		
		This interval is for severe corrosion environment. Inspect wing for corrosion and missing or loose fasteners. Refer to Section 2A-14-15, Supplemental Inspection Document 57-11-02, for inspection procedure. Not Applicable	Initial: 10 years; Repeat: 5 years.	12	510, 520, 610, 620		
6/30/2022	TTAF 4119.8 TACH 160.2	This interval is for mild/moderate corrosion environment. Inspect wing splice joint at strut attach. Refer to Section 2A-14-16, Supplemental Inspection Document 57-11-03, for inspection procedure. Inspected wing splice joint at strut attach for corrosion as per Section 2A-14-16, Supplemental Inspection Document 57-11-03. None found.	Initial: 20 years; Repeat: 10 years	10	510, 610	Jun-32	
		This interval is for severe usage environment. Inspect wing splice joint at strut attach. Refer to Section 2A-14-16, Supplemental Inspection Document 57-11-03, for inspection procedure. Not Applicable	Initial: 10 years; Repeat: 5 years.	12	510, 610		
6/30/2022	TTAF 4119.8 TACH 160.2	This interval is for mild/moderate corrosion environment. Inspect wing root rib. Refer to Section 2A-14-17, Supplemental Inspection Document 57-12-01, for inspection procedure. Inspected wing root rib for corrosion as per Section 2A-14-17, Supplemental Inspection Document 57-12-01. None found.	Initial: 5 years; Repeat: 5 years	11	510, 610	Jun-27	
		This interval is for severe corrosion environment. Inspect wing root rib. Refer to Section 2A-14-17, Supplemental Inspection Document 57-12-01, for inspection procedure. Not Applicable	Initial 3 years; Repeat: 3 years	13	510, 610		
6/30/2022	TTAF 4119.8 TACH 160.2	This interval is for typical usage environment. Inspect wing strut and strut tube. Refer to Section 2A-14-18, Supplemental Inspection Document 57-40-01, for inspection procedure. Inspected wing strut and strut tube for cracks and corrosion as per Section 2A-14-18, Supplemental Inspection Document 57-40-01. None found.	Initial: 12,000 hours or 20 years, whichever occurs first; Repeat: 2,000 hours or 10 years, whichever occurs first.	19	510, 610	Jun-32	2160.2
		This interval is for severe usage environment. Inspect wing strut and strut tube. Refer to Section 2A-14-18, Supplemental Inspection Document 57-40-01, for inspection procedure. Not Applicable	Initial: 6,000 hours or 10 years, whichever occurs first; Repeat: 1,000 hours or 5 years, whichever occurs first.	20	510, 610		

D. R. Humble
Darrin R Humble, A&P 3446181

**Listing of OEM Programs (CAP), (SID), (CPCP)
Compliance and Current Status**

N705R

S/N R18202037

6/30/2022

Complied With		TASK Compliance Action	INTERVAL	OPERATION	ZONE	Next Due	
Date	Tach					Date	Tach
6/30/2022	TTAF 4119.8 TACH 160.2					Inspect aileron hinges, hinge bolts, hinge bearings and hinge and pushrod attach fittings. Refer to Section 2A-14-19, Supplemental Inspection Document 57-51-01, for inspection procedure. Inspected aileron hinges, hinge bolts, hinge bearings and hinge pushrod attach fittings as per Section 2A-14-19, Supplemental Inspection Document 57-51-01. All okay at this time.	Initial: 3,000 hours or 10 years, whichever occurs first; Repeat: 500 hours or 5 years, whichever occurs first.
6/30/2022	TTAF 4119.8 TACH 160.2	This interval is for mild/moderate corrosion environment. Inspect flap tracks for corrosion. Refer to Section 2A-14-20, Supplemental inspection Document 57-53-01, for inspection procedure. Inspected flap tracks for corrosion as per Section 2A-14-20, Supplemental Inspection Document 57-53-01. None Found.	Initial: 20 years; Repeat: 1 years.	10	510, 610	Jun-32	
		This interval is for severe corrosion environment. Inspect flap tracks for corrosion. Refer to Section 2A-14-20, Supplemental Inspection Document 57-53-01, for inspection procedure. Not Applicable	Initial: 10 years; Repeat: 5 years	12	510, 610		
6/30/2022	TTAF 4119.8 TACH 160.2	Inspect tubular engine mount. Refer to Section 2A-14-21, Supplemental Inspection Document 71-20-01, for inspection procedure. Inspected engine tubular motor mounts for cracks and rust as per Section 2A-14-21, Supplemental Inspection Document 71-20-01. None Found. Paint in good condition	Initial: 10,000 hours or 20 years, whichever occurs first; Repeat at Engine Overall.	9	120		2022.4
6/30/2022	TTAF 4119.8 TACH 160.2	Fuselage lower internal structure beneath the floor panels. Make sure you inspect these areas: 1. Cabin structure under floorboards. NOTE: Corrosion Prevention and Control Program Inspection item (baseline interval, refer to Section 2A-30-00 for additional inspection information). Inspected lower internal strut beneath floor panels for damage and corrosion as per Section 2A-30-00. None Found. Applied Corrosion X Treatment for Corrosion prevention.	Every 60 months	6	210	Jun-27	
6/30/2022	TTAF 4119.8 TACH 160.2	Fuselage internal structure in upper fuselage. Make sure you inspect these areas: 1. Cabin bulkhead corners. 2. Fuselage skin. NOTE: Corrosion Prevention and Control Program Inspection item (baseline interval, refer to Section 2A-30-00 for additional inspection information). Inspected fuselage internal structure in upper frame for damage and corrosion as per Section 2A-30-00. None Found. Applied Corrosion X Treatment for Corrosion prevention.	Every 60 months	6	210	Jun-27	
6/30/2022	TTAF 4119.8 TACH 160.2	Areas of the cabin structure. Make sure you inspect these areas: 1. Firewall. 2. Firewall attachments. NOTE: Corrosion Prevention and Control Program Inspection item (baseline interval, refer to Section 2A-30-00 for additional inspection information). Inspected areas of the cabin structure for damage and corrosion as per Section 2A-30-00. None Found. Applied Corrosion X Treatment for Corrosion prevention.	Every 60 months	6	210		
6/30/2022	TTAF 4119.8 TACH 160.2	Passenger/Crew door retention system. Make sure you inspect these areas: 1. Bell cranks. 2. Pushrods. 3. Handle. 4. Pin retention. 5. Pins. 6. Lockplates and guides. 7. Hinges. 8. Internal door framing. NOTE: Corrosion Prevention and Control Program Inspection item (baseline interval, refer to Section 2A-30-00 for additional inspection information). Note: Remove interior panels for access. Inspected R/H and L/H entrance doors for damage, function, correct operation, internal mechanism for condition as per Section 2A-30-00. All okay at this time. Applied Corrosion X Treatment for Corrosion prevention.	Every 48 months	5	210	Jun-26	

D-RH
Darrin R Humble, A&P 3446181

**Listing of OEM Programs (CAP), (SID), (CPCP)
Compliance and Current Status**

N705R

S/N R18202037

6/30/2022

Complied With		TASK	INTERVAL	OPERATION	ZONE	Next Due	
Date	Tach					Date	Tach
6/30/2022	TTAF 4119 8 TACH 160.2	Areas of the cabin structure for the passenger/crew door. Make sure you inspect these areas. 1. Door frames. 2. Door hinges. NOTE: Corrosion Prevention and Control Program Inspection item (baseline interval, refer to Section 2A-30-00 for additional inspection information) Inspected cabin structure for passenger/crew door, frame and hinge - all okay, no damage or corrosion found as per Section 2A-30-00. All okay at this time. Applied Corrosion X Treatment for Corrosion prevention.	Every 48 months	5	210	Jun-26	
6/30/2022	TTAF 4119 8 TACH 160.2	Areas of the cabin structure. Make sure you inspect these areas: 1. Cabin door forward and aft frames. 2. Window frames with emphasis at stringers and channel assemblies from aft of door frame to aft bulkhead. 3. Seat attachment structure. 4. Aft Cabin Bulkhead. NOTE: Corrosion Prevention and Control Program Inspection item (baseline interval, refer to Section 2A-30-00 for additional inspection information). Inspected cabin structure, cabin door forward and aft frames, window frames, aft cabin bulkhead as per Section 2A-30-00. No damage found.	Every 60 months	6	210	Jun-27	
6/30/2022	TTAF 4119 8 TACH 160.2	Flaps. 1. Check flap travel cable tension and travel time. 2. Check flap cable system, control cables and pulleys, in accordance with the flight cable inspection procedures in Section 2A-20-01, Expanded Maintenance, Control Cables. Inspected flap travel cable tension and travel time, cable and pulleys as per Section 2A-20-01.	Initial: 100 hours, Repeat: every 600 hours or 12 months, whichever occurs first.	23	210, 510, 610	Apr-25	760.2
6/30/2022	TTAF 4119 8 TACH 160.2	Aileron. 1. Check aileron travel and cable tension. 2. Check aileron cable system, control cables and pulleys, in accordance with the flight cable inspection procedures in Section 2A-20-01, Expanded Maintenance, Control Cables. Inspected aileron travel and cable tension, control cable and pulleys per Section 2A-20-01.	Initial: 100 hours, Repeat: every 600 hours or 12 months, whichever occurs first.	23	210, 510, 520, 610, 620	Apr-23	760.2
6/30/2022	TTAF 4119 8 TACH 160.2	Elevator. 1. Check elevator travel and cable tension. 2. Check elevator cable system, control cables and pulleys, in accordance with the flight cable inspection procedures in Section 2A-20-01, Expanded Maintenance, Control Cables. Inspected elevator travel and cable tension, control cable and pulleys per Section 2A-20-01.	Initial: 100 hours, Repeat: every 600 hours or 12 months, whichever occurs first.	23	210, 310, 320, 330		
6/30/2022	TTAF 4119 8 TACH 160.2	Elevator Trim. 1. Check elevator trim travel and cable tension. 2. Check elevator trim cable system, control cables and pulleys, in accordance with the flight cable inspection procedures in Section 2A-20-01, Expanded Maintenance, Control Cables. Inspected elevator trim travel and cable tension, control cable and pulleys per Section 2A-20-01.	Initial: 100 hours, Repeat: every 600 hours or 12 months, whichever occurs first.	23	210, 310, 320, 330	Apr-23	760.2
6/30/2022	TTAF 4119 8 TACH 160.2	Rudder. 1. Check rudder travel and cable tension. 2. Check rudder cable system, control cables and pulleys, in accordance with the flight cable inspection procedures in Section 2A-20-01, Expanded Maintenance, Control Cables. Inspected rudder travel and cable tension, control cable and pulleys per Section 2A-20-01.	Initial: 100 hours, Repeat: every 600 hours or 12 months, whichever occurs first.	23	210, 310, 340	Apr-23	760.2
6/30/2022	TTAF 4119 8 TACH 160.2	Wing structure internal. Make sure you inspect these areas: 1. Main spar upper and lower carry-thru fittings. 2. Main spar upper and lower caps. 3. Main spar web. NOTE: Corrosion Prevention and Control Program Inspection item (baseline interval, refer to Section 2A-30-00 for additional inspection information). Inspected wing internal structures, main spar upper and lower carry through fittings as per Section 2A-20-01. No Damage or corrosion found. Very clean. Applied corrosion prevention, Corrosion X Treatment.	Every 12 months	2	510, 520, 610, 620	Apr-23	760.2


Darrin R Humble, A&P 3446181

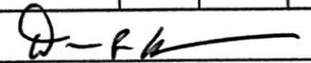
**Listing of OEM Programs (CAP), (SID), (CPCP)
Compliance and Current Status**

N705R

S/N R18202037

6/30/2022

Complied With		TASK Compliance Action	INTERVAL	OPERATION	ZONE	Next Due	
Date	Tach					Date	Tach
6/30/2022	TTAF 4119.8 TACH 160.2	Wing structure external. Make sure you inspect these areas: 1 Wing front spar and lower spar caps 2 Upper and lower wing attach spar fittings. 3. Wing lower skins. NOTE: Corrosion Prevention and Control Program Inspection item (baseline interval, refer to Section 2A-30-00 for additional inspection information) Inspected wing skin overlaps, access panels, rear spar, upper and lower caps, rear spar web as per Section 2A-30-01. No damage or corrosion found. Corrosion prevention applied, Corrosion X Treatment.	Every 60 months	6	510, 520, 610, 620	Jun-27	
6/30/2022	TTAF 4119.8 TACH 160.2	Wing structure external. Make sure you inspect these areas: 1 Skin with emphasis at skin overlaps and under access panels. 2 Rear spar upper and lower caps. 3. Rear spar web. NOTE: Corrosion Prevention and Control Program Inspection item (baseline interval, refer to Section 2A-30-00 for additional inspection information). Inspected wing skin overlaps, access panels, rear spar, upper and lower caps, rear spar web as per 2A-30-00. No damage or corrosion found. Corrosion prevention applied, Corrosion X Treatment.	Every 60 months	6	510, 520, 610, 620	Jun-24	
6/30/2022	TTAF 4119.8 TACH 160.2	Aileron attachments. Make sure you inspect these areas: 1 Aileron hinges. 2. Hinge bolts. 3. Hinge bearings. 4. Hinge and pushrod support structure. NOTE: Corrosion Prevention and Control Inspection Item (baseline interval, refer to Section 2A-30-00 for additional inspection information). NOTE: Do not apply LPS-3 Heavy Duty Rust Inhibitor on hinge bearing. Inspected aileron hinges, bolts, and bearings. Lubed hinges with LPS-2 as per 2A-30-00.	Every 24 months	3	520, 620	Jun-27	
6/30/2022	TTAF 4119.8 TACH 160.2	Vertical stabilizer structure Make sure you inspect these areas: 1. Forward spar attachment to tailcone bulkhead. 2. Aft spar attachment to lower stabilizer spar 3. Front and rear spars. 4. Rear spar rudder hinges. NOTE: Corrosion Prevention and Control Program Inspection Item (baseline interval, refer to Section 2A-30-00 for additional inspection information). Inspected vertical stabilizer, forward spar and aft spar attachment, front and rear spars as per 2A-30-00. No damage or corrosion found. Corrosion prevention applied, Corrosion X treatment.	Every 60 months	6	310, 340	Jun-27	
6/30/2022	TTAF 4119.8 TACH 160.2	Inspect main landing gear axle assembly. Make sure you inspect these areas: 1 Main gear axle and attach bolts 2 Wheel halves NOTE: Corrosion Prevention and Control Program Inspection item (baseline interval, refer to Section 2A-30-00 for additional inspection information) NOTE: Do not apply LPS-3 Heavy-Duty Rust Inhibitor to the bearing. NOTE: Coordinate with tire change. Inspected main gear axels, attach bolts, and wheel halves as per 2A-30-00. No damage or corrosion found.	Every 36 months	4	721, 722	Jun-25	
6/30/2022	TTAF 4119.8 TACH 160.2	Nose gear trunion, steering assembly, torque link assembly, nose gear fork and axle. Make sure you inspect these areas: 1. Nose gear trunion surface. 2. Steering collar and steering collar attach bolt. 3. Torque link, torque link attach pin and attach bolt 4. Nose gear fork 5. Nose gear axle. NOTE: Corrosion Prevention and Control Inspection Item (baseline interval, refer to Section 2A-30-00 for additional inspection information). Inspected nose gear trunion steering assy, torque links, fork and axel, steering collar, etc. as per 2A30-00. No damage or corrosion found.	Every 36 months	4	720	Jun-25	
6/30/2022	TTAF 4119.8 TACH 160.2	Inspect nose gear trunion, torque link assembly and nose gear fork. Make sure you inspect these areas: 1. Nose gear trunion upper and lower inner bore surface and bearn. 2. Torque link bolt and attach pin inner bore surface. 3. Nose gear fork lug inner bore surface. NOTE: Corrosion Prevention and Control Inspection Item (baseline interval, refer to Section 2A-30-00 for additional inspection information). Inspected nose gear trunion link assy, upper and lower inner bore surfaces and associated hardware as per 2A30-00. No damage or	Every 36 months	4	720	Jun-25	


Darrin R Humble, A&P 3446181

**Listing of OEM Programs (CAP), (SID), (CPCP)
Compliance and Current Status**

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S/N R18202037

6/30/2022

Complied With		TASK Compliance Action	INTERVAL	OPERATION	ZONE	Next Due	
Date	Tach					Date	Tach
6/30/2022	TTAF 4119.8 TACH 160.2	Nose landing gear outer barrel assembly. Make sure you inspect these areas: 1. Outer barrel assembly 2. Upper strut end and lower collar assembly. NOTE: Corrosion Prevention and Control Inspection Item (baseline interval, refer to Section 2A-30-00 for additional inspection information) NOTE: do not apply LPS-3 Heavy-Duty Rust Inhibitor to the sliding surfaces of the oleo strut. Inspected nose gear outer barrel, upper strut end and lower collar assy as per Section 2A-30-00. No damage or corrosion found.	Every 36 months	4	720	Jun-25	
6/30/2022	TTAF 4119.8 TACH 160.2	Nose gear axle assembly. Make sure you inspect these areas: 1. Nose gear axle and attach bolt. 2. Wheel halves. NOTE: Corrosion Prevention and Control Program Inspection Item (baseline interval, refer to Section 2A-30-00 for additional inspection information) NOTE: Disassemble the nose gear strut to get access. NOTE: Do not apply LPS-3 Heavy-Duty Rust Inhibitor to the sliding surfaces of the oleo strut. NOTE: Coordinate with the change. Inspected nose gear axle and both wheel halves as per Section 2A-30-00. No damage or corrosion found.	Every 60 months	6	720	Jun-27	
6/30/2022	TTAF 4119.8 TACH 160.2	Horizontal stabilizer structure. Make sure you inspect these areas: 1. Stabilizer attachment to the tailcone bulkhead. 2. Front and rear spars. NOTE: Corrosion Prevention and Control Program Inspection Item (baseline interval, refer to Section 2A-30-00 for additional inspection information). Inspected horizontal rear tailcone bulkhead structure, front and rear spars as per Section 2A-30-00. No damage or corrosion found. Corrosion prevention applied, Corrosion X Treatment.	Every 60 months	6	320, 330	Jun-27	
6/30/2022	TTAF 4119.8 TACH 160.2	Elevator trim system. Make sure you inspect these areas: 1. Elevator trim brackets. 2. Actuator support brackets and bearings. 3. Pulleys and attaching structure. NOTE: Corrosion Prevention and Control Inspection Item (baseline interval, refer to Section 2A-30-00 for additional inspection information) NOTE: Do not apply LPS-3 Heavy-Duty Rust Inhibitor on hinge bearing. Inspected trim systems, brackets, actuator support, bearings, pulleys, etc. as per Section 2A-30-00. No damage or corrosion found. Treated structural areas around trim with Corrosion X Treatment.	Every 24 months	3	320, 330	Jun-25	
6/30/2022	TTAF 4119.8 TACH 160.2	Rudder attachments. Make sure you inspect these areas: 1. Hinge brackets. 2. Hinge bolts. 3. Hinge bearings. NOTE: Corrosion Prevention and Control Inspection Item (baseline interval, refer to Section 2A-30-00 for additional inspection information) NOTE: Do not apply LPS-3 Heavy-Duty Rust Inhibitor on hinge bearing. Inspected rudder attachments, hinge brackets, bolts, bearings as per Section 2A-30-00. No damage or corrosion found.	Every 24 months	3	340	Jun-25	
6/30/2022	TTAF 4119.8 TACH 160.2	Rudder structure. Make sure you inspect these areas: 1. Skin. 2. Forward and aft spars at hinge locations. NOTE: Corrosion Prevention and Control Inspection Item (baseline interval, refer to Section 2A-30-00 for additional inspection information). Inspected rudder structure skins and spars as per Section 2A-30-00. No damage or corrosion found. Corrosion prevention applied, Corrosion X Treatment.	Every 24 months	3	340	Jun-24	
6/30/2022	TTAF 4119.8 TACH 160.2	Engine support structure. Make sure you inspect these areas: 1. Engine truss. Pay particular attention to vicinity of welds. NOTE: Corrosion Prevention and Control Program Inspection Item (refer to Section 2A-30-00 for additional inspection information). Inspected engine support structure, engine truss and welds as per Section 2A-30-00. All okay at this time, no damage found and paint in good condition.	Every 12 months	2	120	Jun-23	

D. R. Humble
Darrin R Humble, A&P 3446181

NAME/
SIGNATURE

LICENCE:
AME or
AMO/ACA

6-30-2022
Tach 160.2

The TAB-5 MOUNT + RA-2A ANTENNAS + ASSOCIATED COAX RUN FROM ANTENNAS TO BNC CONNECTORS AT TOP OF WING STRUT HAVE BEEN INSTALLED, TWO RESTRICTED PLACARDS INSTALLED, RESTRICTED CATEGORY AIRWORTHINESS CERTIFICATE DISPLAYED, WEIGHT + BALANCE AMENDED TO REFLECT THE INSTALLATION OF ANTENNAS MOUNTS + COAX EXTENSION,
D-RP ARP 3346181

7-12-2022
Tach 169.5

REMOVED THE TAB-5 MOUNT + RA-2A ANTENNAS + ASSOCIATED COAX RUN FROM ANTENNAS TO BNC BULKHEAD AT THE TOP OF WING STRUTS, TWO RESTRICTED PLACARDS REMOVED STANDARD AIRWORTHINESS ^{CERTIFICATE} DISPLAYED, WEIGHT + BALANCE AMENDED
D-RP ARP 3346181

Date 8-1-22 Tach Time 172.0

Installed the TAB-5 Mount and RA-2A Antennas and associated Coax that runs from antennas to BNC Bulkhead at top of the wing struts. Two restricted placards installed, the Restricted Category Airworthiness Certificate displayed. Weight and Balance Amended to reflect the installation of Antenna Mount and Coax Extensions.

D-RP ARP 3346181
Signature and A&P #

Date 8-2-22 Tach Time 174.7

Removed the TAB-5 Mount and RA-2A Antennas and associated Coax that runs from antennas to BNC Bulkhead at top of the wing struts. Two restricted placards removed, the Standard Airworthiness Certificate displayed. Weight and Balance Amended.

D-RP ARP 3346181
Signature and A&P #

Date 8-4-22 Tach Time 179.2

Installed the TAB-5 Mount and RA-2A Antennas and associated Coax that runs from antennas to BNC Bulkhead at top of the wing struts. Two restricted placards installed, the Restricted Category Airworthiness Certificate displayed. Weight and Balance Amended to reflect the installation of Antenna Mount and Coax Extensions.

D-RP ARP 3346181
Signature and A&P #

Date 8-7-22 Tach Time 184.3

Removed the TAB-5 Mount and RA-2A Antennas and associated Coax that runs from antennas to BNC Bulkhead at top of the wing struts. Two restricted placards removed, the Standard Airworthiness Certificate displayed. Weight and Balance Amended.

D-RP ARP 3346181
Signature and A&P #

* Date 8-18-22 Tach Time 194.5

Installed the TAB-5 Mount and RA-2A Antennas and associated Coax that runs from antennas to BNC Bulkhead at top of the wing struts. Two restricted placards installed, the Restricted Category Airworthiness Certificate displayed. Weight and Balance Amended to reflect the installation of Antenna Mount and Coax Extensions.

D-RP ARP 3346181
Signature and A&P #

Date 8-19-22 Tach Time 195.1

Removed the TAB-5 Mount and RA-2A Antennas and associated Coax that runs from antennas to BNC Bulkhead at top of the wing struts. Two restricted placards removed, the Standard Airworthiness Certificate displayed. Weight and Balance Amended.

D-RP ARP 3346181
Signature and A&P #

Section 1: Airframe Log

Tach Time 203.0 hrs.
TTAF: 4162.6 hrs.
Engine TT: 1159.4 hrs.
Eng. SMOH: 180.6 hrs.
Prop TT 4162.6hrs.
Prop SMOH 180.6 hrs.

N705R
Cessna TR182
Lycoming O-540_L3C5D

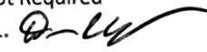
May 1, 2023
S/N R18202037
S/N L-27116-40E

Performed Annual Inspection this date I/A/W Cessna R182 and TR182 Series Service Manual, P/N D2069-3-13, Section 2, Checklist Para 2-59, Para 2-60 pages 2-38 through 2-54. Also including Service Manual Temporary Revisions TR5 through TR9.

Removed oxygen bottle and had hydrotested, reinstalled in aircraft. Next Hydro test due 4-2028. Replaced worn throttle cable retainer at air box with new, P/N 225234-1 retainer. Replaced cracked mixture cable clamp at carburetor bracket, P/N MS21919-WDG6. Replaced vacuum pump cooling scatt with new, P/N Scatt-4. Secured Nose Strut with MIL-H-5606 Fluid and Nitrogen. Secured wiring around electric fuel pump. Tightened and resafted R/H Forward Exhaust Clamp. Performed gear swing using Viper Aviation 37 item checklist. Greased gear. Complied with A.D. 2011-10-09, dated 6-17-2011, Seat Rail and Roller Housing Inspection I/A/W Para "G" – all okay at this time, next due TACH 303.0 hours or d5-1-2024. A.D. 22-16-03, dated 8-15-2022, Bendix/TCM 1200 Series Mag is N/A, 1200 Series Mag not installed. Complied with ELT Inspection and Function check I/A/W Artex ME406 Doc. Installation and Maintenance Manual, P/N 570-1600 Rev. V, dated 3-19-15, items A through R with reference to FAR 91.207 D1 through 4 – all okay at this time, next inspection due 5-2024, next battery due 5-2028. Installed new aircraft battery, P/N Concord RG24-11M.

2	Performed Cessna Continued Airworthiness Program (CAP), (SID), (SSID), (CPCP), (EWIS).
3	Operation #1, verifying that all Cessna Service Bulletins, Service Letters, Suppliers Service Bulletins, All Airworthiness Directives and all SID inspections are complied with.
4	Operation #2, inspect wing, all spars, lower CRPs, main spar web, engine support structure, engine truss and welds, etc. I/A/W Section 2A-20-01, section 2A-30-01.
5	Operation #23, Checked A.) Flap cable tension and travel time. Checked flap cable system, control cables, pulleys. B.) Elevator trim travel and cable tension, elevator trim cable system, cables and pulleys. C.) Checked Elevator travel and cable tension, elevator cable system control cables and pulleys. D.) Checked rudder travel and cable tension, rudder cable system control cables and pulleys. E.) Checked aileron travel and cable tension, aileron cable system, control cables and pulleys I/A/W Flight Cable Inspection Procedures in Section 2A-20-01.

The Following instructions for continued airworthiness have been performed:

- Garmin GNS 430 WAAS ICA I/A/W Garmin ICA. DOC 190-00356-65 Rev. D
 - Garmin GTX 345 Transponder ICA. I/A/W Garmin Doc 190-00734-11 Rev 8 Section 4
 - FM Transceiver Tecinonics TDF 136B ICA. I/A/W ICA TII Dec. 06Re379 Rev N/C
 - Audio Panel NAT ASM 44 on condition only, periodic maintenance not required
 - Engine Monitor J.P. Instrumars EGT Scanner #102. On condition only, periodic maintenance not required
 - Air Filter Donaldson P13-1367 ICA I/A/W DOC P469075 Rev 8 Repaired at 0.0 Tach. Next replacement due per A.D. 86-24-02 EA 500 hours
 - McCauley Propeller B3D32C407-C I/A/W McCauley Propeller Systems Manual, P/N MPC 26-05 Rev 5, dated 8-1-2020 Intro Section 4
 - S-TEC ST-255 Electric Trim System ICA S-TEC Doc. 81364
 - BAS Inc. Inertia Reel Shoulder Harness System ICA I/A/W DOC 1 Items 1. Through 9.
 - Telonics Inc. Tracking Antennas ICA See 337 form page 4 of 4 para 1
 - Zeftronics Model No. R25400 ICA See DOC/Drawing ZOO1D5 Rev A, dated 5-17-99
 - M-20 AIR/OIL Separator No ICA Listed
 - SK210-174B Pilot Secondary Seat Stop ICA I/A/W SEB07-5 Rev 4 ICA Rev A page 14-7
 - R/STOL System ICA. DOC Robertson Aircraft Corp. Service and Maintenance Manual 10, Series Rev. dated 4-1-74, sections 1 through 7
 - Flint Aero Inc. 20 each 12 gallon Aux Fuel Tanks No ICA. Listed. Check Fittings for leaks, fuel cap and seal for condition inspect and clean fuel pump filter and main fuel strainer
 - Shadin Fuel Flow Indicator System Mini-flow 912041T-D on Condition Only. Periodic Maintenance not Required
- A.D.'s current to 2023-09. Maintenance Performed by Evan Bishop, A&P and Darrin Humble A&P3346181. 

I certify that this Aircraft has been inspected in accordance with an Annual inspection and was determined to be in airworthy condition.



Darrin R Humble
A&P 3346181 I.A.

Tach Time 291.8 hrs.
TTAF: 4251.4 hrs.
Engine TT: 1248.2 hrs.
Eng. SMOH: 269.4 hrs.
Prop TT 4251.4 hrs.
Prop SMOH 269.4 hrs.

N705R
Cessna TR182
Lycoming O-540_L3C5D

August 25, 2023
S/N R18202037
S/N L-27116-40E

Performed Annual Inspection this date I/A/W Cessna R182 and TR182 Series Service Manual, P/N D2069-3-13, Section 2, Checklist Para 2-59, Para 2-60 pages 2-38 through 2-54. Also including Service Manual Temporary Revisions TR5 through TR9.

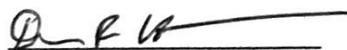
Removed shimmy dampner, disassembled, cleaned, replaced o-rings. Reassembled, bled dampner several times to get all air out. Reinstalled on aircraft. Removed Nose Tire, greased wheel bearings, removed old balance weights, cleaned and balanced with new balance weights, reinstalled on aircraft. Adjusted alternator cable from back of alternator and installed anti-chafe spiral wrap to prevent it from chaffing on forward baffle. Saftied inboard fuel drains. Replaced anti-chafe tape at lower wing strut root fairing contact point. A.D. 2023-09-09, dated 7-17-2023, V-Band Exhaust Clamps due Tach time 315.8 hrs. A.D. 23-10-02, dated 6-26-2023, Radar Altimeter – N/A not installed on aircraft. Performed gear swing using Viper Aviation 37 item checklist. Greased gear. Complied with A.D. 2011-10-09, dated 6-17-2011, Seat Rail and Roller Housing Inspection I/A/W Para "G" – all okay at this time, next due TACH 391.8 hours or 8-2024. Complied with ELT Inspection and Function check I/A/W Artex ME406 Doc. Installation and Maintenance Manual, P/N 570-1600 Rev. V, dated 3-19-15, items A though R with reference to FAR 91.207 D1 through 4 – all okay at this time, next inspection due 8-2024, next battery due 5-2028.

2	Performed Cessna Continued Airworthiness Program (CAP), (SID), (SSID), (CPCP), (EWIS).
3	Operation #1, verifying that all Cessna Service Bulletins, Service Letters, Suppliers Service Bulletins, All Airworthiness Directives ad all SID inspections are complied with.
4	Operation #2, inspect wing, all spars, lower CRPs, main spar web, engine support structure, engine truss and welds, etc. I/A/W Section 2A-20-01, section 2A-30-01.

The Following instructions for continued airworthiness have been performed:

- Garmin GNS 430 WAAS ICA I/A/W Garmin ICA. DOC 190-00356-65 Rev. D
 - Garmin GTX 345 Transponder ICA. I/AW Garmin Doc 190-00734-11 Rev 8 Section 4
 - FM Transceiver Tecnosonics TDF 136B ICA. I/A/W ICA Til Dec. 06Re379 Rev N/C
 - Audio Panel NAT ASM 44 on condition only, periodic maintenance not required
 - Engine Monitor J.P. Instrumars EGT Scanner #102. On condition only, periodic maintenance not required
 - Air Filter Donaldson P13-1367 ICA I/A/W DOC P469075 Rev 8 Repaired at 0.0 Tach. Next replacement due per A.D. 86-24-02 EA 500 hours
 - McCauley Propeller B3D32C407-C I/A/W McCauley Propeller Systems Manual, P/N MPC 26-05 Rev 5, dated 8-1-2020 Intro Section 4
 - S-TEC ST-255 Electric Trim System ICA S-TEC Doc. 81364
 - BAS Inc. Inertia Reel Shoulder Harness System ICA I/A/W DOC 1 Items 1. Through 9.
 - Telonics Inc. Tracking Antennas ICA See 337 form page 4 of 4 para 1
 - Zeftronics Model No. R25400 ICA See DOC/Drawing ZOO1D5 Rev A, dated 5-17-99
 - M-20 AIR/OIL Separator No ICA Listed
 - SK210-174B Pilot Secondary Seat Stop ICA I/A/W SEB07-5 Rev 4 ICA Rev A page 14-7
 - R/STOL System ICA. DOC Robertson Aircraft Corp. Service and Maintenance Manual 10, Series Rev. dated 4-1-74, sections 1 through 7
 - Flint Aero Inc. 20 each 12 gallon Aux Fuel Tanks No ICA. Listed. Check Fittings for leaks, fuel cap and seal for condition inspect and clean fuel pump filter and main fuel strainer
 - Shadin Fuel Flow Indicator System Mini-flow 912041T-D on Condition Only. Periodic Maintenance not Required
- A.D.'s current to 2023-17. Maintenance Performed by Evan Bishop, A&P and Darrin Humble A&P3346181. 

I certify that this Aircraft has been inspected in accordance with an Annual inspection and was determined to be in airworthy condition.



Darrin R Humble
A&P 3346181 I.A.

11-16-23
TALL 292.1

Reweighed Aircraft
Empty weight 2035.00
CG 38.50 gear up 40.00
MAX GROSS WEIGHT 3100
USEFUL LOAD 1065.0

AIP374618/

D-R/D

NAME/
SIGNATURE

LICENCE:
AME or
AMO/ACA

Section 1: Airframe Log

Tach Time 314.4 hrs.
TTAF: 4274.0 hrs.
Engine TT: 1270.8 hrs.
Eng. SMOH: 292.0 hrs.

N705R
Cessna TR182
Lycoming O-540_L3C5D
Prop TT 4274.0 hrs.

January 22, 2025
S/N R18202037
S/N L-27116-40E
Prop SMOH 0.0 hrs.

Performed Annual Inspection this date I/A/W Cessna R182 and TR182 Series Service Manual, P/N D2069-3-13, Section 2, Para 2-59 "J" page 2-46 through 2-48 including TR5 through TR9. Replaced L/H Main Tire with new Goodyear Flight Custom III, P/N 15-600-6 6ply, reconditioned R/H and L/H brake pressure and back plate assemblies and installed new brake linings, P/N 8 each 066-10500 Linings and f16 each P/N 105-00200 Brake Rivets. Removed emergency landing gear extension hand pump due to leaks, disassembled, cleaned, inspected, resealed with new orings, reassembled, sealed threads, reinstalled in aircraft. Tested and leak check good. Replaced Beacon Bulb, P/N 34-0226010-91, replaced dome light bulb, P/N GE1820. Replaced Control Wheel Trim Switch, P/N CA3112-6. Performed gear swing using Viper Aviation 37 item checklist, greased gear. Complied with A.D. 2023-09-09, dated 7-17-23, Turbo V-Band Coupling, replaced with new clamp, P/N S1921-2. Total Time 0.0 hrs. Next Clamp Replacement due at Tach 814.3 hours. Complied with A.D. 2011-10-09, dated 6-17-2011, Seat Rail and Roller Housing Inspection, inspected I/A/W Para "G" – all okay at this time. Next due Tach 414.3 hours or f1-2026. Complied with ELT Inspection and Function Check I/A/W Artex ME406 Doc. Installation and Maintenance Manual, P/N 570-1600, Rev. V, dated 3-19-15, items A through R with reference to FAR 91.207 D1 through 4. All okay at this time, next inspection 1-2026. Next Battery due 5-2028.

1	Performed Cessna Continued Airworthiness Program (CAP), (SID), (SSID), (CPCP), (EWIS).
2	Operation #1, verifying that all Cessna Service Bulletins, Service Letters, Suppliers Service Bulletins, All Airworthiness Directives and all SID inspections are complied with.
3	Operation #2, inspect wing, all spars, lower CRPs, main spar web, engine support structure, engine truss and welds, etc. I/A/W Section 2A-20-01, section 2A-30-01.
4	Operation #4, inspect main gear axles attach bolts and wheel halves. Inspected nose gear trunion steering assembly torque links, fork and axle steering collar, etc. Inspected nose gear trunion link assembly, and upper and lower inner boar surfaces and association hardware as per 2A30-00 – no damage or corrosion found.
5	Operation #22, inspect forward door post and surrounding structure, refer to section 2A-14-08, Supplemental Inspection Document 53-12-01. Complied with per A.D. 2020-18-01 and Cessna SEB95-19. C/W on 8-23-2023, TACH 291.8 hrs.
6	Operation #23, inspect flap travel, cable tension and travel time cable and pulleys. Inspect aileron travel and cable tension control cable and pulleys. Inspect elevator travel and cable tension, control cable and pulleys. Inspect elevator trim travel and cable tension, control cable and pulleys. Inspect rudder travel and cable tensions, control cable and pulleys, per Section 2A-20-01

The Following instructions for continued airworthiness have been performed:

- Garmin GNS 430 WAAS ICA I/A/W Garmin ICA. DOC 190-00356-65 Rev. D
 - Garmin GTX 345 Transponder ICA. I/A/W Garmin Doc 190-00734-11 Rev 8 Section 4
 - FM Transceiver Tecnosonics TDF 136B ICA. I/A/W ICA Til Dec. 06Re379 Rev N/C
 - Audio Panel NAT ASM 44 on condition only, periodic maintenance not required
 - Engine Monitor J.P. Instrumars EGT Scanner #102. On condition only, periodic maintenance not required
 - Air Filter Donaldson P13-1367 ICA I/A/W DOC P469075 Rev 8 Repaired at 0.0 Tach. Next replacement due per A.D. 86-24-02 EA 500 hours
 - McCauley Propeller B3D32C407-C I/A/W McCauley Propeller Systems Manual, P/N MPC 26-05 Rev 5, dated 8-1-2020 Intro Section 4
 - S-TEC ST-255 Electric Trim System ICA S-TEC Doc. 81364
 - BAS Inc. Inertia Reel Shoulder Harness System ICA I/A/W DOC 1 Items 1. Through 9.
 - Telonics Inc. Tracking Antennas ICA See 337 form page 4 of 4 para 1
 - Zeftronics Model No. R25400 ICA See DOC/Drawing ZOO1D5 Rev A, dated 5-17-99
 - M-20 AIR/OIL Separator No ICA Listed
 - SK210-174B Pilot Secondary Seat Stop ICA I/A/W SEB07-5 Rev 4 ICA Rev A page 14-7
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 - Flint Aero Inc. 20 each 12 gallon Aux Fuel Tanks No ICA. Listed. Check Fittings for leaks, fuel cap and seal for condition inspect and clean fuel pump filter and main fuel strainer
 - Shadin Fuel Flow Indicator System Mini-flow 912041T-D on Condition Only. Periodic Maintenance not Required
- A.D.'s current to 2025-01. Maintenance Performed by Shane Babcock, A&P and Darrin Humble A&P3346181.

I certify that this Aircraft has been inspected in accordance with an Annual inspection and was determined to be in airworthy condition.


Darrin R Humble
A&P 3346181 I.A.

246

NAME/
SIGNATURE

LICENCE:
AME or
AMO/ACA

Avid Avionics LLC
CRS#: AVDR404E
WON#: 202504015

2890 25th St SE
Salem, OR 97302
Phone: 971-332-9699

Date	Tach Time	Hobbs Time	Aircraft Make, Model, S/N, Registration
4/22/25	0318.2	3250.1	CESSNA R182, R-18202037, N705R

Description:

1. Performed 14 CFR 91.411 certifications of United altimeter (P/N: 5934PA-1, S/N: 30004), static system, and altitude reporting system and all have been found to meet the requirements of Part 43 Appendix E to a pressure altitude of 20,000 ft.
2. Performed 14 CFR 91.413 certification of the transponder. The Garmin GTX 345 (P/N: 011-03302-00, S/N: 3EG024326) has been tested and found to meet the requirements of 14 CFR Part 43 Appendix F.

The work described above was performed in accordance with 14 CFR part 43, was subject to final inspection, and is approved for return to service only for the work performed.

Name: REESE WHITWORTH Signature:  Cert #: AVDR404E

6/12/25 Tach 318.9 Removed original Rudder P/N 1231001-13
 Tach 4278.5 Removed original VERTICAL FIN Assy P/N 1231057-18
 Installed Repaired VERTICAL Fin Assy P/N 1231057-18 Repaired by
 Airframe Components by Williams, Inc under w/p # 25383 DARD 2-24-25
 See 8130-3, Installed Repaired Rudder Assy 1231001-13 Repaired by
 Airframe Components by Williams, Inc under w/p # 25382 DARD 2-24-25
 See 8130-3, Both Surfaces Aired + Painted Rudder Balance checked
 I/A/W CESSNA PART 182 Service MANUAL P/N D2069-7-13 CHAPTER 17
 Pg 17-12 to 17-14 FIG 17-3 sheet 135, SURFACES INSTALLED I/A/W
 CHECK 4 PART 4-14 + 4-16 CHAPTER 10 PART 10-7
 D-R A-P 3346181